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HONEYWELL INC HOPKINS MN DEFENSE SYSTEMS DIV  
PRODUCIBILITY ENGINEERING AND PLANNING (PEP) OF THE XM74 GEMSS --ETC(U)  
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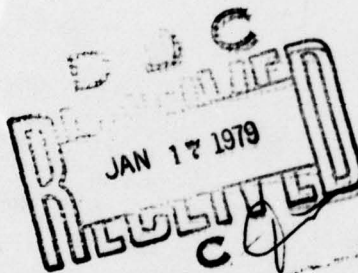


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**PRODUCIBILITY ENGINEERING AND PLANNING (PEP)  
OF THE  
XM74 GEMSS EXTENDED TRIPLINE SENSOR**

AD A063424



30 August 1978

**Final Technical Report**

**For Period**

**May 1977 to August 1978**

This document has been approved  
for public release and sale; its  
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**Prepared For:**

**United States Army  
Armament Research and Development Command  
Dover, New Jersey 07801**

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(Cont) The section devoted to sensor production engineering presents material on the initial design and describes proposed design changes. A second  
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major section deals with 1. Extended Range Tripline Sensor production engineering, and  
with 2. Machine line and test equipment design.

Also included is support effort such as Reliability, Availability, Maintainability, Safety and Human Factors Engineering.

The section devoted to sensor Production Engineering presents material on the initial design and proceeds to describe proposed design changes.

Cost data is also presented to cover the initial design and the final design resulting from the approved changes.

A final section presents recommendations for continuing activities related to the XM74 GEMSS Extended Tripline Sensor.

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## I. INTRODUCTION

### A. PURPOSE AND SCOPE

This final technical report is being submitted in fulfillment of the requirements of Contract DAAK10-77-C-0047, DD Form 1423 Data Item A004. The report covers Honeywell activities pursuant to this contract, which called for a Producibility Engineering and Planning (PEP) effort on the XM74 GEMSS Extended Tripline Sensor. The overall purpose of the effort was to support Army mine programs by maximizing the producibility of the Extended Tripline Sensor and by evolving in parallel an automated assembly machine production line and associated processes for high volume, low cost production. The work was conducted for ARRADCOM, and technical management was provided by Joe Rehkamp of the Large Caliber Weapons System Laboratory.

### B. REPORT ORGANIZATION

This report is organized to provide separate sections for the two primary activities involved (Extended Tripline Sensor Production Engineering and Machine Line and Test Equipment Design) as well as supporting effort; i.e., reliability, availability, maintainability, safety, etc. The section on sensor Production Engineering is organized to present the initial design first and then to describe the design changes proposed in the program. This enables a clear perception of what changes were evolved and how they affected the baseline design. Cost data are contained in a separate section and cover the initial design cost and the cost as the result of approved changes. The last section of this report contains recommendations for on-going activities related to the XM74 GEMSS Extended Tripline Sensor.

### C. CONTRACTUAL MATTERS

The original contract (DAAK10-77-C-0047) covered a two Phase effort running from 11 May 1977 to 30 November 1977. Contract history is as follows:

---



Letter Contract

11 May 1977

\$98,679

Mod P00001

6 June 1977

Definitized contract

Mod P00002

27 September 1977

Extension for end of  
contract to 30  
December 1977.

Mod P00003

9 February 1978

Supplement to scope of  
work added \$131,248  
and extended end of  
contract to 28 February  
1978.

Mod P00004

20 July 1978

Extended end of contract  
to 30 August 1978.

## II. SUMMARY

The program was initiated on 19 May 1977 with the intent to maximize the producibility (minimize the production cost) of the Extended Tripline Sensor and to evolve an automated assembly machine production line. The effort was conducted in two phases. Phase I activities concentrated on Tripline Sensor producibility, the design baseline for automated assembly, and the initial cost estimate. In Phase II, the emphasis was on finalization of the automated assembly machine production line and preparation of supporting technical data, including process descriptions, Quality inputs, RAM, Safety, and Human Factors Engineering.

### A. PRODUCTION ENGINEERING EFFORT, XM74 GEMSS EXTENDED TRIPLINE SENSOR

In the GEMSS Extended Tripline Production Engineering effort, change proposals involving ten (10) drawings were prepared and submitted of which nine (9) were approved (see Table 1). Changes were made to the Tripline Sensor baseline design only in those cases where Project Office approval of the change proposal was received.

The approved changes increased the automated producibility of the GEMSS Sensor. Changes which increased producibility and/or directly reduced costs included the incorporation of features and configuration revisions to permit automatic handling such as the hole in the case and sleeve.

The proposed changes to the bobbin are needed to permit automatic winding in conjunction with the assembly of the weight to the bobbin.

For each proposed change, an evaluation was made to confirm the predicted benefits and to ensure no adverse effect on performance, reliability and/or safety.

The final configuration reflects a tripline sensor fully capable of automatic assembly which was not true for the baseline design.

On the basis of substantial improvements in producibility, it is concluded that Production Engineering Objective #1 for the XM74 GEMSS Extended Range Tripline Sensor was achieved during the program.

Objective #2 requires the following information pertinent to Extended Tripline Sensor production which is common (identical to the ADAM mine tripline sensor assembly facility):



TABLE 1. CHANGE PROPOSALS

Drawing Change	Description	Date Submitted	Response
0. Original	Baseline Design	10/27/77	N/A
1. 9292972	Sensor Assembly Show Piece Part Revision	8/15/77	Approved
2. 9298578	Spring Ejection	8/15/77	Rejected
3. 9292976	Sensor Case and Bobbin Assembly Pictorial Changes	8/15/77	Approved
4. 9292981	Eyelet, Interface Revise to Washer Configuration	8/15/77	Approved
5. 9292982 9292983	Bobbin Assembly Weight, Bobbin	8/15/77	Approved
6. 9292985	Bobbin Add 30° Taper, Relocate .03 Slot	8/15/77 7/31/78	Approved Open
7. 9292986	Release Mech and Case Assembly Pictorial Changes	8/15/77	Approved
8. 9292987	Case, Sensor Add (2) Sets of (4) Holes Add (4) Flutes	8/15/77	Approved
9. 9292988	Sleeve Add (2) Sets of (4) Holes	8/15/77	Approved
10. 9292989	Washer, Case Enlarge Chamber	8/15/77	Approved

- Production tooling and fixturing
- Assembly and inspection facility designs
- Manufacturing and assembly process descriptions
- Software
- Provide drawings, specifications and technical data as needed to replicate common facility items.

This information was submitted as the Common Design Study, dated 1 November 1977, and includes the following common automatic assembly machines:

Machine No. 8	Diaphragm Assembly
Machine No. 9	Housing Assembly
Machine No. 10	Release Mechanism Assembly
Machine No. 10A	Terminal Assembly

Objective #3 requires the following information pertinent to Extended Tripline Sensor production which is similar to the ADAM mine tripline sensor assembly facility and which may be modified for Extended Tripline Sensor production:

- Production tooling and fixturing
- Manufacturing and assembly process descriptions
- Software
- Design modifications and technical data to implement fabrication of modified facilities.

This information was submitted 12 June 1978 and includes the following automatic assembly machines which are similar to the ADAM machines:

Machine No. 11	Release Mechanism and Case
Machine No. 12	Sensor Case and Bobbin Assembly
Machine No. 13	Sensor and Breakwire Assembly
Machine No. 13A	Epoxy Dispenser

Objective #4 requires the following information pertinent to Extended Tripline Sensor production which is unique to the Extended Tripline Sensor assembly facility:

- Production tooling and fixturing
- Manufacturing and assembly process descriptions
- Software
- Designs and technical data as required to construct these unique facilities.



This information was submitted 12 June 1978 and includes the following automatic assembly machines which are new and unique to the Extended Tripline Sensor:

Machine No. 7      Bobbin Assembly  
Machine No. 12B    Sensor Case and Spring Assembly

Objective #5 requires a cost estimate for final development, fabrication and installation of the complete production facility. This information is presented in Section VII of this report.

#### B. MACHINE LINE DESIGN

The design of the machine line was predicated on providing a new, highly automated production configuration to achieve the volume output capability required (200,000 units per month on a one shift, 8-hour day, 5-day week (1-8-5) shift basis and using a 50 minute hour and 21 days per month) at the lowest unit product cost. The PEP effort involved a new sensor; however, there were applicable design resources by way of machine designs from the ADAM sensor line which proved to benefit the GEMSS Sensor program. These resources were used to the maximum extent possible to facilitate the machine line design for the Extended Tripline Sensor. (Ref: Figure 1. Assembly Flow Chart.)

The basic equipment required for the machine line was determined by means of relating the Extended Tripline Sensor to the ADAM sensor. Consideration given to commonality and similarity between the two sensors made it possible to take advantage of existing ADAM machines and equipment for application to the extended range sensor. The resulting three categories; common, similar and new, have been described above. Preliminary design work and breadboard models resulted in one entirely new machine, No. 12B. This was necessary in order to avoid unacceptable complexity in machine No. 12. One of the ADAM machines has no equivalent in the GEMSS Sensor line. This is machine No. 7A, for Bobbin taping. The extended range sensor does not use tape at this point, since the Bobbin Weight retains the tripline in the Bobbin. The two assembly lines then have the same number of machine types, namely ten (10). This we believe to be a notable accomplishment since the extended range sensor has four (4) additional piece parts compared to the ADAM sensor.

One other factor in determining a machine line to meet the capability requirement involved the Bobbin Winder. Since the Extended Tripline uses more than twice the amount of line compared to the ADAM sensor, the quantity of Bobbin Winding machines required is doubled. Honeywell has developed a new high speed winding station on this contract which reduces the quantity of Bobbin winders required. Initially, it was determined that five (5) of these

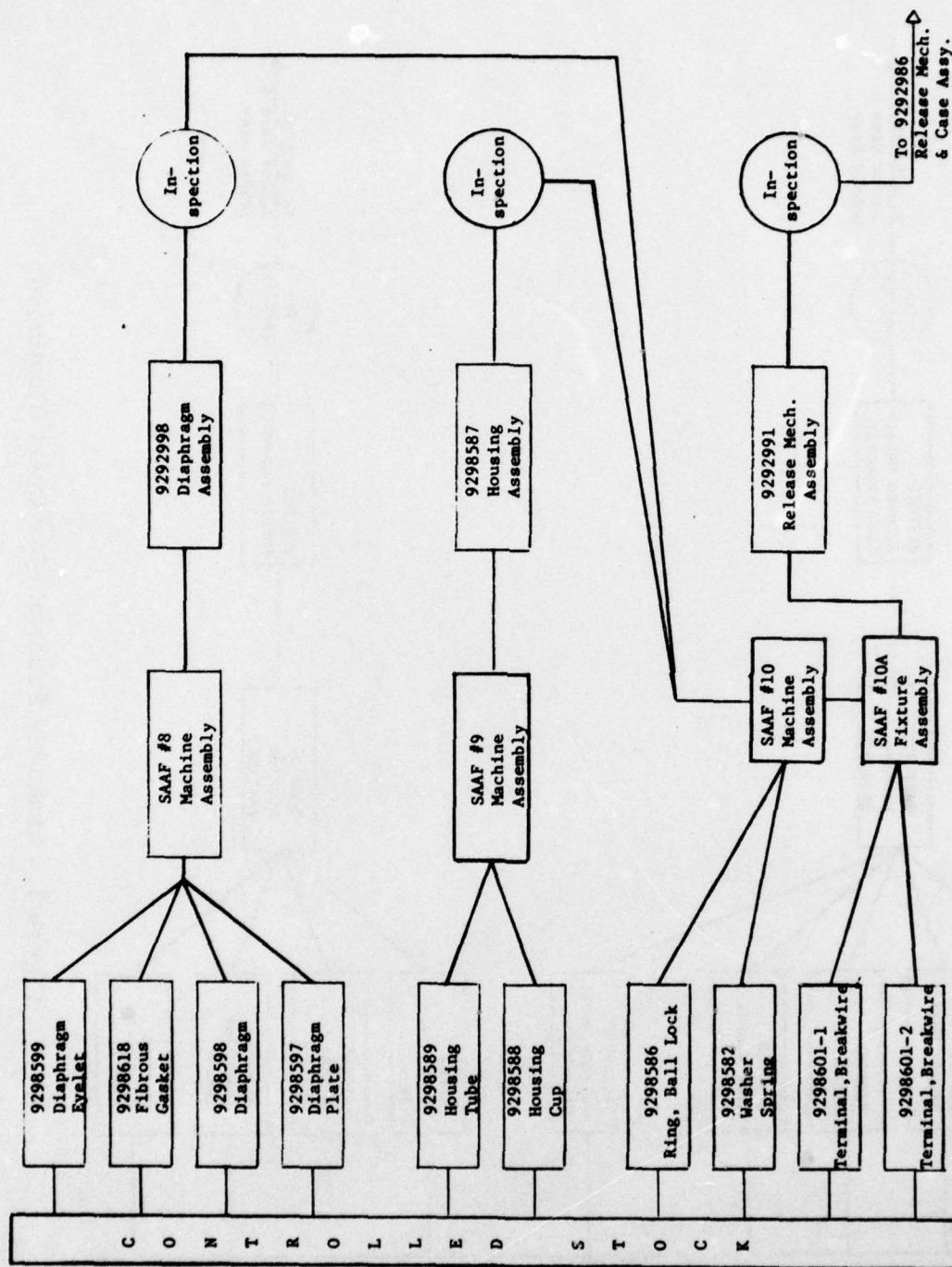


Figure 1. Assembly Process Flow Chart

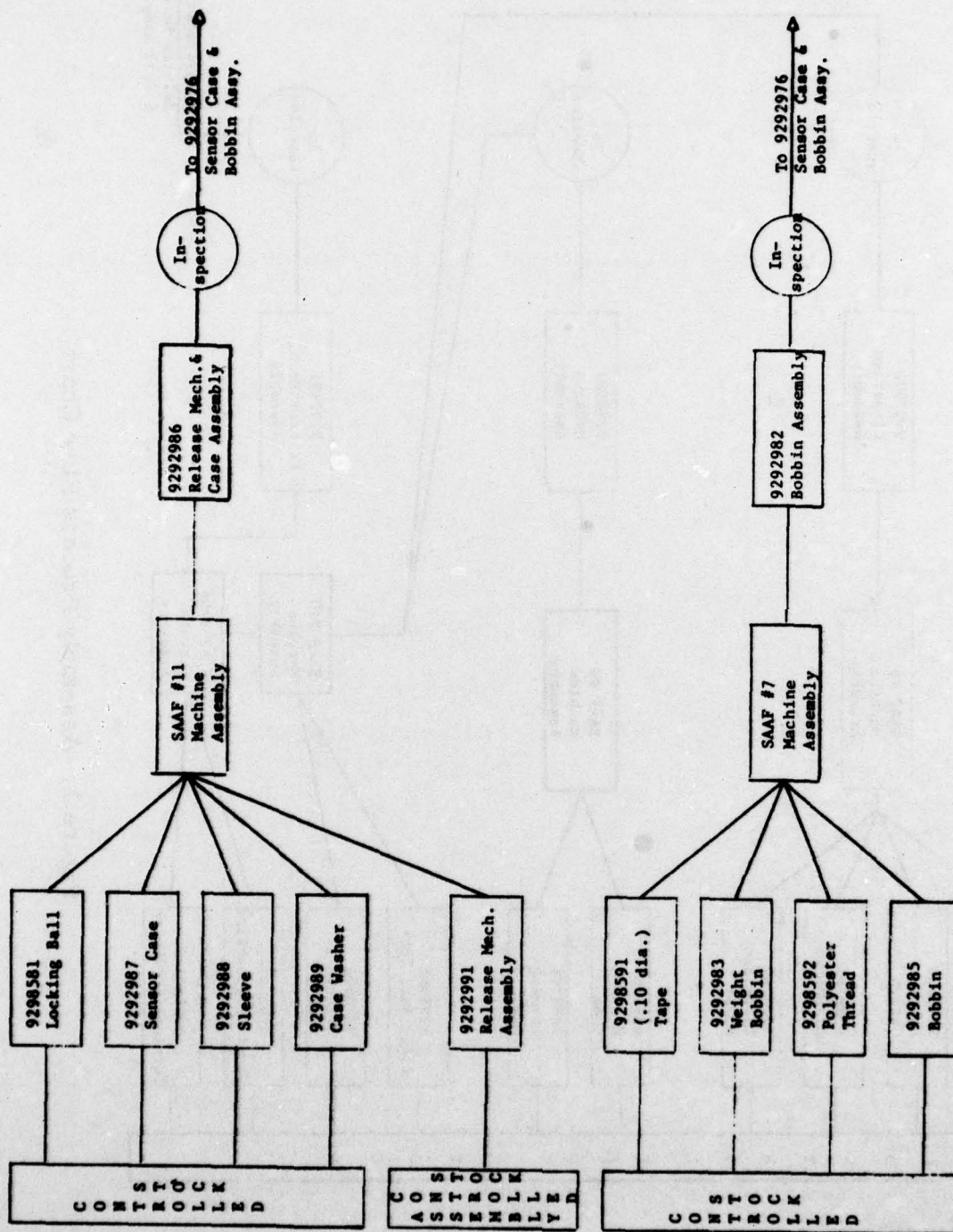


Figure 1. Assembly Process Flow Chart (Continued)



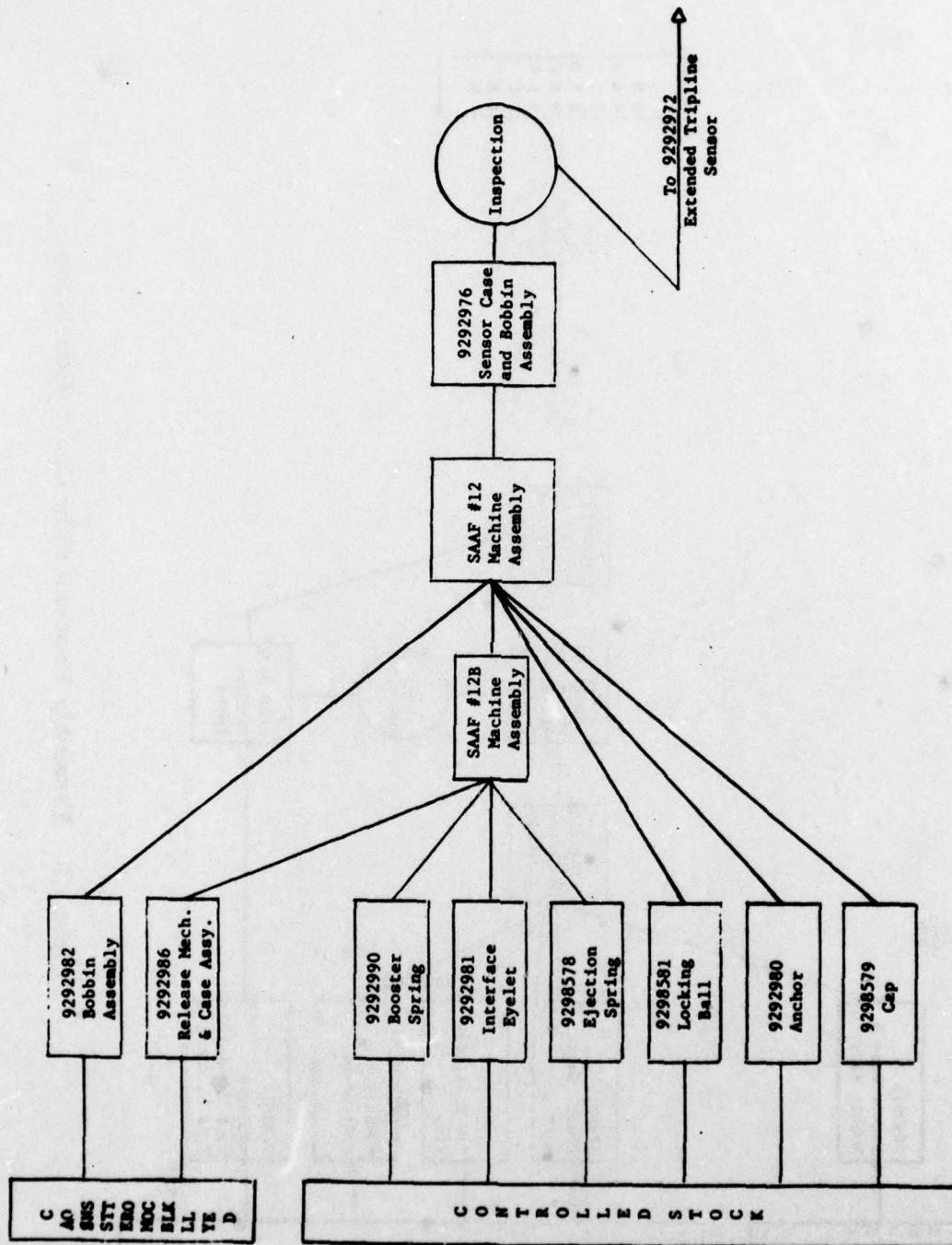


Figure 1. Assembly Process Flow Chart (Continued)

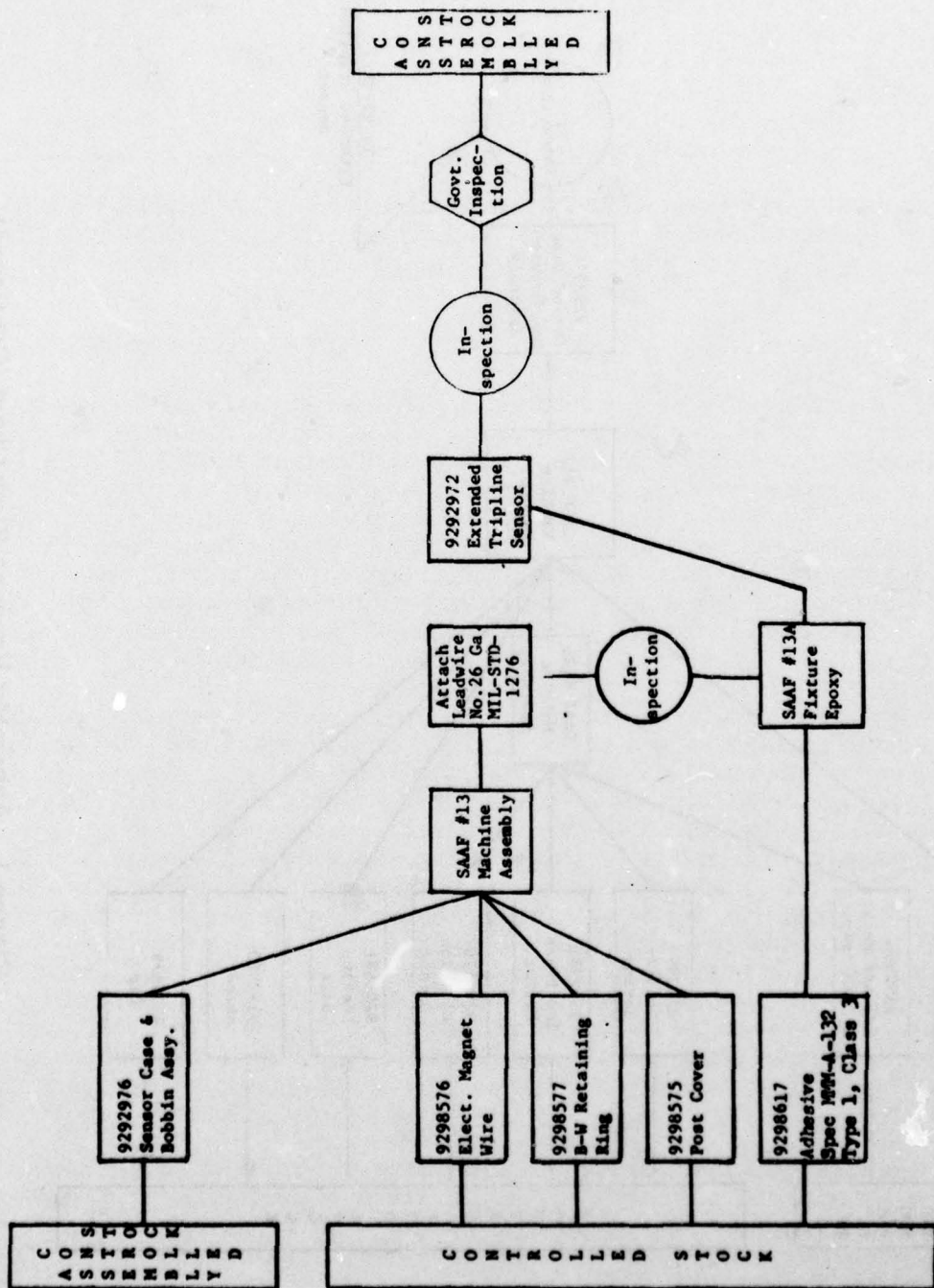


Figure 1. Assembly Process Flow Chart (Concluded)

machines would be required for the 200,000 per month capability. The improved winding station reduces this requirement to four (4) machines.

#### C. ENGINEERING SUPPORT

Throughout the program, reliability and safety inputs were provided to Production Engineering on a continuing basis for the PEP tasks on the Extended Tripline Sensor. Support was also provided for the Machine Development effort in the area of reliability, availability and maintainability (RAM) plus safety and human factors engineering (HFE). This resulted in optimization of the production line within the context of these speciality disciplines.

#### D. INSPECTION AND TEST EQUIPMENT

A listing of inspection and test equipment is presented in Section IV. D of this report. In addition to these specific items, the assembly machines are built to include probes and/or inspection stations to assure that the product meets all quality requirements. The probe stations are designated on the machine dial schematics contained in Equipment Technical Data Package Specification in Appendix B of this report.

#### E. COST DATA

The estimated unit product cost (UPC) of the initial baseline Extended Range Tripline Sensor was \$4.51 each. The UPC for the PEP version of the sensor is \$3.03 each. This shows a potential savings of \$1.48 per unit or \$3,552,000 for a calculated second year production quantity of 2,400,000 sensors. This savings, in reality, could be greater than the figure given. One assumption taken for the initial estimate has proved to be premature. This concerns the requirement for leadwire attachment. At the beginning of the program it was assumed that the leadwires (2) would be soldered to the sensor using an automatic assembly machine. This does not appear possible without additional machine development. Design changes in the sensor configuration will not alleviate the problem. The leadwires must be soldered to the sensor as a separate operation and this cost element appears in the post PEP UPC determination. Eventual automation can increase the savings to \$1.72 for a total price of \$2.79 each per unit. Tooling can be provided to improve the operation as a semi-automatic process. Putting such tooling on an automatic assembly machine does not appear feasible for initial production at this time. Continued study would be conducted during the production phase in order to develop a suitable method of doing this with automation. A concept drawing for a leadwire assembly station is included in Section IV of this report. (Ref: Figure 2). A summary of cost data is contained in Section VIII and Appendix D of this report.



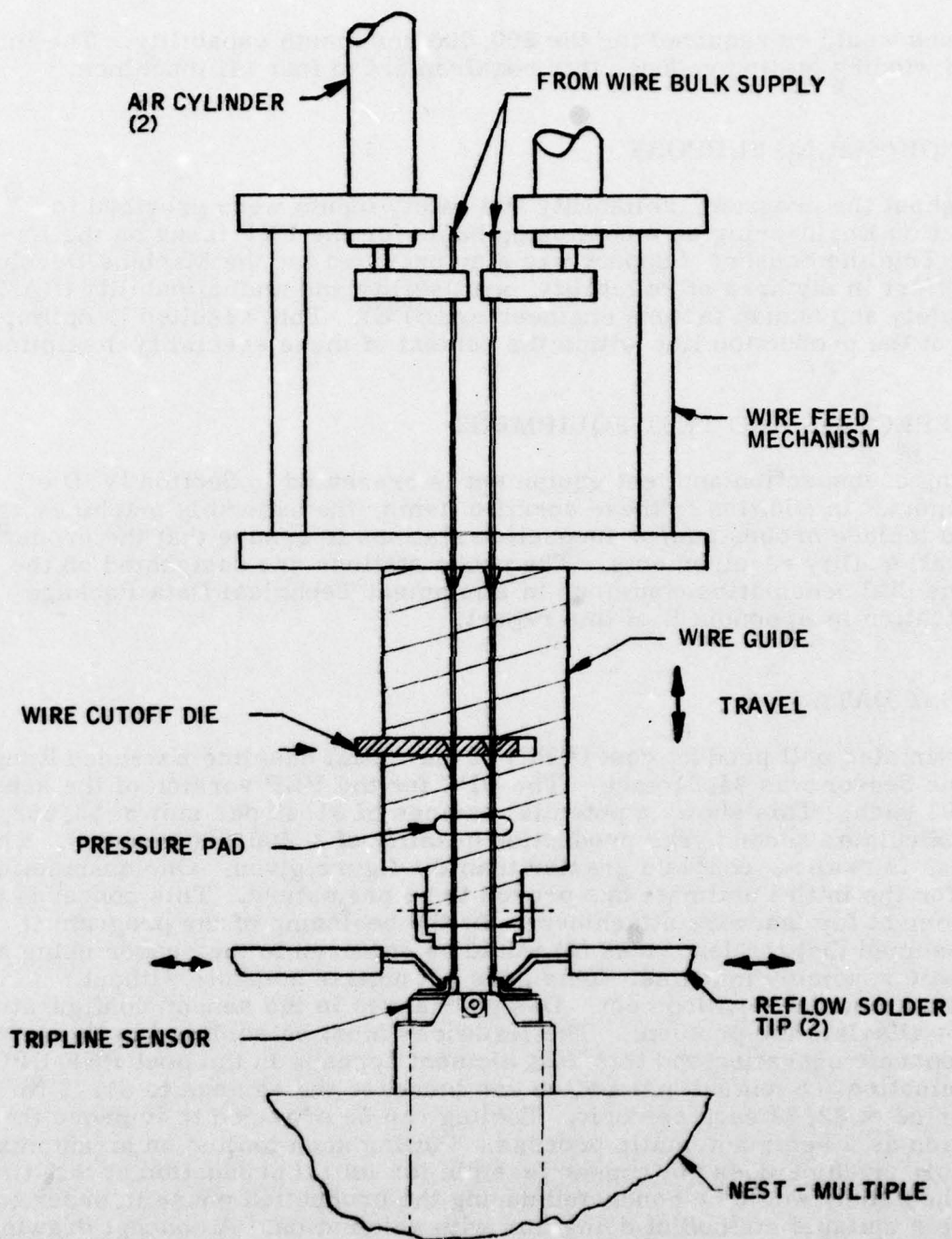


Figure 2. Leadwire Assembly Station

#### F. RECOMMENDATIONS

The completion of this program has provided fulfillment of specified objectives needed to describe a facility and process for an automatic assembly line to perform efficient, high volume production of reliable Extended Range Tripline Sensors which meet the quality requirements at a high confidence level and at a low cost. It is therefore recommended that the machine designs and technical documentation composed in this program be established as the initial production capability baseline design for the XM74 GEMSS Extended Range Tripline Sensor and the proposed machine line.

### III. PRODUCTION ENGINEERING EFFORT

The objective of the Production Engineering effort was, to evolve a design of the GEMSS sensor suitable for economical high-volume production and compatible with safety and reliability requirements established for the parent mine. An additional objective was to achieve maximum commonality with the ADAM sensor since this would contribute to the optimum utilization of facilities to the maximum extent possible for both sensors. The GEMSS sensor Production Engineering effort was conducted in close coordination with the machine line and test equipment design task to ensure complete integration of the production line on a continuing basis.

The result of the Production Engineering effort was a number of design changes which enhanced the producibility/commonality of the sensor configuration. In the following discussion, the initial baseline design is described first in order to establish the proper setting from which the design changes, which are subsequently reported, can be clearly and easily related.

#### A. DESCRIPTION, INITIAL BASELINE DESIGN

The design configuration for the XM74 GEMSS Extended Tripline Sensor adopted as a baseline for the PEP program is delineated in the ARRADCOM Engineering Release Record, DRDAR-LCU-DM, No. A7N2501 dated 24 June 1977. This list has been transcribed and presented in this report as Table 2. The drawings of piece parts which have been recommended for PEP revision are indicated in the right hand column.

#### B. PRODUCIBILITY/COMMONALITY CHANGES

The beginning task in the PEP program consisted of an Engineering study of drawings and specifications in order to determine the changes needed to achieve automatic assembly for high volume, mass production at the lowest possible cost. The sensor subassemblies and final assembly were each evaluated in relation to the recommended changes so as to avoid any compromise in the safety and reliability of the item. A summary of the proposed changes is shown in Table 1. None of these revisions have any affect upon the commonality already existing between the ADAM Sensor and the GEMSS Sensor.

The scope and results of each of the listed changes recommended are described in the following paragraphs:



TABLE 2. BASELINE DRAWING LIST

Drawing No.	Revision Date	Nomenclature	PEP Revision
A9298576	- 77-6-24	Wire, Magnet	
A9298592	- 77-6-24	Thread, Polyester	
B9292983	B 77-6-24	Weight, Bobbin	Yes
B9298591	- 77-6-24	Tape	
B9298617	- 77-6-24	Urethane Compound	
B9298619	- 77-6-24	Paper, Gasket	
C9292981	B 77-6-24	Eyelet, Interface	Yes
C9292989	B 77-6-24	Washer, Case	Yes
C9292990	B 77-6-24	Spring, Booster	
C9292998	B 77-6-24	Diaphragm Assembly	
C9298575	- 77-6-24	Cover, Post	
C9298577	- 77-6-24	Ring, Retaining	
C9298581	- 77-6-24	Ball, Locking	
C9298582	- 77-6-24	Spring, Washer	
C9298587	- 77-6-24	Housing Assembly	
C9298589	- 77-6-24	Tube, Housing	
C9298599	- 77-6-24	Eyelet, Diaphragm	
C9298618	- 77-6-24	Gasket, Fibrous	
D9292972	B 77-6-24	Sensor, Assembly	Yes
D9292976	B 77-6-24	Sensor Case & Bobbin	Yes
D9292980	B 77-6-24	Anchor	
D9292982	B 77-6-24	Bobbin Assembly	
D9292986	B 77-6-24	Release Mech. & Case	Yes
D9292987	B 77-6-24	Case	Yes
D9292988	B 77-6-24	Sleeve	Yes
D9298578	- 77-6-24	Spring, Ejection	Yes
D9298579	- 77-6-24	Cap	
D9298586	- 77-6-24	Ring, Ball Lock	
D9298588	- 77-6-24	Cup, Housing	
D9298598	- 77-6-24	Diaphragm	
D9298601	- 77-6-24	Terminal, Breakwire	
F9292985	B 77-6-24	Bobbin	Yes
F9292991	B 77-6-24	Release Mech. Assy	
F9298597	- 77-6-24	Plate, Diaphragm	

### Change Proposal No. 1

The first proposal includes the following parts and assembly:

9292983	Weight, Bobbin
9292985	Bobbin
9292982	Bobbin Assembly

The changes requested are needed to facilitate the press fit assembly of the Bobbin Weight into the Bobbin. This is accomplished by providing a taper in the Bobbin Bore and adding a radius at the ends of the weight. The benefit of these changes will be experienced as a greater yield from the assembly machine since these revisions will reduce misfeeds and shearing during the press fit assembly.

This assembly is accomplished on Machine No. 7.

The Bobbin also requires that the 0.03 slot be relocated from the 4th quadrant to the 1st quadrant in order to permit the thread to lead off in the proper direction on the assembly machine.

### Change Proposal No. 2

The following piece parts are the subject of the second revision:

9292987	Case, Sensor
9292988	Sleeve
9292989	Washer Case

The Case and Sleeve are modified to include two sets of four holes each. These are construction holes which permit the use of pins which serve to retain the ejection spring and booster spring during an intermediate stage of assembly. The springs are assembled on Machine No. 12B and the remaining parts for this subassembly are added on Machine No. 12. The retaining pins hold the springs in place for transfer from 12B to 12.

The case is further modified to include four (4) flutes to act as guides during assembly of the Ejection Spring and Booster Spring. Another feature which will allow easier assembly and seating of the Booster Spring is a larger chamfer at the end of the 0.415 diameter of the case washer.

### Change Proposal No. 3

Changes were requested on two more parts to reduce assembly difficulties anticipated on Assembly Machine No. 12B. These are:

9298578 Spring, Ejection  
9292981 Eyelet, Interface

The Interface Eyelet is currently a brass eyelet. To assemble this eyelet requires an orientation for proper side up and also a pre-assembly with the Ejection Spring prior to assembly into the Sensor Case. A simple flat washer overcomes both of these problems without any adverse affects upon function. The ejection spring could be correspondingly simplified by not having to engage the eyelet, but a change in the spring would eliminate the commonality with the ADAM spring.

These parts are assembled on Machine No. 12B.

#### Change Proposal No. 4

Two assembly drawings are to be revised in order to show changes on the individual piece parts contained in these assemblies. These are:

9292986 Release Mechanism and Case Assembly  
9292972 Sensor Assembly

#### C. DESCRIPTION, FINAL DESIGN

The final design as represented by the incorporation of the previously described changes is functionally identical to the pre-PEP configuration. The flutes added to the case must be limited in length so as not to interfere with the sealing and potting to be accomplished in the mine body. The use of a narrow band of shrink tubing on the outside of the Sensor Body will improve the results of the sealing and potting operation.

Certain changes are being reviewed for incorporation into the ADAM Sensor as a result of VECF activity. These changes will affect the Release Mechanism which is common to both the Extended Tripline and the ADAM Tripline. These changes do not appear to be objectionable for application to the Extended Tripline, however, no space has been given to further discussion of these changes in this report.

#### D. TECHNICAL DATA PACKAGE

The changes described above apply specifically to assembly operations. Machine development and prove-out of concepts have been accomplished on the basis of these recommendations. Additional problems may occur, however, these will not relate to the Sensor design with the proposed changes.



Individual piece parts are normally procured from subcontract sources and no changes have been suggested which pertain to piece part fabrication. The procurement history of 15,000 sets of parts manufactured and purchased for contract DAAA21-75-C-109 and 5,500 sets for AOMC Purchase Order D-428994-OPD did not reflect any significant fabrication problems which require drawing changes.

Quality Assurance provisions for Tripline Sensor Extended Range, QAP-GEMSS-1 (23 June 1977) has been studied and reviewed. This specification is acceptable with the following exceptions:

Reference paragraph 3.3 Non-Release

"The Sensor Assembly, when subjected to a deceleration of 5000 g's minimum, in the direction of deployment, shall not release when tested in accordance with 4.5.1.1". At the present time it is not known if the GEMSS Sensor is capable of meeting this requirement. No qualification testing has been performed as described in paragraph 4.5.1.1 at the 5000 g level. Until such a time when this is accomplished, this requirement cannot be accepted.

3.6 & 4.5.1.4

Breakwire Function

The Pull Force Rate,  $3-1/4 \pm 3/4$  inches/second (ref. 4.5.1.4) should be deleted and replaced with the following:

"A Pull Force shall be applied smoothly and uniformly along the axis of the bobbin post and the force required for the breakwire function shall be noted."

#### IV. MACHINE LINE AND TEST EQUIPMENT

##### A. MACHINE LINE - REQUIREMENTS AND PROGRAM APPROACH

The producibility engineering and planning activities conducted on this program required the design of a machine line for the automatic assembly of the XM74 GEMSS Extended Tripline Sensor capable of producing 200,000 sensors per month, (1:8:5 schedule using a 50 minute hour and 21 days per month). The assemblies must reflect a low unit product cost and meet high reliability and safety requirements. This called for maximum utilization of contemporary technology in automated assembly machine and test equipment design to ensure maximum economy in GEMSS Extended Tripline production operations. Complete design packages have been provided for a total of 10 different machines in the following three categories:

1. Machines common to ADAM automatic assembly machines:

Machine No. 8	Diaphragm Assembly
Machine No. 9	Housing Assembly
Machine No. 10	Release Mechanism Assembly
Machine No. 10A	Terminal Assembly

2. Machines similar to ADAM Sensor machines and differing only by minor alterations:

Machine No. 11	Release Mechanism and Case Assembly
Machine No. 12	Sensor Case and Bobbin Assembly
Machine No. 13	Sensor and Breakwire Assembly
Machine No. 13A	Epoxy Dispenser

3. Machines which are new and unique to the GEMSS Extended Tripline Sensor:

Machine No. 7	Bobbin Assembly
Machine No. 12B	Sensor Case and Spring Assembly

The program approach selected was as follows:

- Conduct research into current machine technology to determine the most appropriate applications and use the results to establish the machine line.
- Prepare performance specifications for the machines selected.



- Prepare and document all designs and present them with associated costs to the Project Office.

As a result of this effort, the Project Office would receive, in effect, a machine line technical data package suitable for the Sensor initial production facility. The documentation requirements for the effort include a process description and inspection procedures.

In addition to assembly machines, certain other machines and equipment are needed for operation of the line. These items are as follows:

- (1) Spring Coiler to fabricate the Ejection Spring,
- (1) Spring Coiler to fabricate the Booster Spring.

DEGREASER: Barin Blakeslee  
Model MSR-120  
460 Volts  
1/60 Phase/cycle

CONTOUR PROJECTOR: Kodak  
Model 4  
50-60 Cycle

DEMAGNETIZER: El Matco  
Model LAC-15VB  
460 Volts  
50/60 Cycle

HUMIDIFIER AND FENT: Not defined

OVEN: Despatch  
Model V-35 HD  
440 Volts  
650°F Temp  
60 Cycle

MICROSCOPE: American Optical Comp.  
Model 570 0.7 to 4.2X

Detailed descriptions and requirements for these machines and equipment is included in the Appendix Section of this report. This information has been prepared in accordance with Appendix B and Appendix C of specification PBM-OSM-70-1 as required in the revised scope of work for this contract.

## B. DESIGN APPROACH

The assembly that is to be automated for high volume production is generally a tested and proven design either in the advanced development or production engineering stage. Because of this, Honeywell's approach in production engineering the existing product design for producibility is to minimize design changes and to emphasize modular automation concepts. The proven reliability, safety, quality, and cost reduction benefits, inherent in the automated assembly concepts developed by and in use at Honeywell, are realized when the considerations for automated assembly are incorporated into the product design.

### Modification of Product Design for Automated Assembly

Design modifications that are necessary to facilitate automated assembly generally fall into the following three categories:

- Changes in Configuration to Facilitate Automated Part Orientation and/or Assembly Location - Many of the parts for an assembly have slots, holes, projections, irregular shapes, etc., that can be used by the automated assembly mechanisms to provide proper part orientation and location. Parts that do not have "identification" features can usually be modified slightly to provide the necessary configuration without affecting the basic device design or function.
- Changes in Fastener and Retaining Members to Permit Handling of Subassemblies from One Machine to the Next - The output of each machine must be secured as a module or assembly to permit handling and feeding to subsequent machines. Parts generally can be modified slightly to provide fastening or retaining means without affecting the basic product design or end function.
- Changes in Subassembly Drawings Reflecting New Assembly Sequences - Initially, the product must be broken down into logical subassemblies or modules for automatic assembly. To facilitate assembly requirements, inprocess checks and critical inspection techniques, modules are further broken down into machine operations. There is an upper limit to the number of operations that can be economically performed by any one automatic assembly machine. An optimum number of work stations for each machine can be determined by considering balance between operations on the machine, machine complexity and production capacity. Parts can usually be changed slightly to facilitate new assembly sequences without affecting the function of the end device in any manner.



### Development of Assembly Sequence

To develop efficient and reliable assembly machines, the overall device must be considered in relation to its makeup and function. The first step for efficient and reliable utilization of automation is to establish an automated assembly sequence for the device based on a modular relationship with preliminary assembly and inspection operations for each module. As development progresses, the assembly operations for the machines are defined and subsequently refined to provide shortcuts and continuity. Also, features may be added to improve quality control.

The modules or assemblies resulting from the machine operations must be considered to determine the means of interim handling and transportation to the next operation. Minor subassemblies without delicate features can be handled in bulk from one machine to the hopper feeder of the next machine. Other modules require special magazines for protection as well as for maintaining orientation to the next feeding station. The design of the magazines and any special safety and handling provisions they must include are also considered throughout the assembly sequence development.

As the modules develop, their characteristics are defined in terms of assembly, function, and safety requirements. Provisions for performing safety and quality checks as part of the machine operations are incorporated wherever possible. The operations of the checking stations are reviewed to assure that all requirements are met. Redundant inspection stations are provided for all critical checks. On occasions where machine functional checks are impractical, verification provisions for inspector monitoring or sampling are made. These considerations are part of the key to efficient and reliable module machine development.

### C. SPECIAL AUTOMATIC ASSEMBLY FIXTURES (SAAFS)

The assembly machines proposed for fabrication of the XM74 GEMSS Extended Range Tripline Sensor and previously referred to in Section IV.A are listed in Table 3 to show the parts assembled on each machine.

In the Summary, Section II.E, describing Cost Data, reference is made to an assembly problem relating to leadwire fastening. It is recommended that this operation be accomplished using special purpose tooling during the initial production phase of sensor manufacture. Honeywell recommends production experience in order to optimize the design of a station suitable for installation on an assembly machine. Figure 2 is a concept sketch of a bench mounted tool which would be manually loaded and cycled. This will permit refinement of the concept without interrupting machine operation.

**TABLE 3. XM74 GEMSS EXTENDED RANGE TRIPLINE  
SENSOR ASSEMBLY MACHINES**

Machine Number	Machine Name	Parts Assembled		Commonality
7	Bobbin Assembly 9292982	9292983 9292985 9298591 9298592	Weight Bobbin Tape Thread	New
8	Diaphragm Assembly 9292998	9298597 9298598 9298599 9298618	Diaphragm Plate Diaphragm Diaph. Eyelet Fibrous Gasket	Common
9	Housing Assembly 9298587	9298588 9298589	Housing, Cup Housing, Tube	Common
10	Release Mech. Assy. 9292991 Less Terminals	9298582 9298586 9292998 9298587	Washer Spring Ring, Ball Lock Diaphragm Assy. Housing Assy.	Common
10A	Terminal Assy. 9292991 (Release Mech. Assy.)	9298601-1 9298601-2	Terminal, Breakwire Terminal, Breakwire	Common
11	Sensor Case/Sleeve Assembly 9292986	9292987 9292988 9292989 9298581 9292991	Sensor Case Sleeve Case, Washer Locking Ball Release Mech. Assy.	Modified
12B	Sensor Case/Spring Assembly	9292981 9292990 9298578 9292986	Interface Eyelet Booster Spring Ejection Spring Case/Sleeve Assy.	New
12	Sensor Case/Bobbin Assembly 9292976	9292980 9298579 9298581 9292986 9292982	Anchor Cap Locking Ball Release Mech./Case Assy. Bobbin Assembly	Modified
13	Sensor & B/W Assy.	9298575 9298576 9298577 9292976	Post, Cover Magnet Wire B/W Retaining Ring Sensor Case/Bobbin Assy.	Modified
13A	Epoxy Dispenser 9292972 Extended Range Tripline Sensor	9298617	Adhesive Spec. MMM-A-132 Type 1, Class 3	Modified



### Tripline-Functional Outline of Leadwire Assembly Station

Assembly of the leadwires to the release mechanism terminals can be accomplished by equipment outlined in Figure 2.

#### Sequence of functions:

1. Feed mechanism advances wires 0.050 inch.
2. Form die anvil actuates and bends leadwires at a 90° angle.
3. Wire feed mechanism advances leadwires one-half inch (0.50).
4. Head moves down until leadwires touch terminals on release mechanism assembly.
5. Reflow solder tips move into position and reflow solder on terminals. (If insufficient solder is available on terminals a form of paste solder may have to be added in a prior step).
6. Cut off die cuts wire to predetermined length.
7. Wire guide head moves up, thus completing assembly.
8. Nest rotates, moving new part into position for sequence of operation to begin again.

A similar situation exists with the Diaphragm Assembly machine with respect to notching of the diaphragm. The first approach to this operation was to notch the diaphragm as it was being fed at the assembly station on Machine No. 8. Later study of this plan revealed too many problems, among them being orientation of the diaphragm. It is now proposed to accomplish the notching when the assembly is complete. This is the method in current use on the hand line. We have found that the hand tool built for this purpose could, in fact, be mechanized and added as the final working station on Machine No. 8. This machine could still be a "common" machine except when running ADAM Sensors the notching station would be disengaged.

Funding limitations did not permit design effort to delineate this station during the PEP program. It is suggested that this task be accomplished on the IPF program.

#### D. INSPECTION AND TEST EQUIPMENT

The following items of inspection and test equipment are required in the manufacture and assembly of the GEMSS Extended Tripline Sensor:

<u>EQUIPMENT NAME</u>	<u>DRAWING NO.</u>	<u>REVISION</u>
Deployment Fixture	H28112266-E2	B
Force and Displacement Gage	H28112270-G1	A
0.460 Dia. Go-Ring Gage	H28112271-G1	A
Spring Test Nest	H28112271-G2	A
Functional Location Gage	H28112281-G1	A
0.460 Dia. Go-Ring Gage	H28112284-G1	A
Impact Tester	No Drawing - Purchase	

## V. RAM/SAFETY/HFE

The following paragraphs summarize the significant aspects of activities involving Reliability, Availability and Maintainability (RAM), Safety, and Human Factors Engineering (HFE) support during the program. Table 4 shows the associated data items completed as specified in the program.

TABLE 4. ASSOCIATED DATA ITEMS

TASKS	DATA ITEM	J	F	M	A	M	J	J	A
RAM PROGRAM PLAN	A011				▲				
HFE PROGRAM PLAN	A015				▲				
CONCEPT/DESIGN REVIEWS				CONTINUOUS AS NECESSARY					
RAM PROGRAM REPORT	A007, A010 A012 (1)						▲		
HFE TEST PLAN	A016					▲			
HFE FINAL REPORT	A018, A017 (2)								▲
SAFETY REPORTS	A020					▲	▲	▲	
RAM/SAFETY/HFE INPUTS FOR FINAL REPORT	A004								▲

- (1) The RAM Report incorporated the following three program required Data Items: the RAM allocations, assessments and analysis, the Maintainability Reports, and the Maintainability Prediction Data.
- (2) The HFE Report incorporated the HFE progress report and final HFE report.



## A. RELIABILITY, AVAILABILITY AND MAINTAINABILITY (RAM)

### 1. Reliability

The primary Reliability Engineering effort was to insure that the machine line would achieve the Reliability, mean time between failures (MTBF), requirements as defined in the contract. The activities pursued to accomplish this effort are presented in the Reliability-Availability-Maintainability Program Plan which was written in response to Item A001 of the DD Form 1423 CDRL.

There was continuous participation in concept revision and changes to ensure awareness and consideration of Reliability criteria. All machine designs and specifications were reviewed for operational reliability. There was also participation in the preparation of machine performance specifications to ensure consideration of reliability requirements.

The Reliability-Availability-Maintainability Report was presented in response to Item A007, A010, and A012 of DD Form 1423 CDRL and includes the reliability (MTBF) analysis of all final machine designs. The analysis presents the predicted MTBF values for each type of machine in XM74 Tripline Sensor Machine Line. The prediction is based on data and prediction methods recently developed at Honeywell and currently used on the ADAM Mine Machine Contract, DAAK10-77-C-0018. All machines have surpassed the predicted MTBF's beyond the contract requirement which is a MTBF = 7 hours, the design goal.

### 2. Availability

The primary objective was to ensure that the machine line would achieve the availability requirements defined in the contract as 85 percent under ideal conditions, i. e., trained personnel, repair parts and tools readily available, etc. The contract further defined a production rate of 200,000 units/month on a 1-8-5 shift bases using a 50 minute hour and 21 days per month. These were the assumptions used in the availability analysis of the machine line.

Availability activities that were followed in this program are presented in the Reliability-Availability-Maintainability Program Plan. Machine concepts, including station descriptions and dial layouts, were reviewed for availability considerations. There was also continuous participation in all concept reviews and changes to ensure awareness and consideration of availability criteria.



Predictions were made of each machine's availability using data and prediction methods recently developed at Honeywell and currently used on the ADAM Mine Machine Contract DAAK10-77-C-0018. All machines had a predicted availability as good or better than the contract requirement. The result of the availability analysis and predictions were presented in the Reliability-Availability-Maintainability Report.

### 3. Maintainability

The primary objective was to ensure that the machine line would achieve the maintainability, mean time to repair (MTTR), requirements as defined in the contract. The maintainability efforts were closely associated with the reliability and availability activities discussed in the previous paragraphs. The tasks that were followed to reach this objective are presented in the Reliability-Availability-Maintainability Program Plan. This included continuous participation in concept reviews and changes to ensure awareness and consideration of Maintainability criteria. All machine designs and specifications were reviewed for maintainability requirements.

The Reliability-Availability-Maintainability Report was presented in response to Items A007, A010 and A012 of the DD Form 1423 CDRL and includes the maintainability (MTTR) analysis of the final machine designs. The analysis develops predicted MTTR values for each type of machine used in the XM74 Tripline Sensor Machine Line. These values were presented in the RAM Program Report. The prediction is based on data and prediction methods recently developed at Honeywell and currently used on the ADAM Mine Machine Contract DAAK10-77-C-0018.

Other maintainability activities during the program were developing machine guidelines for the performance specifications, monitoring design activities, reporting concerns and progress, maintaining liaison with other program personnel and reviewing plans for future program plans.

### B. SAFETY

The Safety Engineering activities were closely coordinated with the Human Factors Engineering activities. These safety activities primary involved concern for safety of personnel associated with the automated equipment i. e., operators, maintenance personnel, observers, etc. The safety effort in support of machine design activities encompassed initial reviews of the requirements, definition of guidelines for use in design and incorporation of safety criteria in the specifications including requirements established by the Occupational Safety and Health Administration (OSHA).

A Safety Analysis and Hazardous Failure Mode Analysis was performed for each machine and documented in Safety Analysis and Hazard Evaluation Reports in response to Item A020 of the DD Form 1423 CDRL. This analysis included a review of the information generated for a similar data item under the ADAM Mine Machine Contract DAAK10-77-C-0018 with modifications thereto based on equipment design features peculiar to the GEMSS Extended Tripline Sensor. A separate report was written for each type of machine used in the proposed production line.

### C. HUMAN FACTORS ENGINEERING (HFE)

The objective of the HFE effort was to maximize the human factors effectiveness of the machine line and thus ensure high efficiency in all man-machine interactions. The HFE activities were coordinated with the Safety and RAM activities to establish an efficient and effective effort. HFE Machine Design guidelines were developed based on program requirements and objectives, experience with other machine lines, military specifications and standards and sources outside Honeywell relative to HFE evaluation of machine production lines. These guidelines and planned HFE activities were presented in the HFE Program Plan in response to Item A015 of the DD Form 1423 CDRL.

All design and specification activities for the unique and modified machines were monitored to ensure appropriate consideration for human factors. Existing ADAM Mine Machines, contract DAAK10-77-C-0018, used in this program were reviewed for possible HFE improvements. The machine drawings and specifications of the unique and modified machines used in the machine line were reviewed for HFE features and considerations. The HFE analysis is documented in the Human Factors Engineering Report in response to Item A018 and A017 of the DD Form 1423 CDRL.

A Human Factor Test Plan, in response to Item A016 of the DD Form 1423 CDRL, was written. It provides a general outline of test procedures to follow that may be used in a later phase to evaluate the HFE characteristics of the GEMSS XM74 Extended Tripline Sensor production machines.

To summarize, the following data items have been submitted as referenced in this section:

Data Ref. No.	Date Submitted	Report
A004	31 August	Final Report
A007	15 June	TECH/PERF No. 12
A010	15 June	TECH/PERF No. 12
A011	28 April	TECH/PERF No. 10
A012	15 June	TECH/PERF No. 12
A015	28 April	TECH/PERF No. 10
A016	15 June	TECH/PERF No. 12
A017	30 March	TECH/PERF No. 9
A018	31 August	Final Report
A020	15 June	TECH/PERF No. 12



## VI PROCESS DESCRIPTION

### A. SUMMARY

The assembly process flow chart is shown in Figure 1. Copies of the Production Process Summary layouts are included on the pages following. These describe the process for assembly of the XM74 GEMSS Sensor.

In addition, there are certain special process requirements currently in use for the ADAM Sensor which are applicable to the Extended Tripline Sensor. These are as follows:

- Harparizing - The Spring Washer is deburred in a Harparizer machine. This special tumbling is done on all lots to assure uniform, burr free parts for proper function of the Sensor.
- Water Bath and Humidity Chamber - The Bobbin, being made of nylon material is subject to becoming brittle in the winter season due to low humidity in the factory. To correct this problem, the Bobbin piece parts are soaked in water for 8 hours before winding thread on the bobbin core.

The same problem exists when the retaining ring is placed on the bobbin post. This problem is resolved by soaking the Sensor Assemblies in a humidity chamber a minimum of 2 hours prior to machine assembly of the retaining ring.

- X-Ray - The Release Mechanism, and Case Assembly, and the Sensor Case and Bobbin Assembly require Process Controls where X-Raying is employed to determine position and presence of various componentry.

### B. PROCESS SUMMARY LAYOUTS









PC35 PRODUCTION PROCESS SUMMARY										SHEET 1		ISSUE Z	
ONE OPERATION PER SHEET										COPIES		PART NUMBER	
OPERATIONAL { Y = YES, N = NO }										MACHINE #10/10A		9292991-001	
REVISION										8/29/78		54 55 56 57	
OPERATION ITEM (DEVICE)										MACHINE ASSEMBLY		CALLOUT DESCRIPTION	
OPERATION NUMBER										SET-UP		PIECES/HOUR	
DEPT.										OPERATOR		CALLOUT NUMBER	
ISSUE NO.										STANDARD		CALLOUT DESCRIPTION	
NO.										HRS/M		CALLOUT DESCRIPTION	
010A										MACHINE ASSEMBLY DIAPHRAGM AND HOUSING ASSEMBLY		110	
F 20										2		9298582	
F 20										1		9298587	
F 20										1		9292998	
F 26										1		9298586	
010J										PROCESS CONTROL			
014M										SORT PARTS FROM REJECT-EJECT BIN			
014S										INSPECT PER I.P.			
020A										MACHINE ASSEMBLY TERMINAL BW'S			
F 20										1		9298601-001	
F 20										1		9298601-002	
020J										PROCESS CONTROL			
										CONT. NEXT PAGE 9292991			

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P035 PRODUCTION PROCESS SUMMARY  
 ONE OPERATION { Y = YES }  
 PER SHEET { N = NO }  
 SHEET 2 ISSUE Z  
 COPIES PO 1

MACHINE # 10/10A  
 PART NUMBER 9292991-001  
 RELEASE MECH. ASSEMBLY  
 8/29/78  
 32 PLF 35  
 54 55 56 57 58 59 60

OPERATION TITLE  
 CALLOUT DESCRIPTION  
 110  
 111  
 112  
 113  
 114  
 115  
 116  
 117  
 118  
 119  
 120

SORT .157 + .002 DIMENSION  
 024M  
 029M  
 029S

"X" RAY FOR SPRING WASHER POSITION  
 INSPECT PER I.P.

END OR PROCESS

TRANSACTION CODES  
 C1 = COMPLETE CALLOUT DELETE  
 F3 = ADD NEW CALLOUT OR REVISE EXISTING CALLOUT EXCEPT CALLOUT NO.  
 L0 THROUGH L9 = REMARKS  
 C4 = REVISE EXISTING OP TITLE LINE ONLY - EXCEPT OP NO.  
 HP-431 (BOND) HP-431A (2 PART) REV 4/77

TRANSACTION CODES  
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 HP-431 (BOND) HP-431A (2 PART) REV 4/77





















## VII. QUALITY ASSURANCE

Inspection Procedures were generated for each material, piece part, sub-assembly, and final assembly on the Extended Tripline Sensor Drawing List, applying the Quality Assurance provisions developed under Contract DAAA-73-C-0070 and QAP-GEMSS-1, dated 23 June 1977, and are included as Appendix C.

Quality Engineering, in conjunction with the other Honeywell engineering disciplines, developed the necessary inspection equipment designs which were prepared in response to Item A008 of the DD Form 1423 CDRL.

Quality Engineering Acceptance Equipment Operation Instructions and the Calibration Program and Data Record were prepared in response to Items A013 and A014 to the DD Form 1423 CDRL.

There was continuous participation by Quality Engineers with the Machine Designers, Production Engineers, and Reliability Engineers on machine design tasks to assure that the special automatic assembly fixtures (SAAF's) met the requirements and considerations of the Scope of Work. Quality Assurance Demonstration Plans were prepared for each of the ten (10) SAAF's. These were prepared in response to Item A009 of the DD Form 1423 CDRL.

The plans detail Honeywell verification of the SAAF, which includes: identification areas, operation, controls, preliminary run and log book verification. These plans also include the demonstration test, which includes: items addressed preceding the demonstration test, during the test and following the test. Also included with Item A009 is a General Procedure for Demonstration Test that describes the duties of each discipline during the conduct of a demonstration test. The General Procedure also addresses special situations, such as administrative stoppage, a RAM failure, and termination of test for unsatisfactory SAAF performance.

## VIII. CONCLUSIONS AND RECOMMENDATIONS

The XM74 GEMSS Extended Range Tripline Producibility Engineering and Planning Program has been successful in reducing the unit product cost of the GEMSS Sensor by incorporating changes to permit automation. The degree of automation can be extended as experience is gained in actual production as in the case of leadwire soldering. It is believed that this improvement must be done in a production environment in order to properly identify and establish the working parameters for a successful soldering station.

The PEP program has provided complete designs for ten (10) SAAF's for automatic assembly of the GEMSS Sensor. Machine concepts for new stations were built in the lab and proved out for reliability and establish a high confidence level in the design.

The PEP program has provided a complete Quality package conveying all aspects of machine build and acceptance as well as quality data for sensor production.

The PEP program has provided coordinated inputs in the area of RAM, Human Factors Engineering, and Safety for the Extended Tripline facility. This facility has been fully delineated including the factory floor plan and necessary support equipment.

The production facility costs are also presented in detail in Appendix D which includes separate cost information for debug and machine demonstration hardware.

Based upon these accomplishments it is recommended that this information be used for defining the Initial Production Facility for GEMSS Sensor production.



FS 578-4869

APPENDIX A  
EQUIPMENT FUNCTIONAL CRITERIA SPECIFICATION  
FOR  
SPECIAL AUTOMATIC ASSEMBLY LINE  
FOR XM74 GEMSS EXTENDED TRIPLINE SENSOR

SPECIFICATION FS 578-4869  
ORIGINAL ISSUE

DARCOM Project  
578-4869

AUGUST 1978

Preparing Organization  
Honeywell Inc.  
Defense Systems Division  
94580

**EQUIPMENT FUNCTIONAL CRITERIA SPECIFICATION  
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EQUIPMENT FUNCTIONAL CRITERIA SPECIFICATION  
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## 1.0 SCOPE

This equipment functional criteria specification establishes the technical and mission requirements, allocates functional area requirements for the process, identifies interfaces and specifies individual pieces of equipment which are to be configuration items for special automatic assembly of the XM74 GEMSS Extended Tripline Sensor.

This equipment is tabulated (ref. Table A-1) which shows each machine by name and number, a listing of parts assembled and commonality with ADAM Sensor Assembly Machines.

## 2.0 APPLICABLE DOCUMENTS

### 2.1 Government Documents

The following documents of the issue in effect on the date of beginning activity on this contract shall form a part of this specification to the extent specified herein:

#### Standards

MIL-STD-1472	Human Engineering Design Criteria for Military Systems, Equipment, and Facilities
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#### Drawings

9292972	Extended Range Tripline Sensor
QAP-GEMSS-1	Specification Extended Range Tripline Sensor

TABLE A-1. GEMSS EXTENDED RANGE TRIPLINE  
SENSOR ASSEMBLY MACHINES

Machine Number	Machine Name	Parts Assembled		Commonality
7	Bobbin Assembly 9292982	9292983 9292985 9298591 9298592	Weight Bobbin Tape Thread	New
8	Diaphragm Assembly 9292998	9298597 9298598 9298599 9298618	Diaphragm Plate Diaphragm Diaph. Eyelet Fibrous Gasket	Common
9	Housing Assembly 9298587	9298588 9298589	Housing, Cup Housing, Tube	Common
10	Release Mech. Assy. 9292991 Less Terminals	9298582 9298586 9292998 9298587	Washer Spring Ring, Ball Lock Diaphragm Assy. Housing Assy.	Common
10A	Terminal Assy. 9292991 (Release Mech. Assy.)	9298601-1 9298601-2	Terminal, Breakwire Terminal, Breakwire	Common
11	Release Mech. and Case Assembly 9292986	9292987 9292988 9292989 9292581 9292991	Sensor Case Sleeve Case, Washer Locking Ball Release Mech. Assy.	Modified
12B	Release Mech. and Case Assembly 9292986 (Partial)	9292981 9292990 9298578 9292986	Interface Eyelet Booster Spring Ejection Spring Case/Sleeve Assy.	New
12	Sensor Case/Bobbin Assembly 9292976	9292980 9298579 9298581 9292986 9292982	Anchor Cap Locking Ball Release Mech./Case Assy. Bobbin Assembly	Modified
13	Sensor & B/W Assy.	9298575 9298576 9298577 9292976	Post, Cover Magnet Wire B/W Retaining Ring Sensor Case/Bobbin Assy.	Modified
13A	Epoxy Dispenser 9292972 Extended Range Tripline Sensor		Adhesive Spec. MMV-A-132 Type 1, Class 3	Modified

## 2.2 Non-Government Documents

Not applicable.

## 3.0 REQUIREMENTS

### 3.1 Manufacturing Line Definition

3.1.1 General Description -- The basic manufacturing line will consist of ten automatic assembly machines and two spring winders which will assemble individual piece parts and/or subassemblies for the XM74 Extended Range Tripline Sensor Program. The manufacturing line flow is shown in Figure A-1. The output product of each machine is directly acceptable to the machine following in sequence so as to maintain continuity throughout the manufacturing line.

3.1.2 Capability -- A complete manufacturing line, consisting of one each of the ten automatic assembly machines, would have as a nominal line rate the following capability when expressed at the maturity levels indicated:

Cumulative production at 100,000 total	30,000 sensors/mo.		
500,000 total	45,000	"	"
1,000,000 total	55,000	"	"

The line output capacities have been adjusted for efficiency, scrap and salvage and the values expressed are net output of sensors delivered.

The limiting machine in the overall line production rate is the Bobbin Assembly Machine. This machine is a line limiter until three Bobbin Assembly Machines are available when other machines become the pacing item with a limiting output as follows:



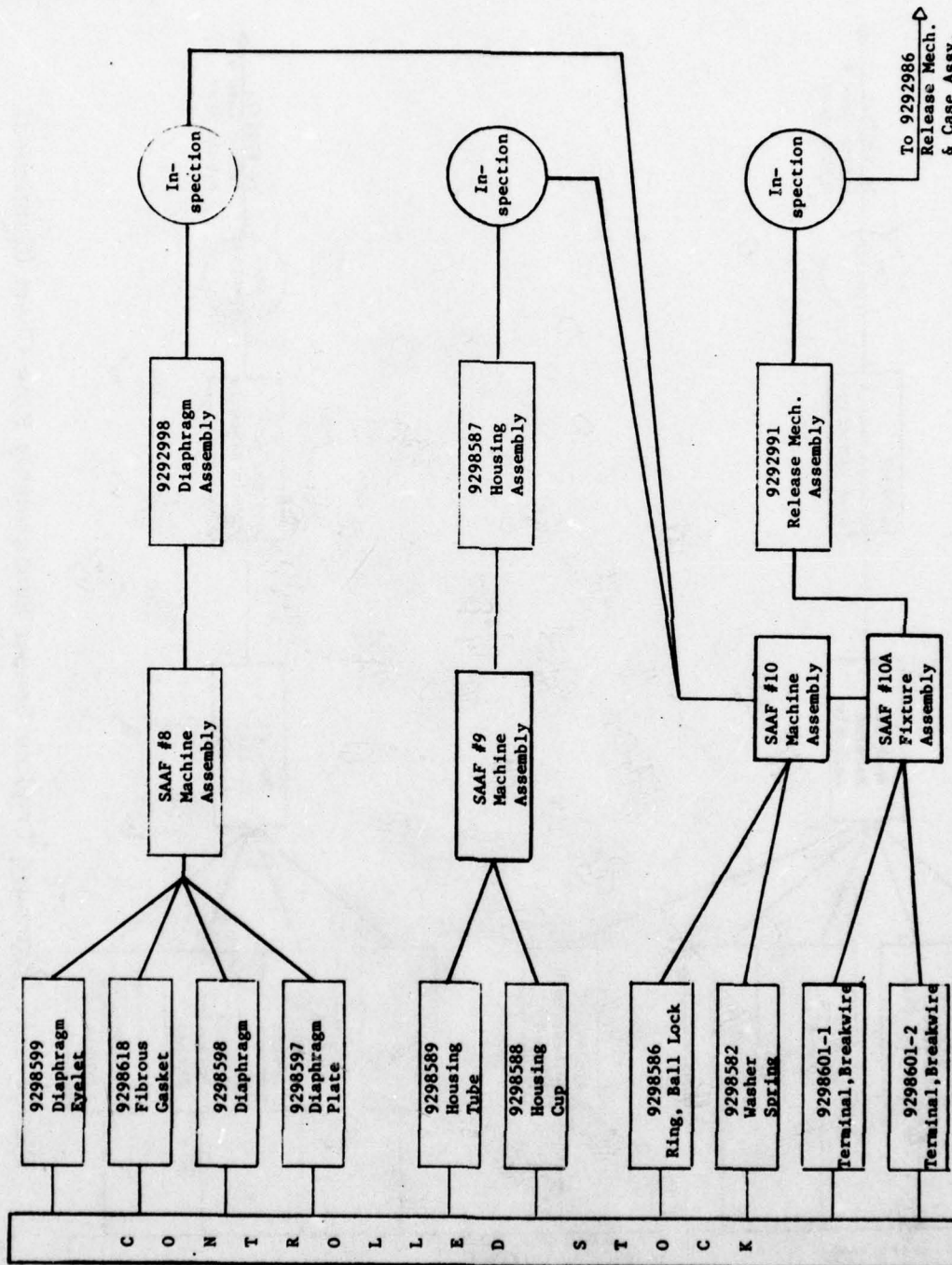


Figure A-1. Extended Tripline Sensor Subassembly Flow Chart

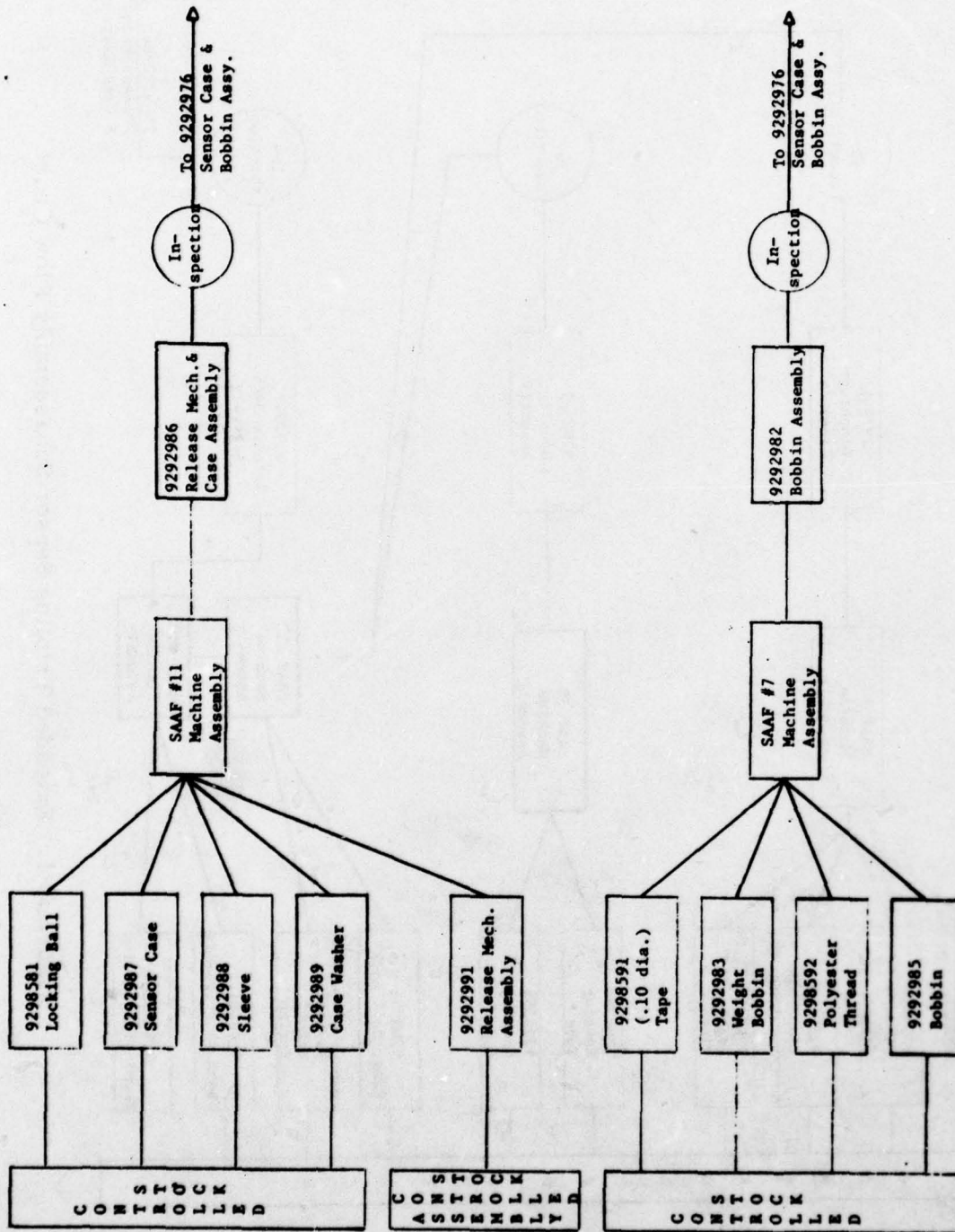


Figure A-1. Extended Tripline Sensor Subassembly Flow Chart (Continued)

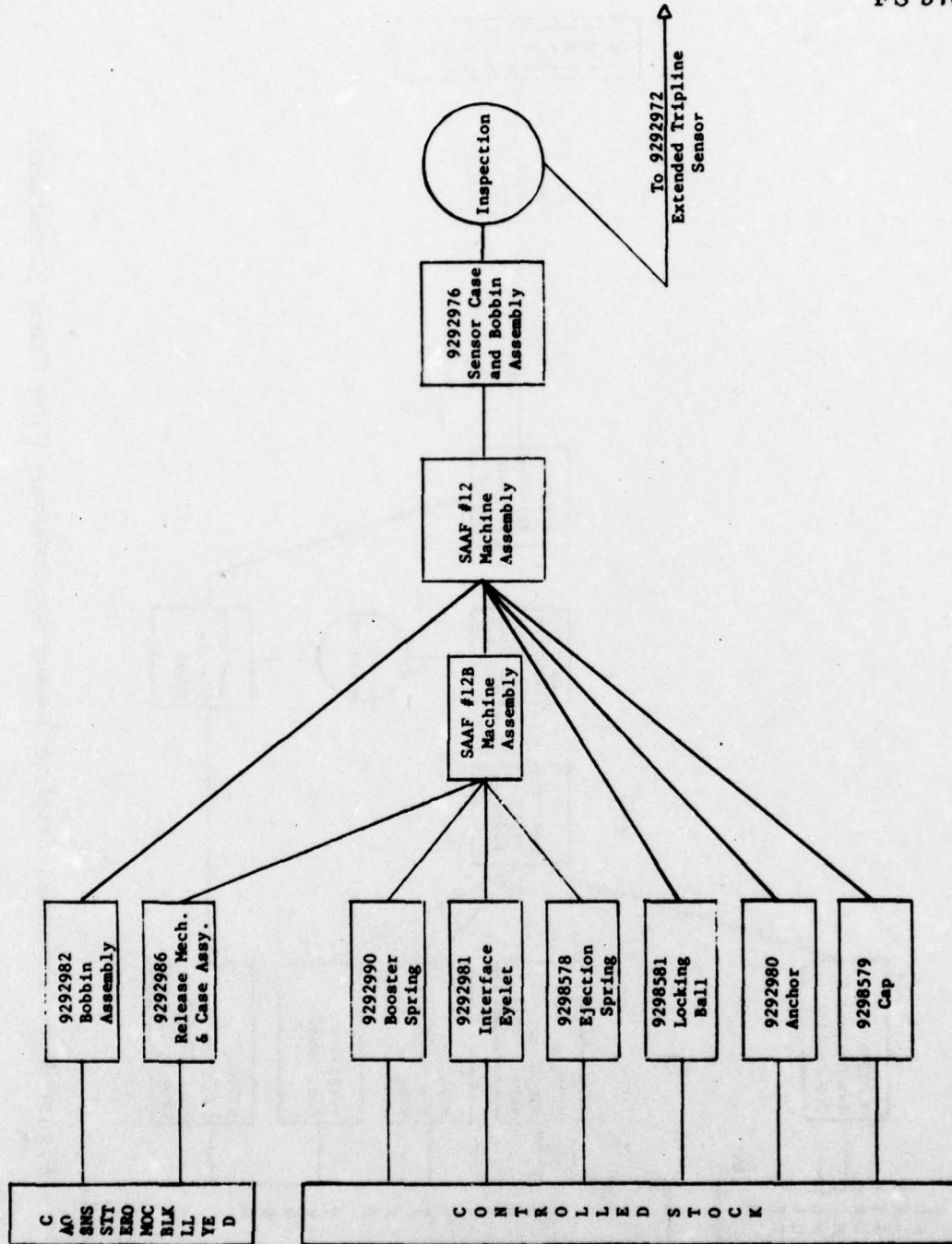


Figure A-1. Extended Tripline Sensor Subassembly Flow Chart (Continued)



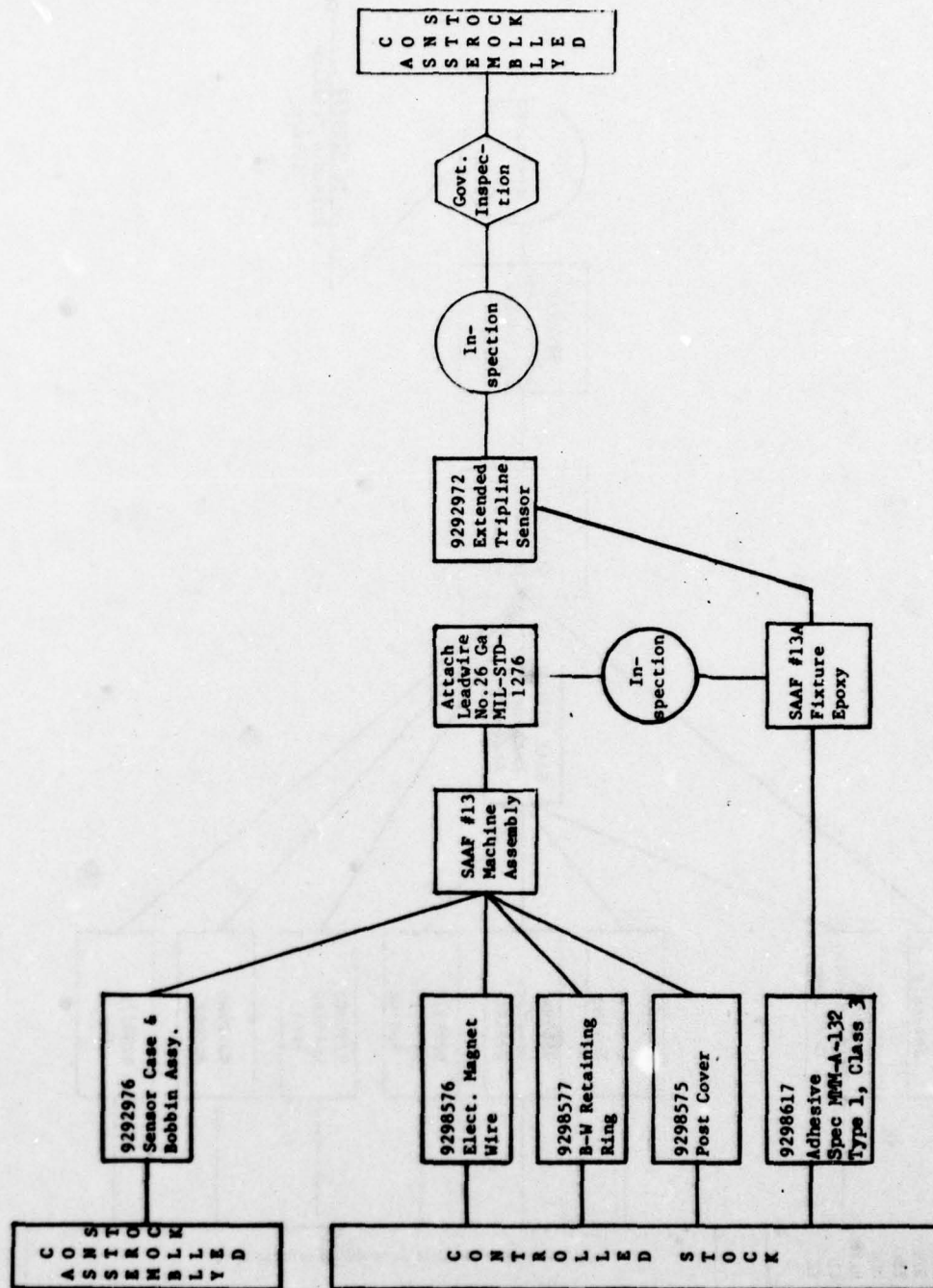


Figure A-1. Extended Tripline Sensor Subassembly Flow Chart (Concluded)

At 100,000 total	90,000 sensors/mo. net			
500,000 total	135,000	"	"	"
1,000,000 total	165,000	"	"	"

The line output for various combinations of machines at the three maturity levels is further delineated in Table A-2.

3.1.3 Manufacturing Line Diagram -- Figure A-1 shows the assembly line flow of piece parts and subassemblies through the manufacturing line. Following is a brief description of each assembly machine which includes a discussion of the major functions conducted. The applicable piece part and subassemblies required are listed for each machine.

#### 3.1.4 Interface Definition --

3.1.4.1 Compatibility with Other Lines -- The special automatic assembly line for production of the GEMSS XM74 Tripline Sensor has commonality to varying degrees with the automatic assembly line used to assemble the ADAM Sensor (ref. Table A-1); however, if the GEMSS Sensor Line is self-contained, there would be no interface situation between the two lines. If a GEMSS line were installed without the common machines (i.e., Machine Nos. 8, 9, 10, and 10A) for the Release Mechanism, then there would be an interface between Machine 10A and Machine 11 where the two lines become separate. This provides an option with respect to the number of machines installed which is dependent upon the combined production levels.

3.1.4.2 Facilities/Utilities -- Because of the similarity between the ADAM Line and the GEMSS Line, there will be common usage of certain facilities and utilities. The following facility items would be shared by both assembly lines:

TABLE A-2. NET PRODUCTION PER MONTH AT 100, 000, 500, 000 AND 1, 000, 000 LEVELS

	MACHINE NO.		QUANTITIES OF MACHINES						
At 100,000 Total Produced	7	1	2	3	4	5	6	7	
	8	1	1	1	2	2	2	3	
	9	1	1	1	1	1	2	2	
	10	1	1	2	2	3	3	4	
	10A	1	1	1	1	2	2	2	
	11	1	1	1	2	2	3	3	
	12B	1	1	1	2	2	3	3	
	12	1	1	2	2	3	4	4	
	13	1	1	2	2	3	4	4	
	13A	1	1	1	2	2	2	3	
	Output (1,000's)	30	60	90	120	150	180	210	
At 500,000	7	1	2	3	4	5			
	8	1	1	1	1	2			
	9	1	1	1	1	1			
	10	1	1	1	2	2			
	10A	1	1	1	2	2			
	11	1	1	1	2	2			
	12B	1	1	1	2	2			
	12	1	1	2	2	2			
	13	1	1	2	2	2			
	13A	1	1	2	2	2			
	Output (1,000's)	45	90	135	180	225			
At 1,000,000	7	1	2	3	4				
	8	1	1	1	1				
	9	1	1	1	1				
	10	1	1	1	2				
	10A	1	1	1	2				
	11	1	1	1	2				
	12B	1	1	1	2				
	12	1	1	1	2				
	13	1	1	1	2				
	13A	1	1	2	2				
	Output (1,000's)	55	110	165	220				



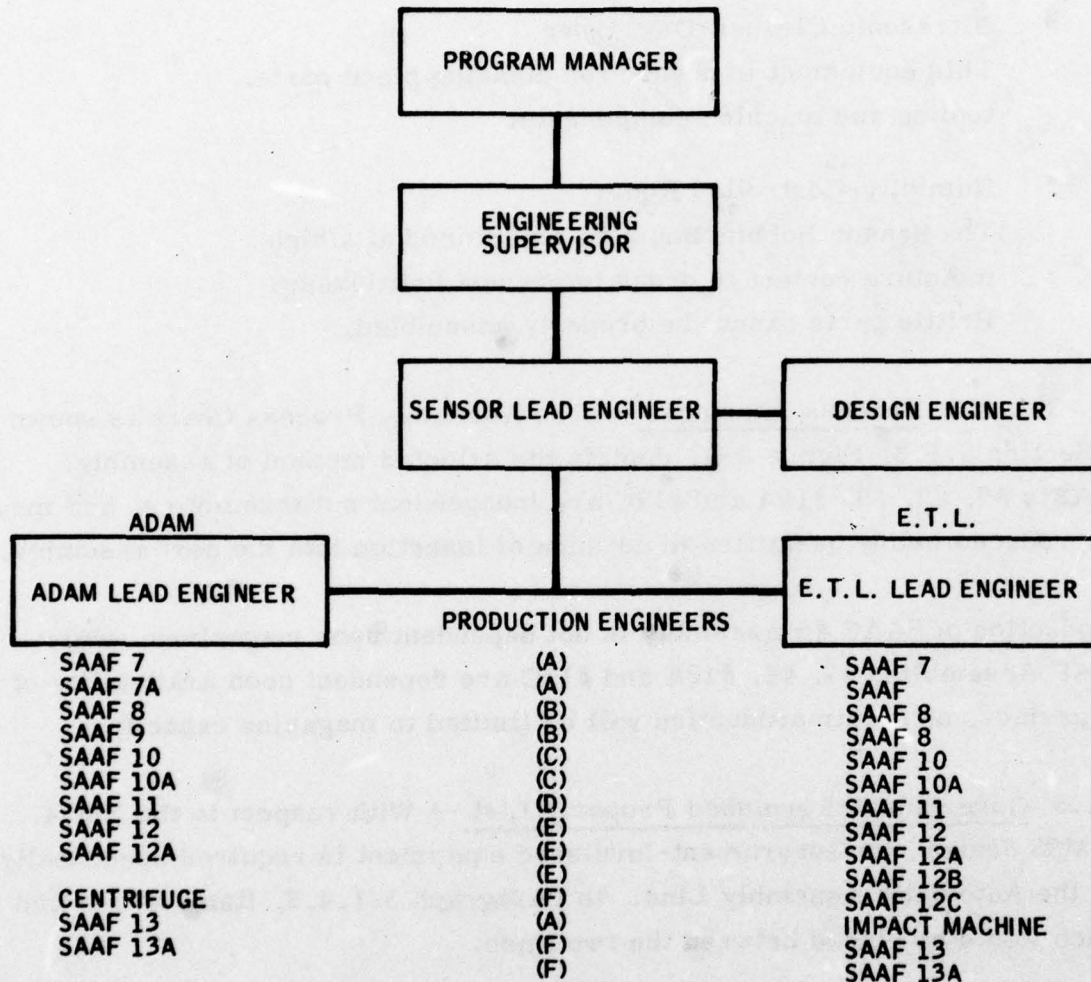
- **Demagnetizer**  
The demagnetizer is used to demagnetize piece parts, tooling and machine components as required.
- **Ultrasonic Cleaner-Degreaser**  
This equipment is needed for cleaning piece parts, tooling and machine components.
- **Humidity-Controlled Room**  
The Sensor Bobbins must be maintained at a high moisture content in order to prevent brittleness. Brittle parts cannot be properly assembled.

3.1.4.3 Process Interfaces -- The Assembly Process Chart as shown in Section 3.1.3, Figure A-1, depicts the selected method of assembly. SAAF's #7, #8, #9, #12A and #12C are independent subassemblies, and may be produced in any quantities in advance of insertion into the next assembly.

Production of SAAF #9 Assembly is not dependent upon magazines, while SAAF Assemblies #7, #8, #12A and #12C are dependent upon availability of magazines, and their production will be limited to magazine capacity.

3.1.5 Government-Furnished Property List -- With respect to the XM74 GEMSS Sensor, no Government-furnished equipment is required specifically for the Automatic Assembly Line. In Paragraph 3.1.4.2, items are listed which would be placed between the two lines.

3.1.6 Organizational Concepts -- The close relationship between the GEMSS and ADAM Sensor will result in the assignment of engineering personnel with dual responsibility for similar automatic assembly fixtures.



### 3.2 Characteristics

3.2.1 Performance Characteristics -- Special automatic assembly fixtures will produce 750 to 1250 completed assemblies per hour, depending upon the complexity of the assembly.

When the SAAF is double nested, as is the case in GEMSS #9, the output is approximately double the 750 to 1250 assemblies per-hour rate.

The life expectancy of any SAAF is indeterminate. Each component used in the basic machine and the special tooling is replaceable on a modular basis. For example, the contractor has converted WAAPM SAAF's for use on the ADAM Machine Line with acceptable results.

### 3.2.2 Physical Characteristics --

#### Production Area:

- a. The space required for a typical SAAF is 400 square feet of floor space (20 ft. x 20 ft. square). This area will support parts and assemblies going into the machine, as well as the completed assemblies. The need for aisles and cart storage space is not included in the 400 square feet for the SAAF.
- b. The space required for a typical spring winder (there are two on the E.T.L. Sensor), including stress relief ovens (two), would be 625 square feet of floor space. This area would support the carts used to load and store the completed springs, and to deliver the springs to SAAF's 12 and 12B.
- c. The space required for an ultrasonic degreaser for cleaning of some parts and especially for cleaning SAAF #11 Nest would be 25 square feet (5 ft. x 5 ft. square).



- d. The space required for mixing epoxy for SAAF #13A would be 100 square feet (10 ft. x 10 ft. square). This area would require a special exhaust system to the outside of the building.
- e. The space required for sensor lot acceptance testing would be 600 square feet (60 ft. long x 10 ft. wide). This requires a clear area 20 ft. high mid-point in the length of the test area.
- f. Allow 10 percent of total for aisles, break area and foreman's office.

Production Area Required:

a. 400 sq. ft. x 10 machines	4000 sq. ft.
b. Spring winding area	625 sq. ft.
c. Ultrasonic degreaser	25 sq. ft.
d. Epoxy mixing area	100 sq. ft.
e. Sensor LAT area	600 sq. ft.
Subtotal	5350 sq. ft.
f. 5350 x 10%	535 sq. ft.
Grand Total	5885 sq. ft.

There are no security requirements with the SAAF or the E.T.L. Sensor with regard to the manufacturing area.

There are two areas where health considerations must be addressed, which are listed in Subparagraph "d" above. The outgassing of the epoxy must be exhausted outside of the work area (building) immediately. The epoxy is

dispensed on Machine 13A, so it must be equated with the exhaust system to remove the outgas from the epoxy.

The E.T.L. SAAF will meet all OSHA requirements prior to their installation on the E.T.L. production floor.

**3.2.3 Reliability** -- The reliability of each machine in the GEMSS XM74 Extended Sensor production line will have as a design goal a minimum mean-time-between failure (MTBF) of 8.5 hours. A failure is defined as any unscheduled machine stoppage during a scheduled production period which requires repair maintenance personnel action due to breakage or severe misalignment (i.e., malfunction) of any machine subassembly or component. The time the machine runs is the production time associated with machine cycling, but not including any cycling during repairs or other cycling during which good product could not be produced. During the machine performance demonstrations and qualification testing, the number of failures and the production time will be collected as needed to calculate the machine mean-time-between-failure so that reliability growth can be analyzed.

**3.2.4 Maintainability** -- The maintainability of each machine used in the GEMSS Extended Tripline Sensor production line will have as a design goal a mean-time-to-repair (MTTR) of 1.75 hours. To achieve this goal, the machines will be designed to comply with the following:

- a. **Modularization** - The machines shall be designed to make maximum use of modularized subassemblies for each type of maintenance and to reduce the number and type of repair parts and assemblies required to support maintenance.
- b. **Accessibility** - Maximum use will be made of design techniques that will provide ready accessibility for replacement or servicing.

- c. Interchangeability - Components will be interchangeable without requirements to calibrate or adjust at time of replacement.
- d. Maintenance Periods - The system will be designed so that preventive maintenance is required only in off-production hours.

All machines are analyzed for MTTR using the definition of failure presented in Paragraph 3.2.3. The repair time will include preparation time, fault location time, fault correction time, adjustment/calibration time and check-out time. If two or more repair personnel are involved, only elapsed time will be measured. However, the number, skills, and elapsed time for each repair person shall be recorded. During the machine performance demonstrations and qualification testing, the number, type of failure, and the time to repair will be collected so that machine maintainability characteristics can be analyzed.

3.2.5 Availability -- The machines used in the GEMSS XM74 Extended Trip-line Sensor production line shall be designed to minimize the downtime (MTTR) and to maximize the uptime (MTBF) for each machine. The machines shall be designed to have a minimum inherent availability of 85 percent. The inherent availability is defined as:

$$\text{Availability} = \frac{\text{MTBF}}{\text{MTBF} + \text{MTTR}} \times 100\%$$

Where MTBF = Mean Time Between Failures

MTTR = Mean Time to Repair

3.2.6 Environmental Conditions -- The toxic nature of the outgassing from the epoxy mixture is the only known natural environmental item. The experience gained from the ADAM Sensor is that when the outgassing is exhausted to the outside air, there is no problem, as the epoxy gasses are quickly dispelled.



### 3.3 Design and Construction

3.3.1 Name Plates and Marking -- There is no general criteria used for identification and markings on electrical and pneumatic lines and controls. Identification here is only used when necessary. Lines are identified whenever hydraulic or water lines are concerned to avoid confusion in the machine system.

3.3.2 Interchangeability -- Standardized tooling and castings have been utilized in the design and assembly of assembly machines. Some of the standardized features include: probes, parts feeding mechanisms, magazine feeders, work stations, etc. This major emphasis on standardization throughout reduces the number and type of repair parts required to support maintenance and maintain interchangeability to a maximum.

3.3.3 Safety -- Through the use of good design and the proper selection of material, machine safety shall be in conformance with the intent, and general interpretation of the following safety publications and codes:

- a. AMCR385-100, AMC Safety Manual
- b. AR40-5, Health and Environment
- c. MIL-STD-882A, System Safety Program Requirements
- d. Occupational Safety and Health Act (OSHA)
- e. National Electric Code for Hazardous Operations

The Safety Engineering effort will be coordinated with RAM and HFE efforts to establish an integrated program. A safety review of each machine shall be conducted to determine compliance with the above-listed publications.

3.3.4 Human Performance/Human Engineering -- Each machine in the GEMSS XM74 Extended Tripline Sensor production line shall be in conformance with the intent and general interpretation of the following documents:

- a. MIL-STD-1472B Human Engineering Design Criteria  
for Military Systems Equipment and  
Facilities
- b. MIL-HDBK-759 Military Standardization Handbook,  
Human Factors Engineering Design  
for Army Material

A human factors review of each machine shall be conducted prior to release of the machine for use in the normal production environment to determine compliance with the above-listed documents.

### 3.4 Process/Operation Characteristics/Configuration Items

3.4.1 Flow Diagram -- The process/operation flow diagram is included in Section 3.1.3, Figure A-1.

### 3.4.2 Function and Physical Interfaces --

#### Physical Interface Between SAAF --

Physical Interface of Parts  
From SAAF to SAAF and Assembly Between SAAF is by

8	10	By magazines
9	10	Bulk - into #10 by feed bowl
10	10A	By magazines
10A	11	By magazine
11	12B	By magazine
12C	12B	By spring magazine
12A	12B	By spring magazine
12B	12	By magazine
12	13	By magazine
13	13A	By magazine
7	12	Off #7 to container, then hand loaded into magazines

Functional Interfaces Between SAAF -- The operation accomplished on SAAF #12B has a functional interface between SAAF's 12B and 12. The ejection spring and the booster spring are compressed and a retaining pin is inserted into the assembly on SAAF 12B. The retaining pin is removed on SAAF 12.

This is the only functional interface in the GEMSS Sensor Automatic Assembly Line.

3.4.3 Functional and Flow Diagrams -- Functional diagrams (dial layouts) are included with the description of each of the assembly machines. Line flow diagram (Figure A-1) is included as part of Section 3.1.



MACHINE #7  
BOBBIN ASSEMBLY, 9292982

PARTS

- Bobbin, 9292985
- Thread, 9298592
- Tape, 9298591
- Weight Bobbin, 9292983

Bobbin has two parts:

Core, which thread is wound around, and post which fits into the Release Mechanism Assembly.

PURPOSE OF ASSEMBLY

To store thread, approximately 46 feet, which is wound about the Bobbin Core in a manner that it will easily deploy from the Bobbin when the Sensor Assembly is functioned.

The tape is used only to maintain one end of thread in proper position.

The weight is used to trap the other end of the thread, which will maintain a pull force consistent with the requirement of the drawing; see Figure A-2 and A-3.

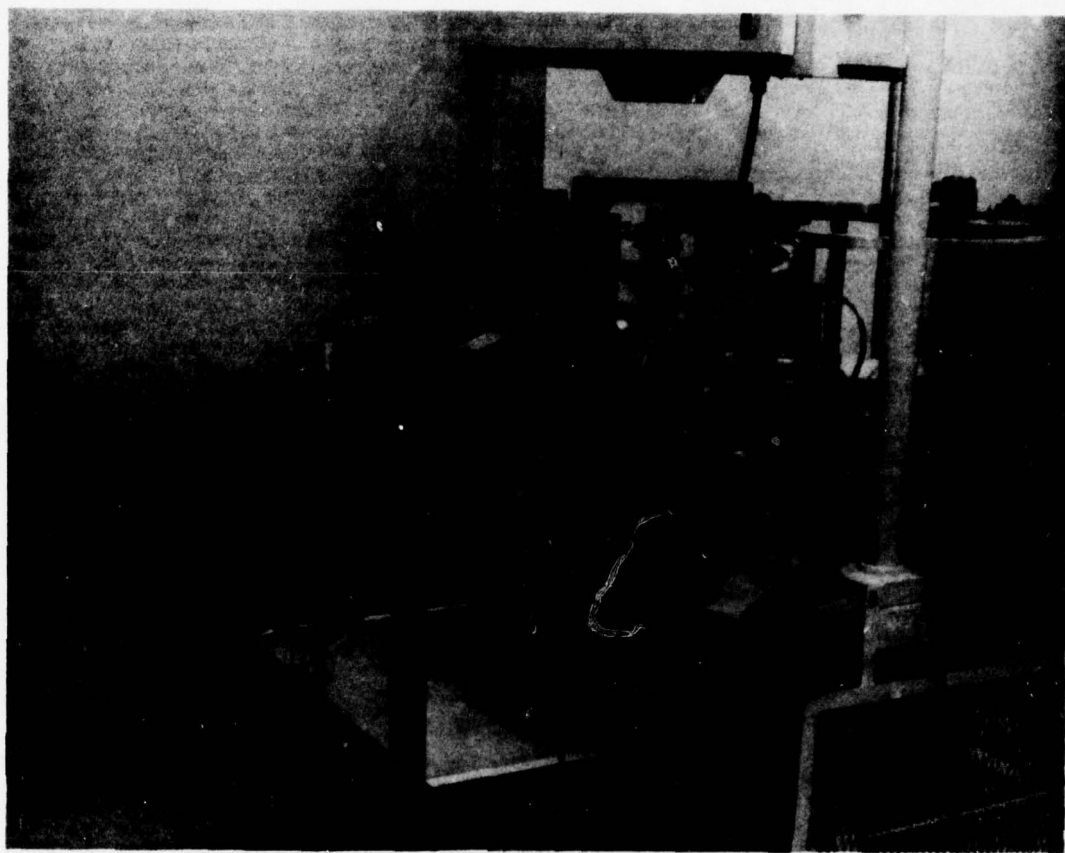


Figure A-2. WAAPM-Type Semi-automatic Bobbin Winding and Taping Machine

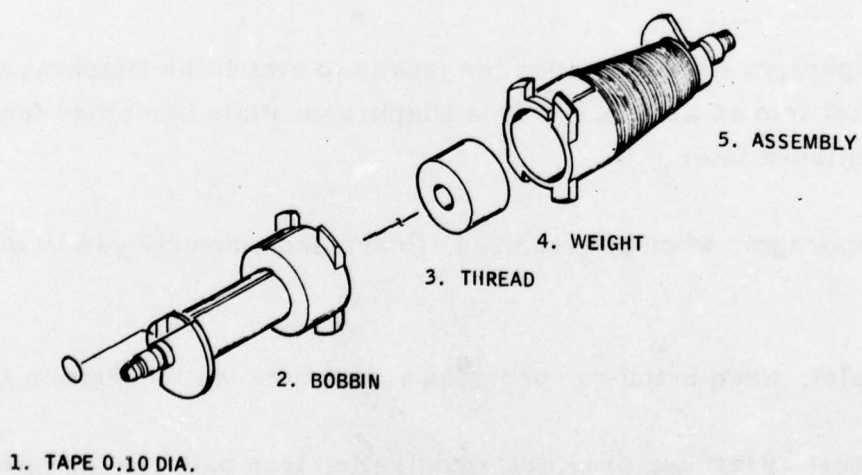


Figure A-3. XM74 Bobbin Assembly

**MACHINE #8**  
**DIAPHRAGM ASSEMBLY, 9292998**

**PARTS**

- Diaphragm, 9298598
- Diaphragm Plate, 9298597
- Eyelet, Diaphragm, 9298599
- Gasket, Fibrous, 9298618

**PURPOSE OF ASSEMBLY**

The Diaphragm Assembly, Figure A-4, is used to convert gas pressure, which is delivered to the Diaphragm, into uniform mechanical motion, which results in sensor release.

The Diaphragm Plate provides the means to attach the Diaphragm Eyelet and the gasket into an assembly. The Diaphragm Plate has other functions, which are mentioned later.

The Diaphragm, when pressurized, flexes and converts gas to mechanical motion.

The Eyelet, when crimped, provides a seal between the Plate and Diaphragm.

The Gasket, Fibrous, provides a controlled leak path for the gas pressure to leak off after the sensor has been released.



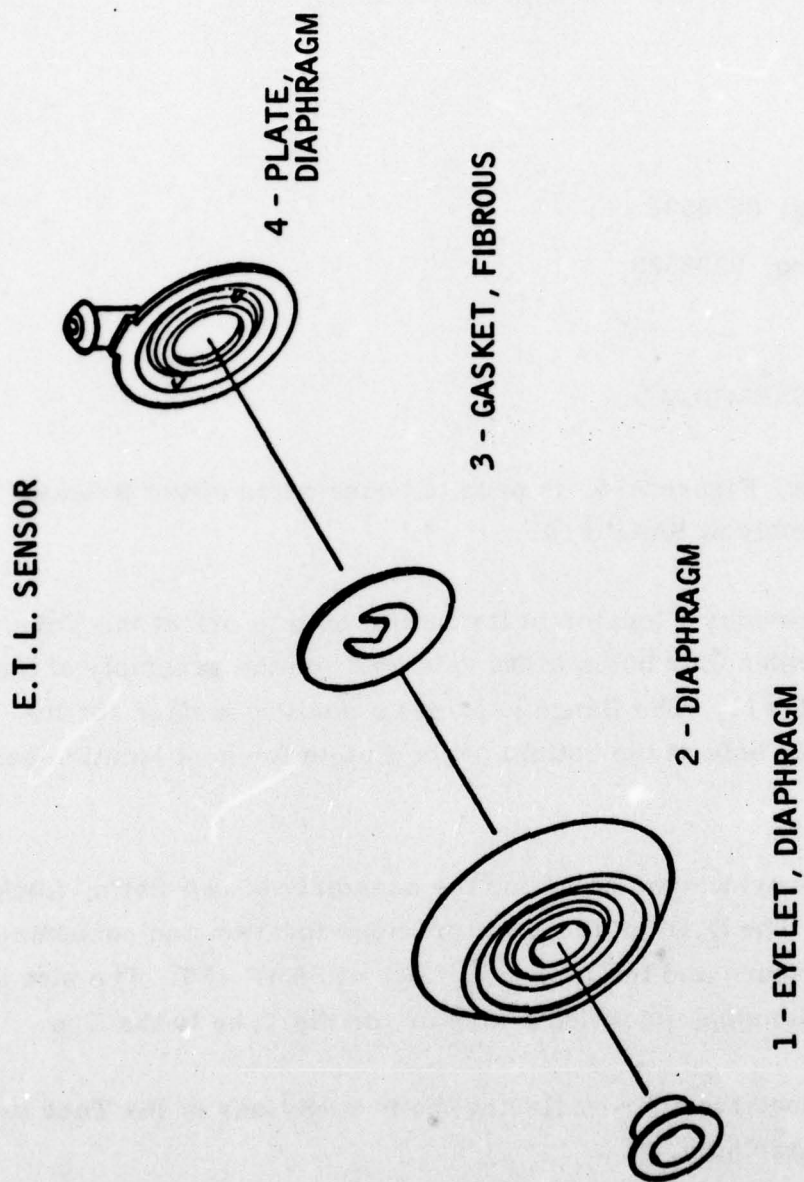


Figure A-4. Machine No. 8 - Diaphragm Assembly

MACHINE #9  
HOUSING ASSEMBLY, 9298587

PARTS

- Cup, Housing, 9298588
- Tube, Housing, 9298589

PURPOSE OF THE ASSEMBLY

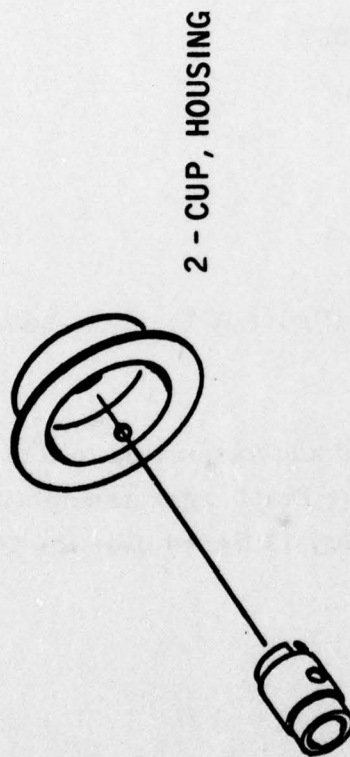
The Housing Assembly, Figure A-5, is used to house parts of the Release Mechanism, the assembly at SAAF #10.

The Cup, Housing, provides a locator in its center hole to orient the Tube, Housing. It also provides four holes in its side wall for the assembly of four Locking Balls at SAAF #11. The flange provides a sealing surface for the Diaphragm. The small hole at the bottom of the Cup is for nest location purposes only.

The Tube, Housing, provides two holes for the assembly of two Balls, Locking, at Machine #12. The O.D. of the Tube provides location and concentricity for the two Spring Washers and the Ring Ball Lock at SAAF #10. The slot in the end of the Tube, Housing, provides a locator for the Tube to the Cup.

The assembly is crimped together by flaring the two sections of the Tube over the Cup, Housing, center hole.

E.T.L. SENSOR



1 - TUBE, HOUSING

Figure A-5. Machine No. 9 - Housing Assembly



MACHINE #10  
RELEASE MECHANISM ASSEMBLY, 9292991

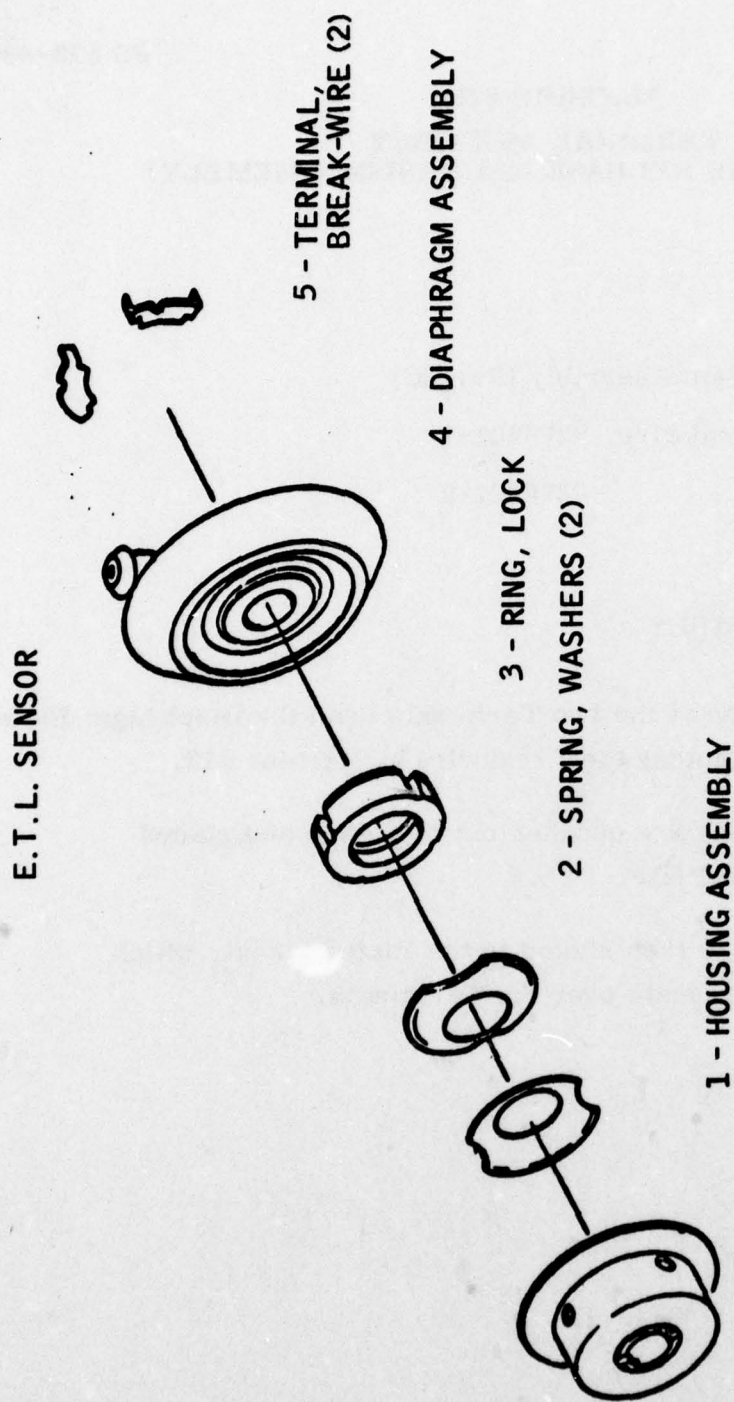
PARTS

- Housing Assembly, 9298587
- Diaphragm Assembly, 9292998
  - Spring Washer (2), 9298582
  - Ring, Ball Lock, 9298586

PURPOSE OF THE ASSEMBLY

The Release Mechanism Assembly, Figure A-6, is the heart of the sensor release function.

The two Spring Washers are oriented and assembled into the Housing Assembly, followed by the Ring, Ball Lock. The Diaphragm Assembly is placed on the assembly, and then the Tube, Housing, is flared over the Diaphragm Eyelet, completing the assembly.



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Figure A-6. Machine No. 10A - Release Mechanism Assembly

MACHINE #10A  
TERMINAL ASSEMBLY  
(PART OF THE RELEASE MECHANISM ASSEMBLY)

PARTS

- Release Mechanism Assembly (Partial)
  - Terminal, Breakwire, 9298601-1
  - 9298601-2

PURPOSE OF THE ASSEMBLY

The placement and location of the two Terminals upon the Diaphragm Plate is to provide a method to solder the Breakwire at Machine #13.

- The two Terminals are punched out of a strip and placed on the Diaphragm Plate.
- The Terminals are then staked to the Plate by heat, which stakes two plastic posts over the Terminals.



MACHINE #11  
RELEASE MECHANISM AND CASE ASSEMBLY, 9292986

PARTS

- Release Mechanism Assembly, 9292991 (Figures A-7 and A-8)
  - Sleeve, 9292988
  - Case, Sensor, 9292987
  - Case, Washer, 9292989
  - Ball, Locking (4), 9298581

PURPOSE OF THE ASSEMBLY

The four Balls are assembled into the four holes of the Sleeve and the Cup, Housing. The crimp of the Sensor Case completes the assembly.

- The four holes in the Sleeve provide the means to contain the four Locking Balls.
- The Case, Sensor, provides the envelope protection for the Sensor, and the case crimp contains all parts in the proper position.
- The Case, Washer, locates the Release Mechanism Assembly in the Sensor Case and captivates the four Locking Balls in the proper position.
- The four Locking Balls retain the Sleeve between the O.D. of the Cup, Housing, and the O.D. of the Case, Washer.

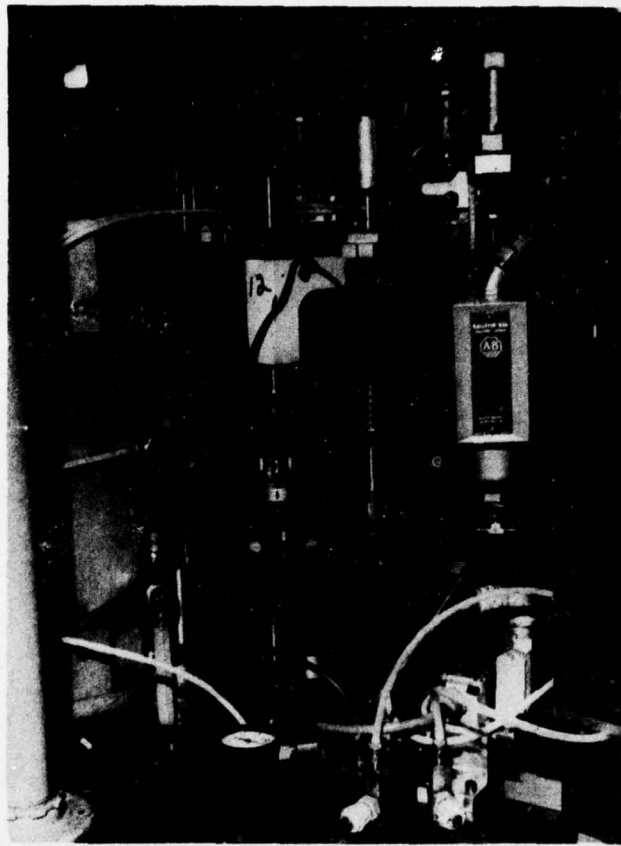


Figure A-7. ADAM Release Mechanism and Case Assembly Machine

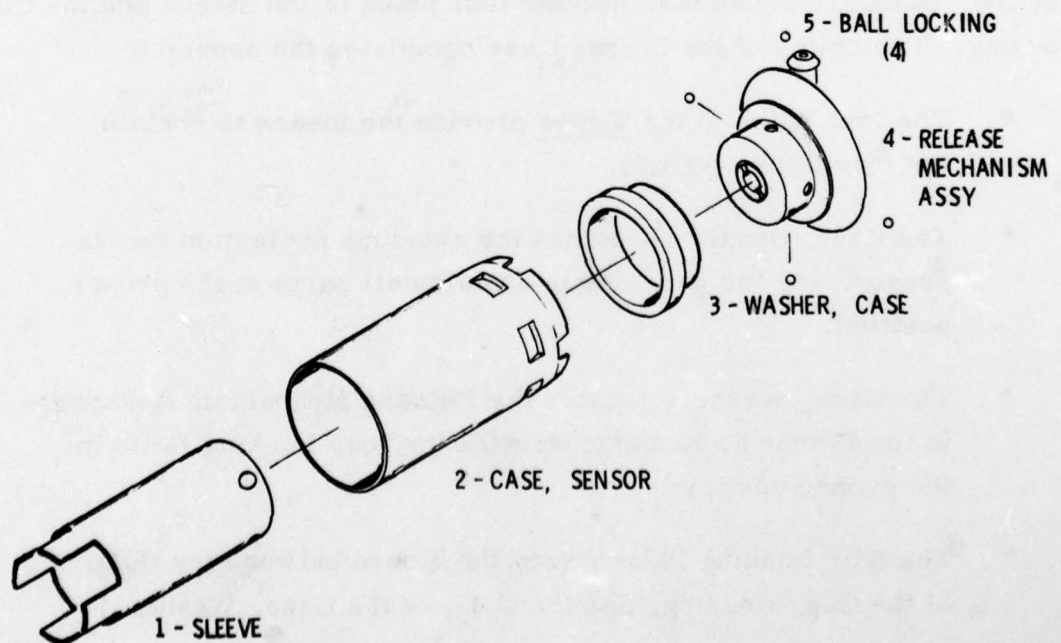


Figure A-8. XM74 Release Mechanism and Case Assembly

MACHINE #12  
SENSOR CASE AND BOBBIN ASSEMBLY, 9292976 (COMPLETE)

PARTS

- Sensor Case and Spring Assembly (from SAAF 12B)
  - Bobbin Assembly, 9292982
  - Ball, Locking (2), 9298581
  - Cap, 9298579
  - Anchor, 9292980

PURPOSE OF THE ASSEMBLY

The two Balls are assembled into the two holes of the Tube, Housing (Machine #9).

The Bobbin Assembly, Figures A-9 and A-10, is assembled into the Tube Housing (Machine #9) and at the same time into the Sleeve.

- The Balls (2) provide the Release Mechanism the capability to lock up the Bobbin Post while the thread of the Bobbin Assembly is being deployed.
- The Anchor is bowled and compressed to fit into the I.D. of the Sensor Case.
- The Cap is a part which adds weight to the Bobbin Assembly for better deployment. It also captivates the Bobbin within the legs of the Sleeve, and the four tabs of the Cap lock over



the four "legs" of the Sleeve, so the assembly will survive the launch environment.

- The Bobbin Assembly is partially drilled on this machine to reduce drilling "heat" on Machine #13.

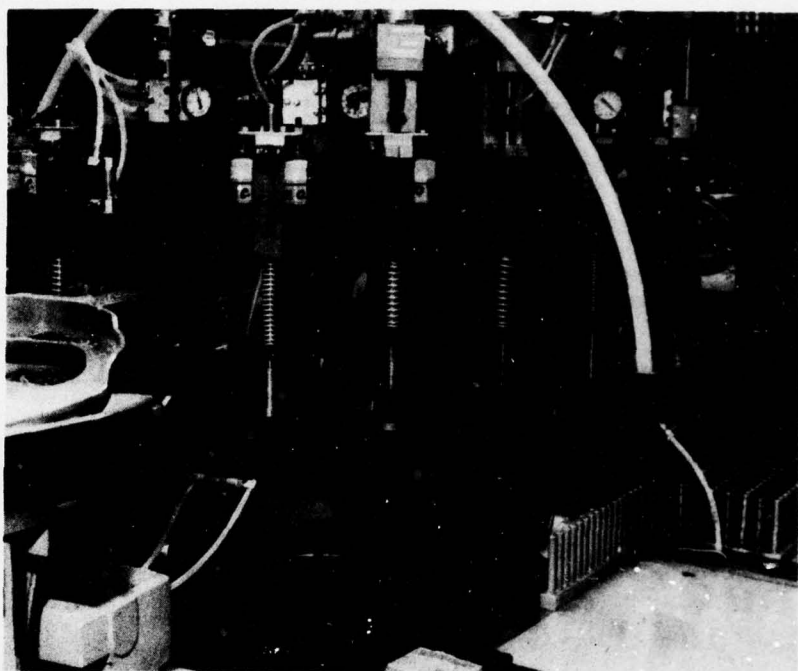


Figure A-9. ADAM Sensor Case and Bobbin Assembly Machine

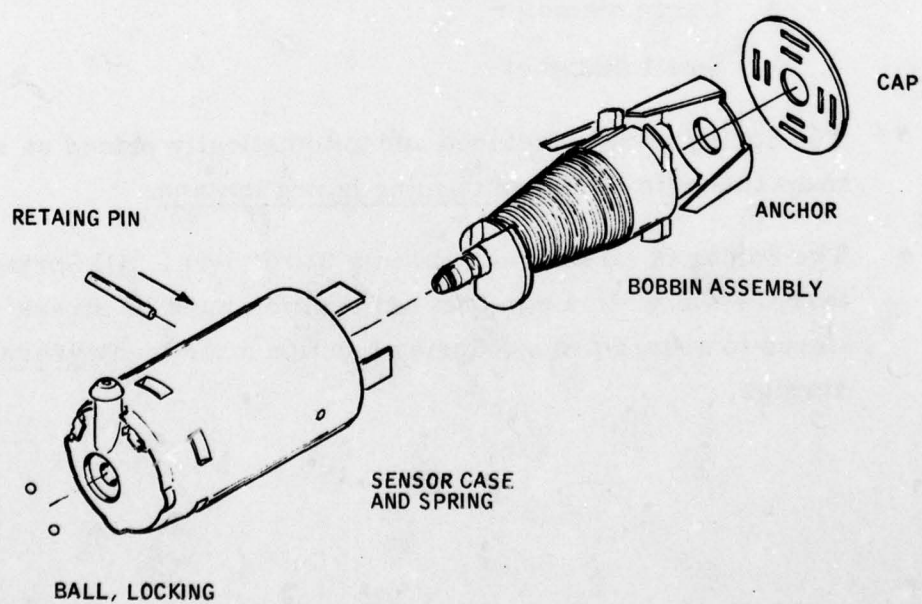


Figure A-10. XM74 Sensor Case and Bobbin Assembly

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MACHINE #12A  
SPRING, EJECTION, 9298578

PARTS

None

PURPOSE OF THE SPRING

The Spring provides the power and weight to deploy the thread past the minimum deployment requirement.

- The Spring has two diameters:
  - a. Large diameter
  - b. Small diameter
- The Spring is manufactured and automatically placed on a magazine with the large opening facing upward.
- The Spring is stress relieved in a 350°F oven. All Springs that are stored in a compressed position must be stress relieved to assure proper Spring function after many years of storage.



MACHINE #12B  
SENSOR CASE AND SPRING ASSEMBLY, 9292976 (PARTIAL)

PARTS

- Release Mechanism and Case Assembly, 9292986 (Figure A-11)
  - Spring, Booster, 9292990
  - Eyelet, Interface, 9292981
  - Spring, Ejection, 9298578

PURPOSE OF THE ASSEMBLY

The Booster Spring and the Ejection Spring provide the power to deploy the thread past the minimum deployment requirement.

The large I.D. of the Booster Spring will lock onto the diameter provided on the Case, Washer.

The small I.D. of the Ejection Spring will lock onto the outside diameter of the Sleeve.

The Eyelet, Interface, separates the Booster Spring from the Ejection Spring. When the thread deploys, it will be contained within the I.D. of the Eyelet.

The Springs (2) will be compressed and remain in the compressed condition until SAAF #12, where a retaining pin will be removed for assembly of the two Locking Balls, Bobbin Assembly, Anchor and the Cap.

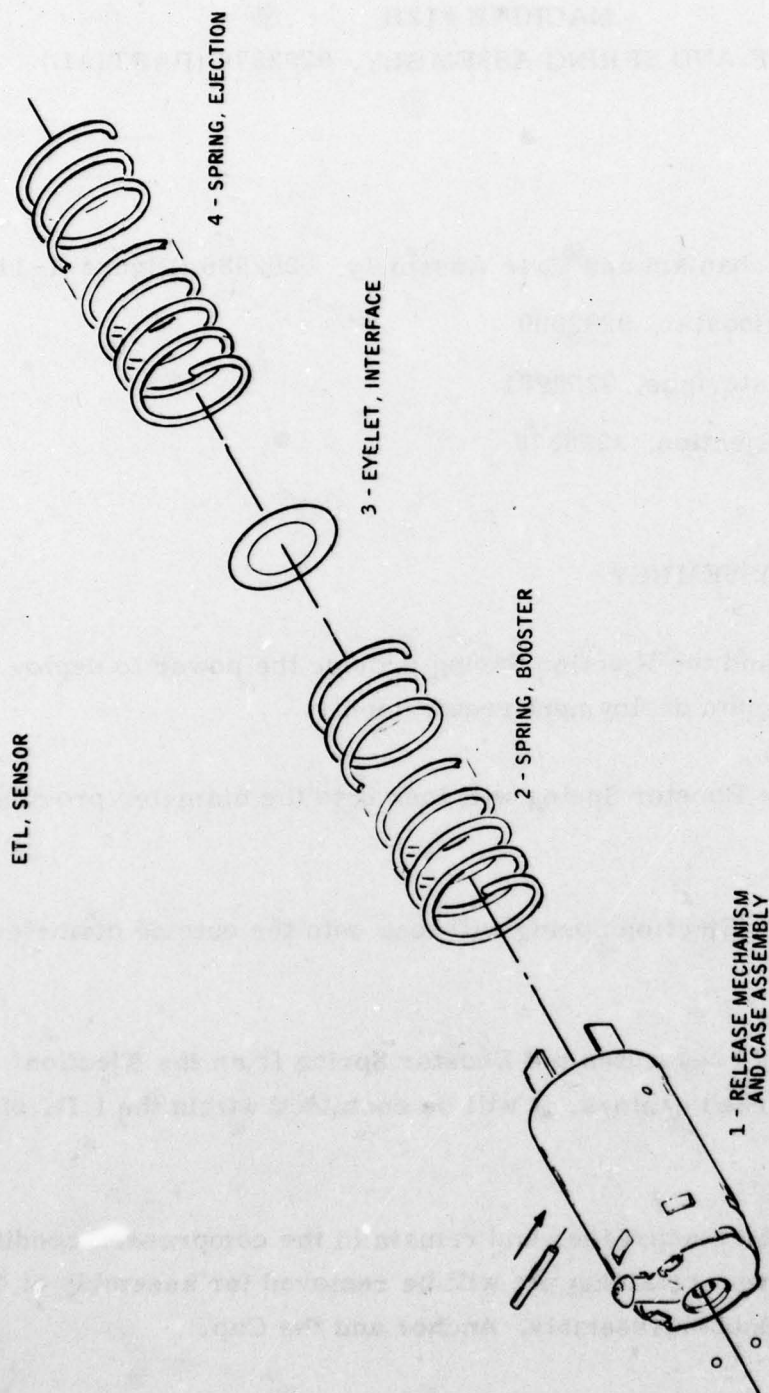


Figure A-11. Machine #12B - Sensor Case and Spring Assembly

MACHINE #12C  
SPRING, BOOSTER, 9292990

PARTS

None

PURPOSE OF THE SPRING

The Spring provides the power and weight to deploy the thread past the minimum deployment requirement.

- The Spring has two diameters:
  - a. Large diameter
  - b. Small diameter
- The Spring is manufactured and automatically placed on a magazine with the large opening facing upward.
- The Spring is stress relieved in a 350°F oven. All Springs that are stored in a compressed position must be stress relieved to assure proper Spring function after many years of storage.



MACHINE #13  
SENSOR ASSEMBLY, 9292972

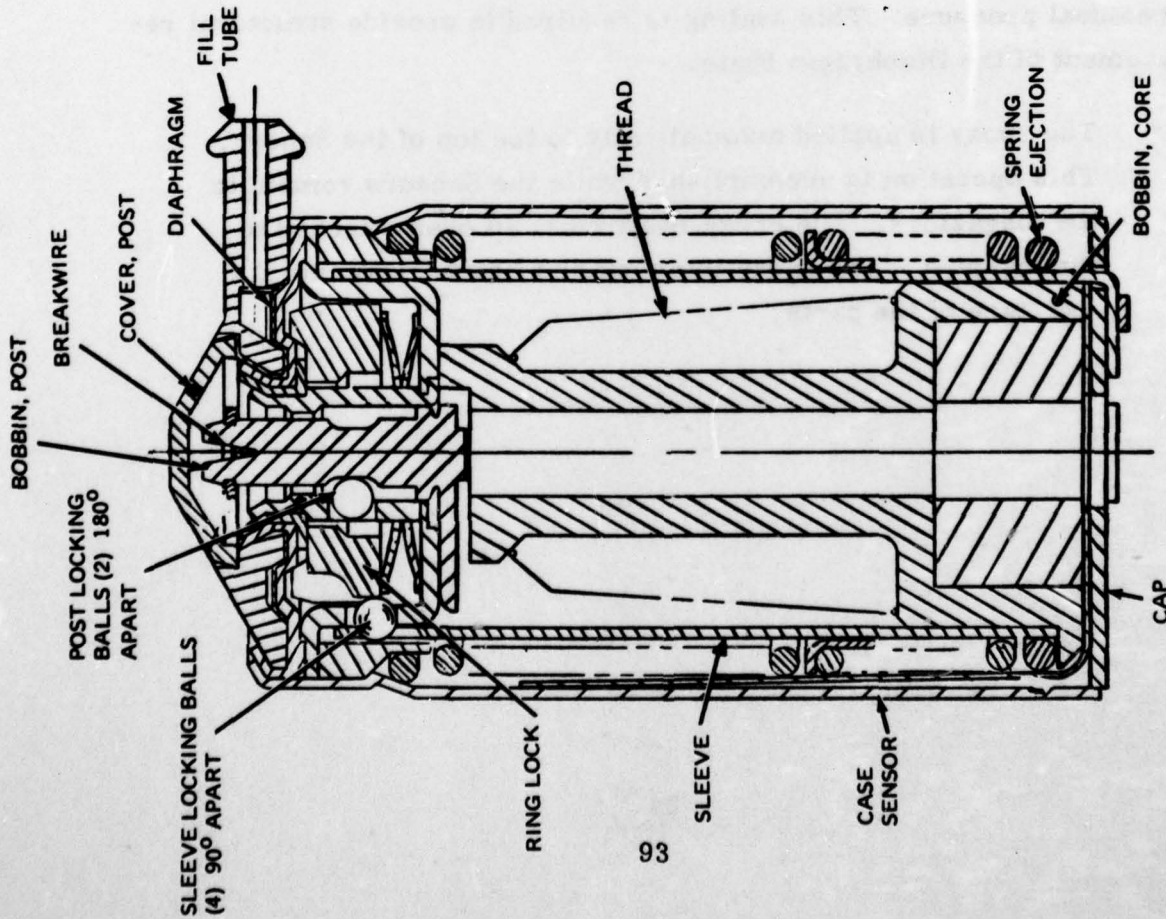
PARTS

- Sensor Case and Bobbin Assembly, 9292976 (Figure A-12)
  - Breakwire, 9292576
  - Ring, Retaining, 9298577
  - Cover, Post, 9298575

PURPOSE OF THE ASSEMBLY

The Breakwire is placed into the 'V' of the Bobbin Post. The Retaining Ring is placed into the radial groove of the Bobbin, the Post Cover is assembled, and at the same time, the Breakwire is pulled taut. The Breakwire is soldered to the Terminals (2) (Machine #10A).

- The Breakwire (.004 insulated copper wire) is broken when the Bobbin thread pulls on the Bobbin Post. The electronics of the device detects the change in the circuitry which causes the mine to detonate.
- The Retaining Ring captivates the Breakwire in the 'V' slot of the Bobbin. Its large diameter will not permit the Bobbin Post to leave the Tube, Housing.
- The Post Cover seals the opening above the Bobbin Post. The Cover fits into the cavity of the Diaphragm Plate. The Post Cover, when seated, will pull the Breakwire taut, which gives uniformity to Breakwire function.



# SEQUENCE OF EVENTS TO FUNCTION SENSOR

1. Gas pressure enters diaphragm plate through port in fill tube.
2. Pressure moves diaphragm downwards moving ring lock, which in turn flattens (2) spring washers (Belville shaped springs).
3. As the ring lock is moving it cams the two post locking balls inwards, holding the post of the bobbin assembly.
4. The ejection spring and booster spring then pushes sleeve, cap anchor and bobbin core and thread free of the sensor case, and deploys approximately 40 feet of tripline.
5. The post locking balls will hold the post firmly until the pressure leaks off and the spring washers will push the ring lock to its original at rest position, releasing the post which is then free to move and is attached to the .004 breakwire.

Figure A-12. Sensor Case and Bobbin Assembly

MACHINE #13A  
EPOXY APPLICATION TO THE SENSOR ASSEMBLY

PARTS

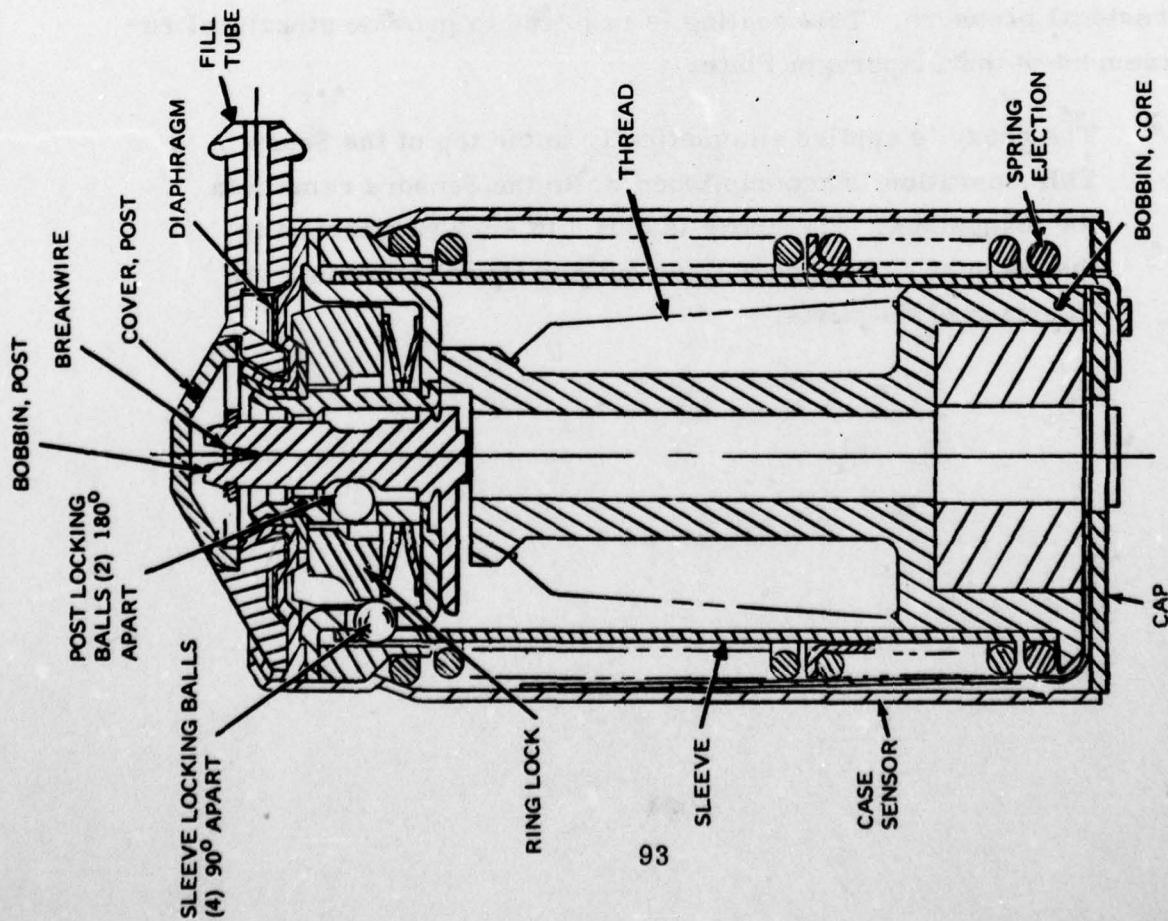
Epoxy

PURPOSE OF THE ASSEMBLY

The application of epoxy to the top of the Sensor Assembly is to seal the interface between the Post, Cover, and the Diaphragm Plate (Figure A-13 and A-14). In addition to sealing the interface between the Diaphragm and the Diaphragm Plate 360°, the epoxy is applied and the cure of the epoxy is effected without any residual pressure. This sealing is required to provide structural reinforcement of the Diaphragm Plate.

- The epoxy is applied automatically to the top of the Sensor. This operation is accomplished while the Sensors remain in the magazines. The epoxy is cured in an oven attached to the machine. The epoxy thins out and flows freely into the interface of the parts.





## SEQUENCE OF EVENTS TO FUNCTION SENSOR

1. Gas pressure enters diaphragm plate through port in fill tube.
2. Pressure moves diaphragm downwards moving ring lock, which in turn flattens (2) spring washers (Belville shaped springs).
3. As the ring lock is moving it cams the two post locking balls inwards, holding the post of the bobbin assembly.
4. The ejection spring and booster spring then pushes sleeve, cap anchor and bobbin core and thread free of the sensor case, and deploys approximately 40 feet of tripline.
5. The post locking balls will hold the post firmly until the pressure leaks off and the spring washers will push the ring lock to its original at rest position, releasing the post which is then free to move and is attached to the .004 breakwire.

Figure A-12. Sensor Case and Bobbin Assembly

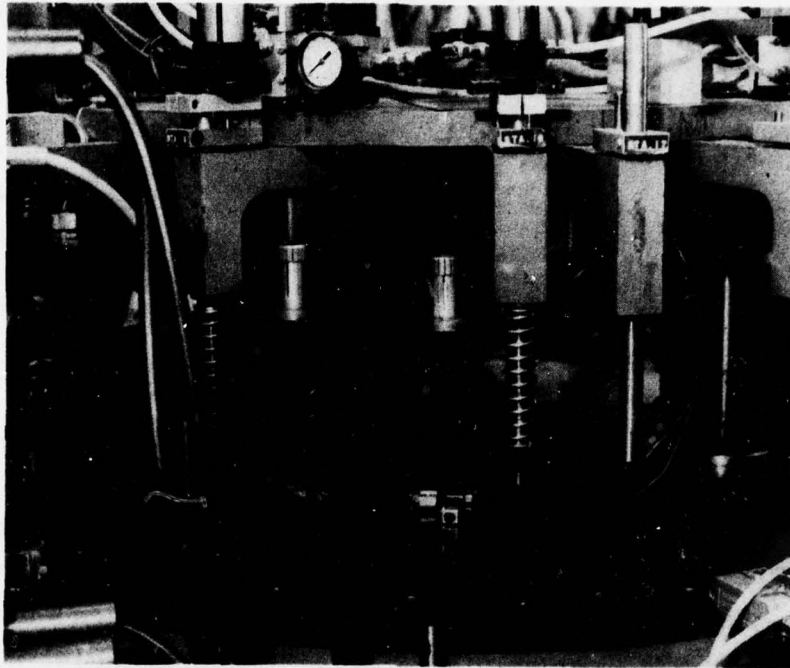


Figure A-13. ADAM Sensor Assembly Machine

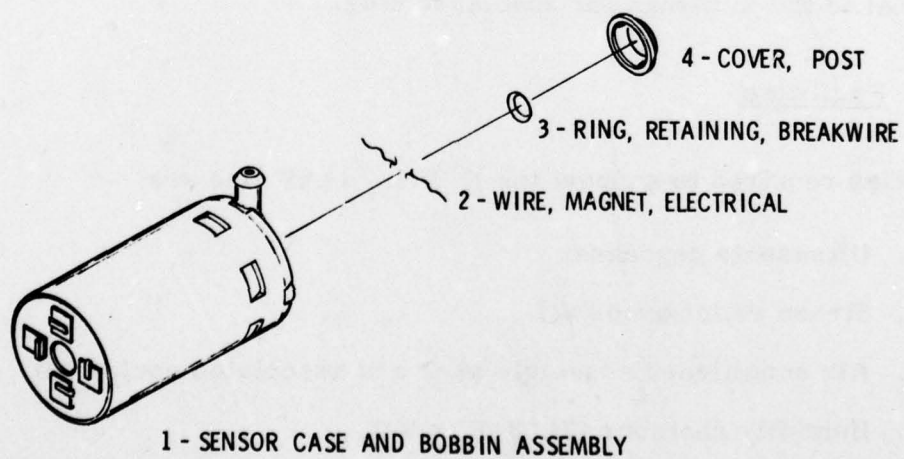


Figure A-14. XM74 Sensor Assembly Machine

### 3.5 LOGISTICS

#### 3.5.1 Support

Critical components for repair or replacement are stocked using a "Min/Max" Tooling System.

The responsible engineer assigned to a SAAF establishes a Min/Max Tooling List for all critical pieces of special tooling which are subject to wear or being damaged. He then orders the maximum quantity to be held in the Tool Handling Crib. As the tooling is used and the minimum quantity is then on hand, the crib attendant notifies the engineer, who places an order for replacement of special tooling to meet the maximum quantity.

The engineering crew establishes min/max quantities on OML parts which are common among all the SAAF's. The reorder of these components is identical to the min/max for special tooling.

#### 3.5.2 Facilities

Facilities required to support the E.T.L. SAAF line are:

1. Ultrasonic degreaser
2. Stress relief ovens (2)
3. Air conditioned assembly area and associated equipment
4. Humidity chamber (2), 8 ft. x 8 ft.
5. Demagnetizer
6. Epoxy mixing equipment

Description of facilities required:

1. Ultrasonic degreaser  
Baron Blakeslee MRS-120  
480V 1 Ø 60-cycle



## 2. Stress relief ovens (2)

Dispatch oven V 35 HP  
440V 3 Ø 60-cycle

## 3. Air conditioned assembly area and associated equipment

No preference

## 4. Humidity chambers (2)

Chamber by contractor  
Humidity equipment - no preference

## 5. Demagnetizer

Electromatic - LAC - 15 VB  
Belt feeding  
460V AC 1 Ø 50-60 cycle

## 6. Epoxy mixing equipment

3.5.3 Utility Consumption

SAAF No.	Electric Power	Compressed Air	Fuel Oil	Gas	Water	Steam
7	2.29	950	-	-	-	-
8	2.29	950	-	-	-	-
9	2.29	950	-	-	-	-
10	2.44	950	-	-	-	-
10A	2.44	950	-	-	-	-
11	2.51	950	-	-	-	-
12	2.67	950	-	-	-	-
12A	.42	950	-	-	-	-
12B	2.67	950	-	-	-	-
12C	.42	950	-	-	-	-
13	13.72	950	-	-	-	-
13A	2.44	950	-	-	-	-
TOTAL	36.60	11,400				
UNIT	KWH/M	CFH/M				

AD-A063 424

HONEYWELL INC HOPKINS MN DEFENSE SYSTEMS DIV

F/G 19/1

PRODUCIBILITY ENGINEERING AND PLANNING (PEP) OF THE XM74 GEMSS --ETC(U)

AUG 78 R FAIRCHILD

DAAK10-77-C-0047

UNCLASSIFIED

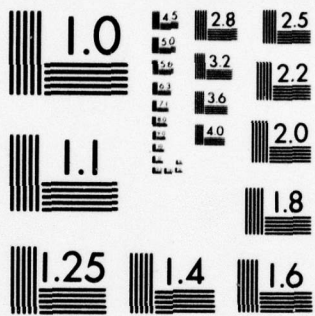
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2 OF 3

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MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS-1963-A



The data listed above is from a 1977 utility usage analysis (Figure A-15) for comparable machines used on the ADAM program at Honeywell's Twin City Arsenal, Building 103, New Brighton, Minnesota.

#### 3.5.4 Personnel

Personnel requirements listed below show the number and type of personnel required for the specified mobilization rate of 200,000 sensors per month. These requirements are also based upon a self-contained, stand alone, mature manufacturing line. Concurrent operation of other lines at the same location will reduce these requirements.

Program Management:	Program Manager	1.0
	Quality Manager	1.0
Engineering Supervision:	Production	1.0
	Quality	1.0
Engineering:	Production Engineers	8.0
	Quality Engineer	1.0
	Quality Technician	1.0
	Evaluation Engineer	1.0
Manufacturing:	Assistant Foreman	1.0
	Machine Operators	20.0
	Group Leader	2.0
Tool Room:	Toolmaker	5.0
Inspection:	Receiving Inspection	1.0
	Floor Inspector	3.0
	Floor Group Leader	1.0
Evaluation	Technician	.5
Chem & Met Lab	Engineer	.5
	Chem/X-ray Technician	2.0
Plastics Lab	Engineer	.2
	Technician	1.0





# UTILITY USAGE ANALYSIS

1977

ADM

PAGE 2 OF 3

103. M77

N. H. D. D. DESCRIPTION G. INTER.	PLANT NUMBER	UNIT NO.	USAGE	HRS/M	AVG. HRS/M	KW/M	APV. L./H	BAM/M	CF/M	CUB/M	LBS. STEAM/M	LBS. STEAM/M	WATER GAL/M	WATER GAL/M
BUILDING 103														
Quens														
G6172 (41)	28110015 TAR	040A		11.76		45.72								
G6205 (45)	28104652-002	040A		4.97										
G6203 (46)														
G6234 (47)														
G6170 (Pre-Heat)														
G6177	28110015 TAR	040A		5.49		115.76		45.72						
G6212	28104652-002	040A		2.75		30.48		30.48						
G6152	28104438-001	060A		6.85		45.72		133.20						
	28108853-001	060A		16.66		10.00		45.72						
	28109882-001	035A		2.22		10.00		10.00						
	28109305-001	020A	(2)	4.44		10.00		10.00						
	28106229-003	060A		4.44		10.00		10.00						
	28107973-001	030A		4.44		10.00		10.00						
	28107916-001	040A		4.44		10.00		10.00						
G30024	28102768-001	020A		4.44		10.00		10.00						
G30025	28102367-001	020A		4.44		10.00		10.00						
	28102768-001	020A		4.44		10.00		10.00						
	28109378-001	020A	(1)	4.44		10.00		10.00						
	28109378-001	030A	(1)	4.44		10.00		10.00						
G31401	28111574-001	020A		4.44		10.00		10.00						
G31403	28110607-001	070A		4.44		10.00		10.00						
Energy Dip & Curve System														
G6148	28102724-001	040A		1.00		171.45								
	28102718-001	010A		1.00		171.45								
	28102767-001	010A		1.00		171.45								
	28102746-001	010A		1.00		171.45								
	281029514-001	010A		1.00		171.45								
Degreasers	281029514-001	030A		1.00		171.45								
G6169 (Pre-Heat)	28110015 TAR	060A		1.00		171.45								
G6149	28104438-001	020A		1.00		171.45								
	28110014-003	050A		1.00		171.45								
	28106829-003	090A		1.00		171.45								
Burner/Blowdown	28111572-001	080A		1.00		171.45								
(Handling with 1/2" 200 S.C.)	28102458-001	080A		1.00		171.45								
	28102458-001	080A		1.00		171.45								
Swaption														
G6167	28102458-001	010A		1.00		171.45								
	28102458-001	010A		1.00		171.45								
Hydraulic Press														
G6166	28110003-001	010A		1.00		171.45								
Bolt Sander														
G6174	28110015 TAR	535A		1.00		171.45								



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[illegible]

#### 4.0 TEST/VERIFICATION

Test and verification of the GEMSS Extended Tripline Sensor Special Automatic Assembly Line shall be accomplished by conducting a demonstration test on each Special Automatic Assembly Fixture (SAAF). Demonstration Test Plans shall be prepared for each SAAF to assure that SAAF requirements are met and that Extended Tripline Sensor Assemblies conform to MIL-S-48755.

- a. During the conduct of the demonstration test, Reliability, Availability and Maintainability (RAM) data shall be recorded by failure indicating:

- Cause of failure
- Corrective action
- Time to determine corrective action
- Time to perform corrective action
- Total down time per failure

This data will be part of the reliability analysis.

- b. During the conduct of the demonstration test, the following shall be recorded:

- Date of demonstration test
- Start time
- Stop time (Stops will be recorded by number displayed and/or reason for stop and duration of stop)
- Observed time
- Total run time
- Total down time

- Cycle rate (CPM0
- Total cycle count
- Total parts count
- Lot number of parts and material assembled

This data will be part of the engineering evaluation.

- c. During and/or following the demonstration test, samples of SAAF-accepted assemblies shall be selected and inspected and/or tested for conformance to MIL-S-48755. The inspection results will be part of the Quality evaluation.
- d. Installation testing and checkout shall consist of the following:
  - All feed, work and probe stations function properly.
  - Control system functions properly.
  - SAAF-accepted assemblies conform to MIL-S-48755



TS 578-4869

APPENDIX B

EQUIPMENT TECHNICAL DATA PACKAGE SPECIFICATION  
FOR  
SPECIAL AUTOMATIC ASSEMBLY LINE  
FOR XM74 GEMSS EXTENDED TRIPLINE SENSOR

SPECIFICATION TS 578-4869

DARCOM Project  
578-4869

AUGUST 1978

Preparing Organization  
Honeywell Inc.  
Defense Systems Division  
94580

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## 1.0 SCOPE

### 1.1 Equipment Data Specification

This Equipment Technical Data Specification establishes the performance, design, test, manufacture and acceptance requirements for Special Automatic Assembly Fixtures (SAAF) used in the fabrication of the XM74 GEMSS Extended Range Tripline Sensor.

This equipment is tabulated (ref: Table B-1) to show each machine by name and number. A listing of parts assembled and commonality with ADAM Sensor Assembly Machine is also given.

## 2.0 APPLICABLE DOCUMENTS

### 2.1 Document Specification

The following documents, of the issue in effect on the date of beginning activity on this contract, shall form a part of this specification to the extent specified herein:

#### Standards

MIL-STD-1472

Human Engineering Design Criteria for Military Systems, Equipment and Facilities.

#### Drawings

9292972

Extended Range Tripline Sensor

#### Specifications

QAP - GEMSS-1

Extended Range Tripline Sensor

TABLE B-1. MACHINE TABULATION

Machine Number	Machine Name	Parts Assembled		Commonality
7	Bobbin Assembly 9292982	9292983	Weight	New
		9292985	Bobbin	
		9298591	Tape	
		9298592	Thread	
8	Diaphragm Assembly 9292998	9298597	Diaphragm Plate	Common
		9298598	Diaphragm	
		9298599	Diaph. Eyelet	
		9298618	Fibrous Gasket	
9	Housing Assembly 9298587	9298588	Housing, Cup	Common
		9298589	Housing, Tube	
10	Release Mech. Assy. 9292991	9298582	Washer Spring	Common
	Less Terminals	9298586	Ring, Ball Lock	
		9292998	Diaphragm Assy.	
		9298587	Housing Assy.	
10A	Terminal Assy. 9292991	9298601-1	Terminal, Breakwire	Common
	(Release Mech. Assy.)	9298601-2	Terminal, Breakwire	
11	Release Mech. and Case Assembly 9292986	9292987	Sensor Case	Modified
		9292988	Sleeve	
		9292989	Case, Washer	
		9298581	Locking Ball	
		9292991	Release Mech. Assy.	
12B	Release Mech. and Case Assembly	9292981	Interface Eyelet	New
		9292990	Booster Spring	
		9298578	Ejection Spring	
		9292986	Case/Sleeve Assy.	
12	Sensor Case/Bobbin Assembly 9292976	9292980	Anchor	Modified
		9298579	Cap	
		9298581	Locking Ball	
		9292986	Release Mech./Case Assy.	
		9292982	Bobbin Assembly	
13	Sensor & B/W Assy.	9298575	Post, Cover	Modified
		9298576	Magnet Wire	
		9298577	B/W Retaining Ring	
		9292976	Sensor Case/Bobbin Assy.	
13A	Epoxy Dispenser 9292972	9298617	Adhesive	Modified
	Extended Range		Spec. MMM-A-132	
	Tripline Sensor		Type 1, Class 3	

## 2.2 NON-GOVERNMENT DOCUMENTS - NOT APPLICABLE

## 3.0 REQUIREMENTS

### 3.1 Item Definition

The XM74 (Extended Tripline) automated machine line described herein is a group of unique machines that result in a complete and acceptable XM74, Extended Range Tripline Sensor.

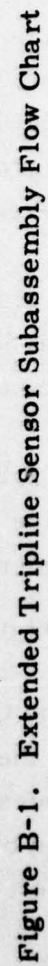
3.1.1 Item Diagrams -- Figure B-1 shows the total machine line flow of the Sensor Assembly. Following this diagram are dial layouts and descriptions indicating the functions performed on each machine.

3.1.2 Interface Definition -- The SAAF for the GEMSS Sensor Assembly Line will consist of 10 Special Built Assembly Fixtures. They will run independent of each other. The controlling factor will be availability of parts and assemblies from previous SAAF's.

The GEMSS Sensor Line will run at various rates (assemblies/per hour) due to the complexity of the various assemblies, therefore, a comparison of hours per month multiplied by the expected RAM rate predicted (PCS/HR) has been compiled to identify the specific functional interface requirements (Ref: Table B-2).

Physical interface relationships are specified on tooling drawings for the individual SAAF Station, Base, Dial, or Tooling Plate. These drawings will be furnished with the SAAF's in almost all instances. Special drawings are required for each station and probe.





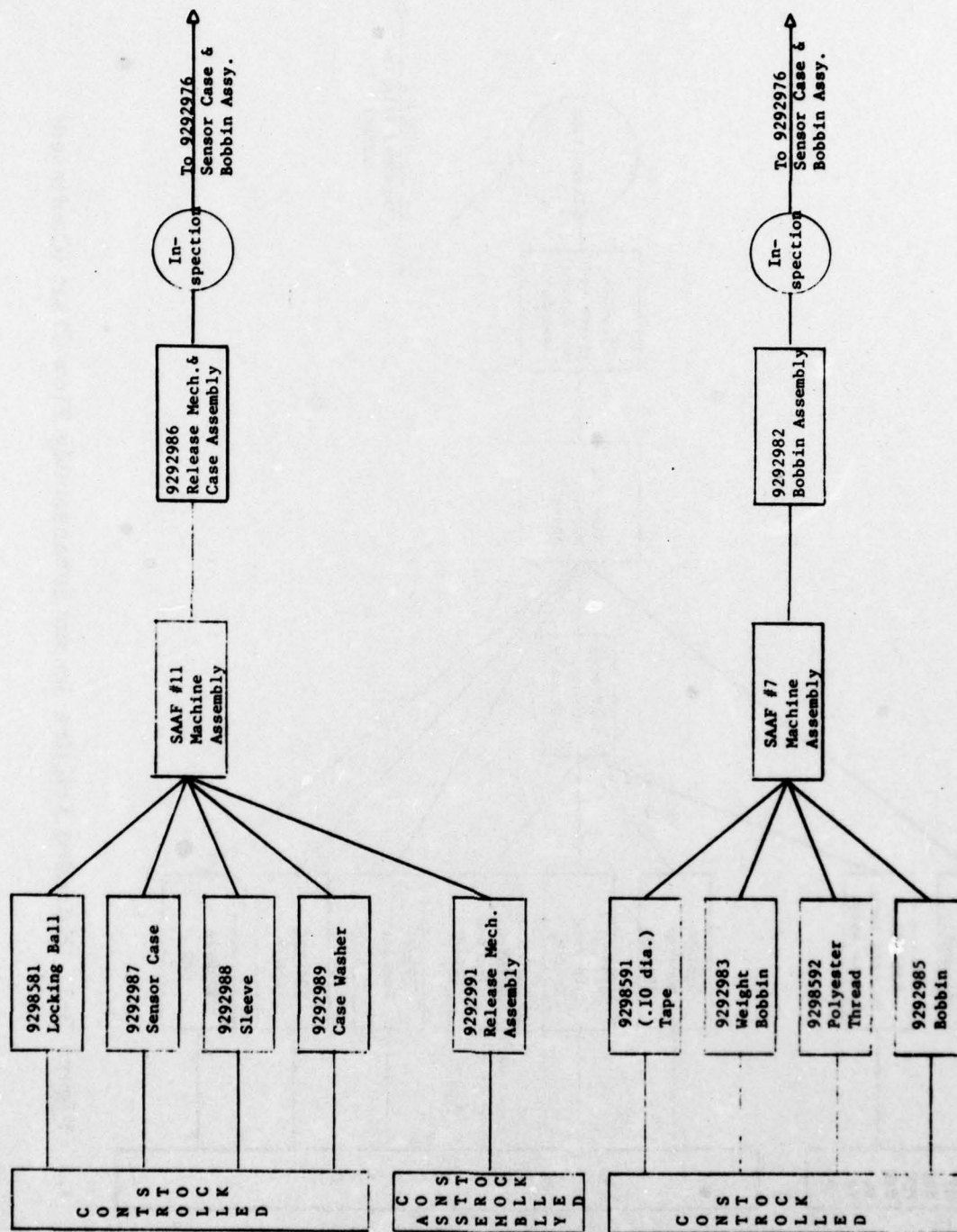


Figure B-1. Extended Tripline Sensor Subassembly Flow Chart (Continued)

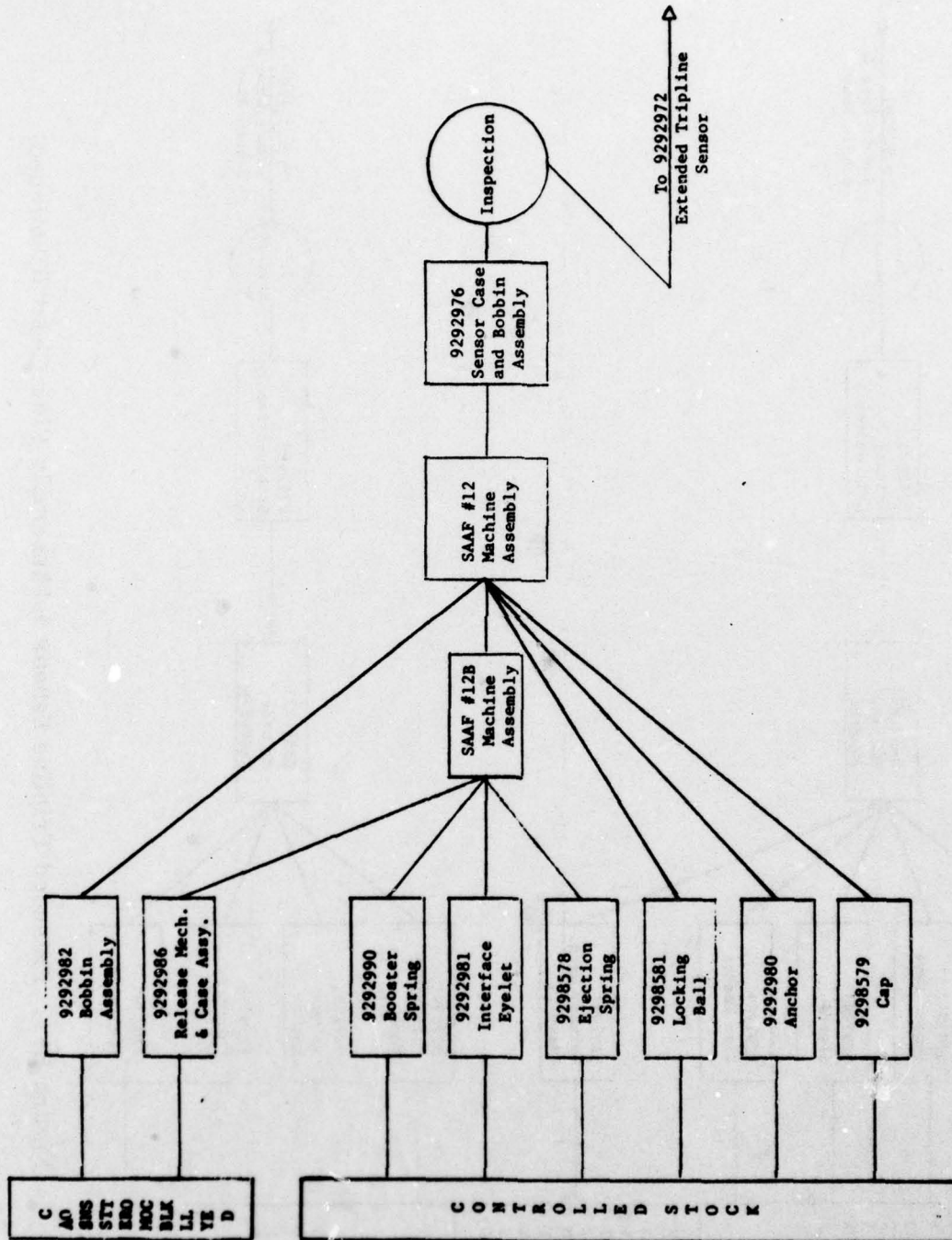


Figure B-1. Extended Tripline Sensor Subassembly Flow Chart (Continued)



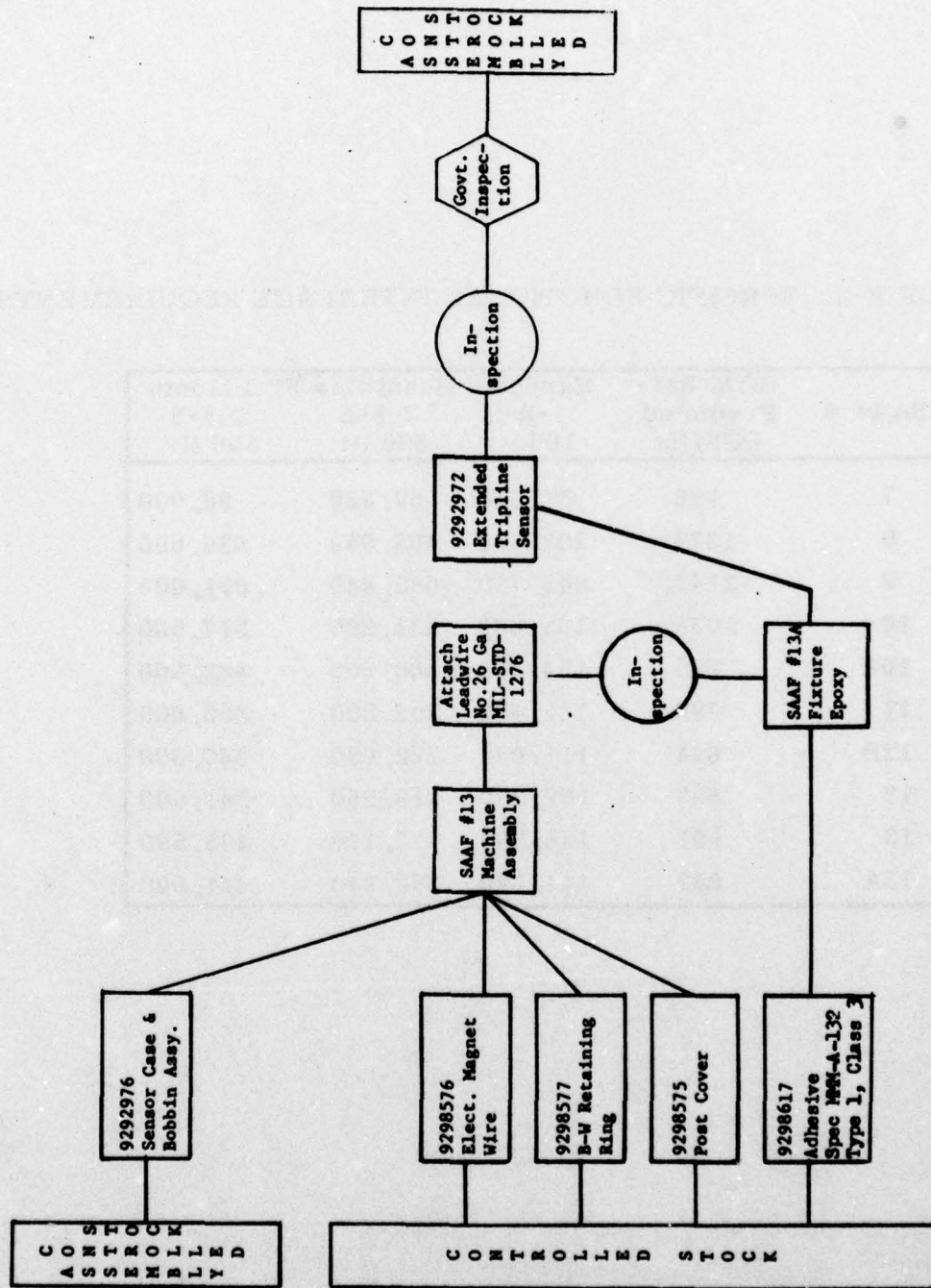


Figure B-1. Extended Tripline Sensor Subassembly Flow Chart (Concluded)

TABLE B-2. SPECIFIC FUNCTIONAL INTERFACE REQUIREMENTS

SAAF #	RAM Rate Predicted PCS/Hr	Expected Quantities PCS/Month		
		1-8-5 160 Hr	2-8-5 320 Hr	3-8-5 500 Hr
7	196	31,360	62,720	98,000
8	1272	203,520	407,040	636,000
9	2142	342,720	685,440	1,071,000
10	1035	165,600	331,200	517,500
10A	965	154,400	308,800	482,500
11	790	126,400	252,800	395,000
12B	694	111,040	222,080	347,000
12	683	109,280	218,560	341,500
13	991	158,560	317,120	495,500
13A	882	141,120	282,240	441,000

As much as possible, the contractor should use standard off the shelf items for the electrical and pneumatic systems.

3.1.3 Major Component List -- Honeywell Ordnance Machine Lab (OML) Automation Equipment is designed and built to standardized configurations whenever possible. The Machine Development Laboratory has designed and built more than 1100 automatic assembly machines in recent years and has available extensive tools and equipment to support machine build programs. The magnitude of OML operations has permitted standardization of special machines to a high degree. Enclosed as an attachment to this specification is a booklet entitled Automation Standard Equipment. This booklet is a catalog of major components which have also been referenced on the design drawings for the assembly machines to be used for building the XM74 GEMSS Extended Range Tripline Sensor.

3.1.4 Government Furnished Property List -- With respect to the XM74 GEMSS Sensor, no Government-furnished equipment is required specifically for the Automatic Assembly Line. Due to similarities between the ADAM and GEMSS lines there will be common usage of the following facility items:

- Demagnetizer
- Ultrasonic Cleaner-Degreaser
- Humidity Controlled Room

These facilities would be placed between the two lines.

### 3.2 Characteristics

3.2.1 Performance Characteristics -- Special automatic assembly fixtures will produce 750 to 1250 completed assemblies per hour, depending upon the complexity of the assembly. Machine descriptions are given on the following pages.



**MACHINE #7  
BOBBIN WINDER**

**APPLICABLE PIECE PARTS**

- |                   |                |
|-------------------|----------------|
| • Bobbin          | Print #9292985 |
| • Weight          | Print #9292983 |
| • Tape            | Print #9298591 |
| • Thread          | Print #9298592 |
| • Bobbin Assembly | Print #9292982 |

**MACHINE OPERATIONS**

This machine (Figure B-2) is required to assemble the bobbin and weight with 46 feet of thread wrapped around the bobbin. The thread is held on the face of the bobbin by an adhesive tape and maintained in the rear by a press fit between the bobbin and weight. The sequence of machine operations is as follows:

1. Bobbin weights are placed onto the nest and positioned.
2. Bobbin is placed onto the nest in front of the weight.
3. The thread is wound onto the bobbin with 1/2 turn located in the weight cavity of the bobbin and with 46 feet along the body. A final 3 wraps is done on the bobbin post.
4. The thread is taped into position on the bobbin face. This tape is sealed by a heating operation.



\_\_\_\_\_

Figure B-2. Machine #7 Bobbin Winder

4 3 2 1  
DON'T FORGET SAFETY

STATION NO.	OPERATION	PART NO.
1	FEED WEIGHTS BOTH NESTS	9292983
2	PROBE WEIGHTS PRES.&POS. BOTH NESTS	
3	IDLE	
4	IDLE	
5	FEED BOBBIN LOWER NEST	9292985
6	PROBE BOBBIN PRES.&POS. LOWER NEST	
7	IDLE	
8	IDLE	
9	FEED BOBBIN UPPER NEST	9292985
10	PROBE BOBBIN PRES.&POS. UPPER NEST	
11	WIND THREAD BOTH NESTS	9292984
12	TAPE STATION BOTH NESTS	9293003
13	HEAT SEAL STATION BOTH NESTS	
14	PRESS WEIGHT TO BOBBINE PROBE - BOTH NESTS	
15	CUT THREAD	
16	CUT THREAD	
17	EJECT TO CONTAINER BOTH NESTS	
18	IDLE	
19	IDLE	
20	IDLE	
21	IDLE	
22	PROBE EMPTY NEST BOTH NESTS	
23	IDLE	
24	IDLE	

REV. DESIGNATOR	REV.	REQ.	NAME	FINISHED SIZE	DESCRIPTION	MATERIAL	HEAT TRY.	PH.
TOLERANCES UNLESS NOTED			SCALE	DATE	Honeywell			
ONE P.L.C. SEC. + 1.000			TL. DESK	8-15-78	TOP LAYOUT			
TWO P.L.C. SEC. + 2.000			DRAWER		BOBBIN ASSEMBLY			
THREE P.L.C. SEC. + 3.000			CHECKED		SAAF #7			
FOUR P.L.C. SEC. + 4.000			DATE	4-19-78	DRAWING NO. XM74			
REV. 0000			REV.	1	77-9610-AE			



TS 578-4869

5. The nest is compounded to press the weight into the weight cavity maintaining the thread in the rear of the bobbin.
6. Excess thread in the operation is removed.
7. The final Bobbin Assembly is ejected off the machine in bulk.

MACHINE #8  
DIAPHRAGM ASSEMBLY

APPLICABLE PIECE PARTS

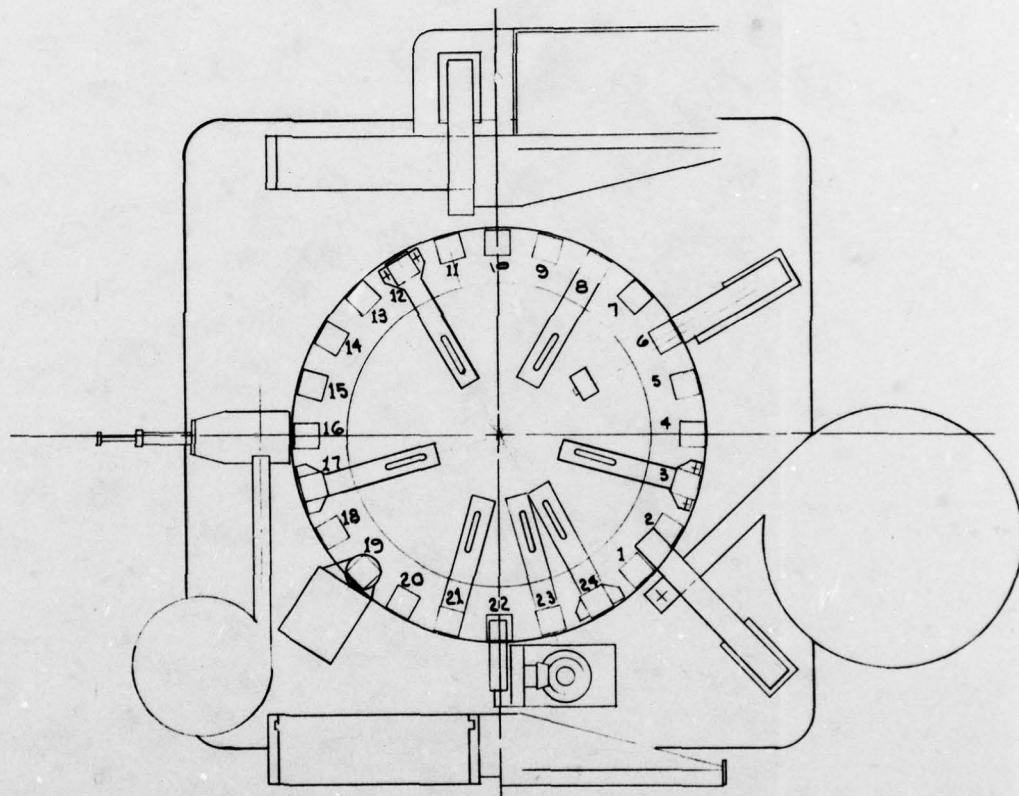
- |                      |                |
|----------------------|----------------|
| • Fibrous Gasket     | Print #9298618 |
| • Eyelet Diaphragm   | Print #9298599 |
| • Diaphragm          | Print #9298598 |
| • Plate Diaphragm    | Print #9298597 |
| • Diaphragm Assembly | Print #9292998 |

MACHINE OPERATIONS

This machine (Figure B-3) is required to assemble an eyelet, diaphragm, plate diaphragm and gasket to form the diaphragm assembly. This completed assembly is fed as a unit to Machine #10. The sequence of machine operations is as follows:

1. First, the plastic diaphragm plate is fed to the nest from a vibratory bowl.
2. Now, a gasket is blanked from a continuous roll of material and placed into position on the diaphragm plate.
3. Next, the diaphragm is fed from magazines to the nests. It is positioned on top of the two previous piece parts.

STEVE TOWARD ERROR-FREE PERFORMANCE



OML 452

DATE	18-1471	CHAS. T. BROWN (REDUCED)	W. C. L.
DATE		ORIGINAL DRAWING T.N. WAS 77-3607-AC	
DATE		REVISIONS	TL. J. B. BROWN

OF 17C 5/78

Figure E-3. Machine #8 Diaphragm Assembly



4 3 2 1  
DON'T FORGET SAFETY

STN No.		
1	HOPPER FEED DIAPHRAGM PLATE	
2	IDLE	D
3	PROBE PRESENCE & POSITION	
4	IDLE	
5	IDLE	
6	BLANK FORM & FEED GASKET	
7	IDLE	
8	PROBE OPTICALLY	
9	IDLE	
10	MAGAZINE FEED DIAPHRAGM	C
11	IDLE	
12	PROBE PRESENCE & POSITION	
13	IDLE	
14	IDLE	
15	IDLE	
16	HOPPER FEED EYELET	
17	PROBE PRESENCE & POSITION	
18	IDLE	
19	STAKE EYELET	B
20	IDLE	
21	PROBE STAKED EYELET	
22	EJECT TO MAGAZINE	
23	REJECT	
24	PROBE EMPTY NEST	A

OML 452 BASE

REF. DESIGNATOR	QTY.	REQ.	NAME	FINISHED SIZE	DESCRIPTION	MATERIAL	HEAT TRT.	FIN.
			TOLERANCES UNLESS NOTED	SCALE	DATE	Honeywell		
				TL. 0000	10-3-77	TOP LAYOUT		
			DETAILER			NAME		
			CHECKED			SENSOR DIAPHRAGM ASSY SAAF		
			BY	JHM	3-14-78	ADAM SAAF		
			CHKD			SHEET NO. 1		
			D.C.			1 1 D H28109250-T8-1		
REF. 0000.								

TS 578-4869

4. Then an eyelet is fed from a vibratory bowl. This eyelet is crimped to hold these piece parts together.
5. Following the crimping operation, the assembly goes through a notching station where the diaphragm is notched to provide clearance for the spout.
6. The final station is the point where the assembly is ejected to a magazine.

MACHINE #9  
HOUSING ASSEMBLY

APPLICABLE PIECE PARTS

- Tube Housing                      Print #9298589
- Cup Housing                        Print #9298588
- Housing Assembly                Print #9298587

MACHINE OPERATIONS

Two piece parts are involved in this assembly (Figure B-4). The tube and cup housing are assembled to form the housing assembly. The sequence of machine operations is as follows:

1. First, the housing tube is fed from a vibratory bowl onto the nest.
2. Next, the housing cup is fed from a vibratory bowl and positioned on top of the housing tube.
3. The two pieces are fixed together by a staking operation.
4. This final assembly is now ejected in bulk.



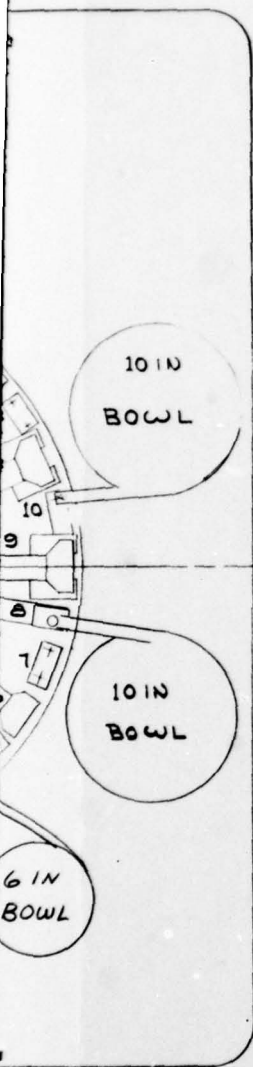


FRONT

A

Figure B-4. Machine #9 Housing Assembly

3  
DON'T FORGET SAFETY



- |    |                           |
|----|---------------------------|
| 1  | HOPPER ORIENT & FEED TUBE |
| 2  | IDLE                      |
| 3  | PROBE PRESENCE & POSITION |
| 4  | HOPPER ORIENT & FEED TUBE |
| 5  | IDLE                      |
| 6  | PROBE PRESENCE & POSITION |
| 7  | IDLE                      |
| 8  | HOPPER FEED CUP           |
| 9  | PROBE PRESENCE & POSITION |
| 10 | HOPPER FEED CUP           |
| 11 | PROBE PRESENCE & POSITION |
| 12 | IDLE                      |
| 13 | IDLE                      |
| 14 | IDLE                      |
| 15 | ORIENT CUPS TO TUBES      |
| 16 | IDLE                      |
| 17 | IDLE                      |
| 18 | IDLE                      |
| 19 | PROBE ORIENT POSITION     |
| 20 | STAKE TUBE                |
| 21 | IDLE                      |
| 22 | PROBE STAKE               |
| 23 | PROBE ORIENT POSITION     |
| 24 | STAKE TUBE                |
| 25 | IDLE                      |
| 26 | PROBE STAKE               |
| 27 | IDLE                      |
| 28 | IDLE                      |
| 29 | EJECT TO CONTAINER (2)    |
| 30 | IDLE                      |
| 31 | REJECT TO CONTAINER       |
| 32 | PROBE EMPTY NEST          |

OML 451 BASE

REF. DES. CREATED	REV.	REV.	NAME	FINISHED DATE	DESCRIPTION	MATERIAL	REPT. TOT. PUL
			TOLENCANES UNLESS NOTED	SCALE TL. 1/8"	DATE 10-24-77	Honeywell	
			WORK 1/2" FORMING 1/2"			TOP VIEW LAYOUT	
			ONE P.L.S. DES. - 1.000			HOUSING ASSY	
			TWO P.L.S. DES. - 1.000			ADCS S&AT 9	
			THREE P.L.S. DES. - 1.000				
			FOUR P.L.S. DES. - 1.000				
REF. 0000.							

2

# MACHINE #10 RELEASE MECHANISM ASSEMBLY

## APPLICABLE PIECE PARTS

- Spring, Washer                      Print #9298582
- Ring Lock                              Print #9298586
- Housing Assembly                    Print #9298587
- Diaphragm Assembly                Print #9292998

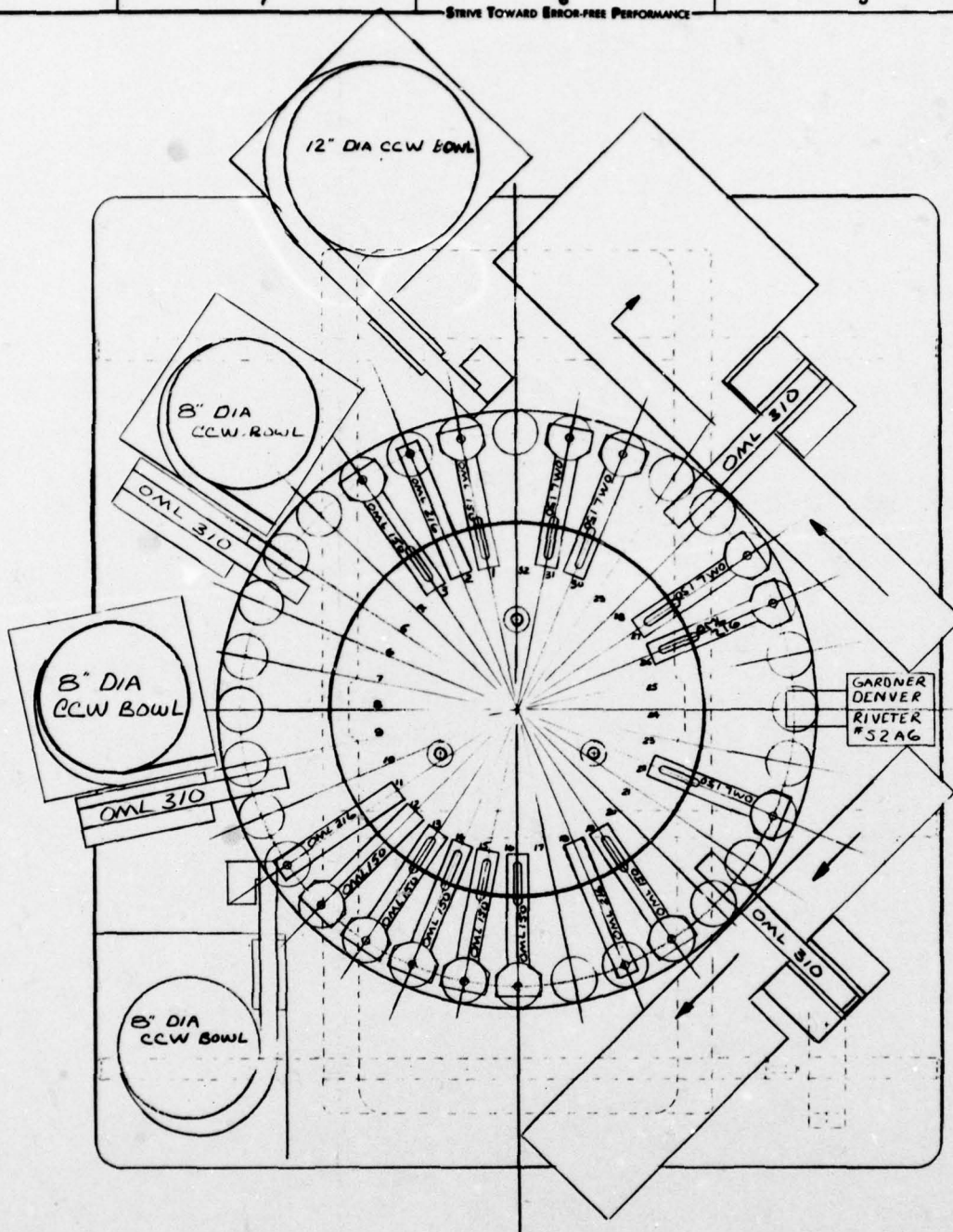
## MACHINE OPERATIONS

This machine is required to assemble the housing assembly (from Machine #9) and diaphragm assembly (from Machine #8) along with two washer springs and a lock ring (Figure B-5). This entire assembly is the release mechanism without the breakwire terminals. The sequence of assembly machine operations is as follows:

1. First, the housing assembly is fed from a vibratory bowl. It is oriented in the nest by a spin station.
2. Next, two springs are fed into the housing assembly. These are also fed from a vibratory bowl.
3. Then a ring lock is positioned on top of the two springs and three separate stations exercise the springs.
4. Now, the springs are compressed and nest slides are used to captivate the assembly.



STREVE TOWARD ERROR-FREE PERFORMANCE



DATE	10/1/77	ADD SECTION 2.6 CHANGE 27	WJG
DATE	9/20/77	ORIGINAL DRAWING T.N. WAS 77-3607-AE	N/JN
DATE		REVISIONS	TL DESIGNER

OF 17C 3/76

Figure B-3. Mechanism Release Mechanism

4 3 2 1  
DON'T FORGET SAFETY

STATION NO.	OPERATION
1	FEED HOUSING ASSEMBLY
2	ORIENT HOUSING
3	PROBE HOUSING ORIENT
4	IDLE
5	FEED SPRING WASHER
6	IDLE
7	IDLE
8	IDLE
9	FEED SPRING WASHER
10	IDLE
11	FEED BALL LOCK
12	PROBE BALL LOCK
13	EXERCISE BALL RING
14	EXERCISE BALL RING
15	EXERCISE BALL RING
16	PROBE
17	IDLE
18	ORIENT BALL RING, CLOSE NEST
19	PROBE BALL RING (POSITION)
20	FEED DIAPHRAGM
21	IDLE
22	PROBE DIAPHRAGM
23	IDLE
24	CRIMP
25	IDLE
26	PROBE CRIMP
27	OPEN NEST
28	EJECT
29	IDLE
30	REJECT
31	PROBE EMPTY NEST
32	IDLE

OML 451 BASE

REF. DESIGNATOR	DET. REQ.	NAME	FINISHED SIZE	DESCRIPTION	MATERIAL	HEAT TMT.	FIN.
MASTER DRAWING 8-10-1959 DEC. 2, 1976	TOLEANCES UNLESS NOTED	SCALE 1/4 X TL DASH N/A	DATE 3/20/77	Honeywell TOP LAYOUT			
	ONE P.L. DEC. +2.000	CHECKED RPN.		SENSOR RELEASE MECH. ASSY.			
	TWO P.L. DEC. +2.000	APPROVED RPN.		ADAM			
	THREE P.L. DEC. +2.000	END RPN.		SAAF # 10			
REF. QWGS.	FOUR P.L. DEC. +2.000	D.C.		SHEET NO. 1	NO. SHEETS 1	SIZE D	H28009650-T13-12

OML 451 MASTER

5. The diaphragm assembly is fed from magazine and placed on top of the nest.
6. The diaphragm assembly is staked to the remainder of the assembly and the nest slides are retracted. Following this the assembly is ejected to a magazine.



**MACHINE #10A**  
**TERMINAL ASSEMBLY FIXTURE**

**APPLICABLE PIECE PARTS**

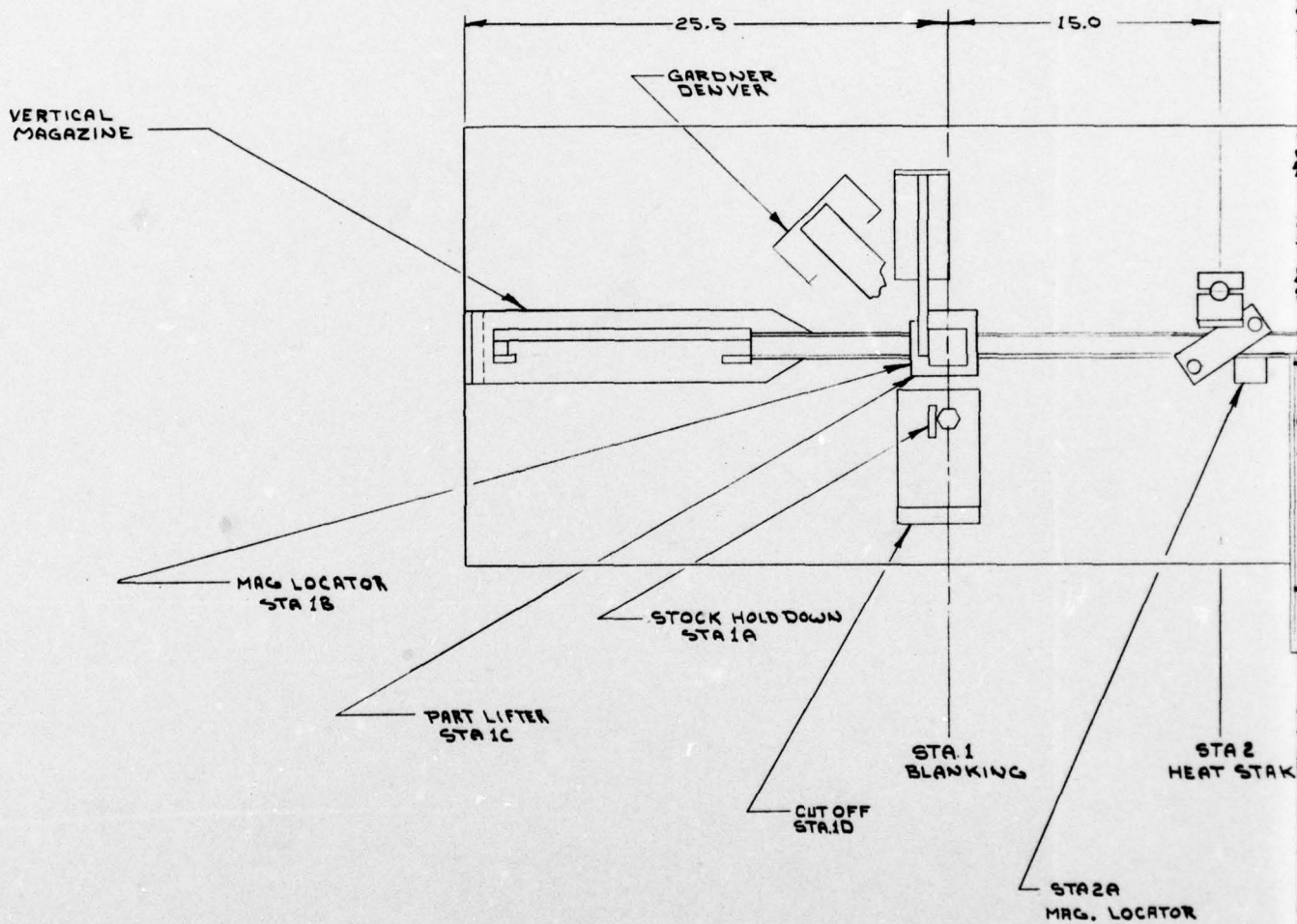
- Terminal, Breakwire                      Print #9298601
- Release Mechanism Assembly              Print #9292991

**MACHINE OPERATIONS**

This fixture (Figure B-6) assembles the terminals (left and right side) to the release mechanism. The sequence of machine operations is as follows:

1. The magazines containing the release mechanisms are utilized as a nest. The fixture indexes the magazines through the assembly operations.
2. Next, breakwire terminals are punched from strip stock and positioned onto the release mechanism.
3. Finally, these terminals are held in place by heating and forming two plastic ears on the release mechanism.

STRIVE TOWARD ERROR-FREE PERFORMANCE



DATE	ENG	D.C.	REV	REVISIONS	TL DESIGNER
11-77					
THIS SHEET REPLACES ORIGINAL					
ORIGINAL DRAWING T.N. WAS 77-9607-AP					

Figure B-6. Machine #10A Terminal Assembly

4 3 2 1  
DON'T FORGET SAFETY

15.0

9.0

STA 2  
HEAT STAKE

STA 2A  
MRG. LOCATOR

NOTE:  
UNLESS OTHERWISE SPECIFIED:  
BREAK UNNECESSARY SHARP CORNERS  
APPROXIMATELY 1/32".  
SCREWS, SPRINGS, AND DOWELS TO SUIT.  
STAMP IDENTITY ON ALL TOOL STEEL  
ITEMS. (A.J.S. TYPE OR BRAND NAME).  
PROTECTED CENTERS PERMISSIBLE.  
ACCEPTANCE CHECK TO BE MADE AT  
80°F. ± 1°.  
DO NOT ALTER THIS DESIGN WITHOUT  
FIRST CONSULTING TOOL DESIGN DEPT.  
APPLICABLE SPECIFICATIONS:  
MIL-STD-11000 - GAUGES, DIM. CONTROL

MARK THE FOLLOWING INFORMATION  
ON NEW OR DUPLICATE ITEMS AS SHOWN  
ON THE TOOL ORDER:  
PREFIX, GAUGE NUMBER, COPY NUMBER,  
OWNING AGENCY  
CUSTOMER GAUGE NUMBER,  
GAUGE DESIGN REVISION LETTER

100-22

REF. DESIGNATOR	REV.	REV.	NAME	FINISHED SIZE	DESCRIPTION	MATERIAL	HEAT TMT.	FIN.
TOLERANCES UNLESS NOTED			SCALE	DATE	Honeywell			
ONE P.L.C. DEC. ± 0.010			TL. 0.008	11-1-77	LAY OUT			
TWO P.L.C. DEC. ± 0.005			CHECKED		RELEASE MECHANISM FIX.			
THREE P.L.C. DEC. ± 0.002			APPROV.	3-15-78	ADAM			
FOUR P.L.C. DEC. ± 0.001			ENR.	10/1	SAAE 10A			
REF. 0000			D.C.		SHEET NO.	NO. SHEETS	ISS.	
					1	1		
					D H28009650-T1412			



**MACHINE #11**  
**RELEASE MECHANISM AND CASE ASSEMBLY**

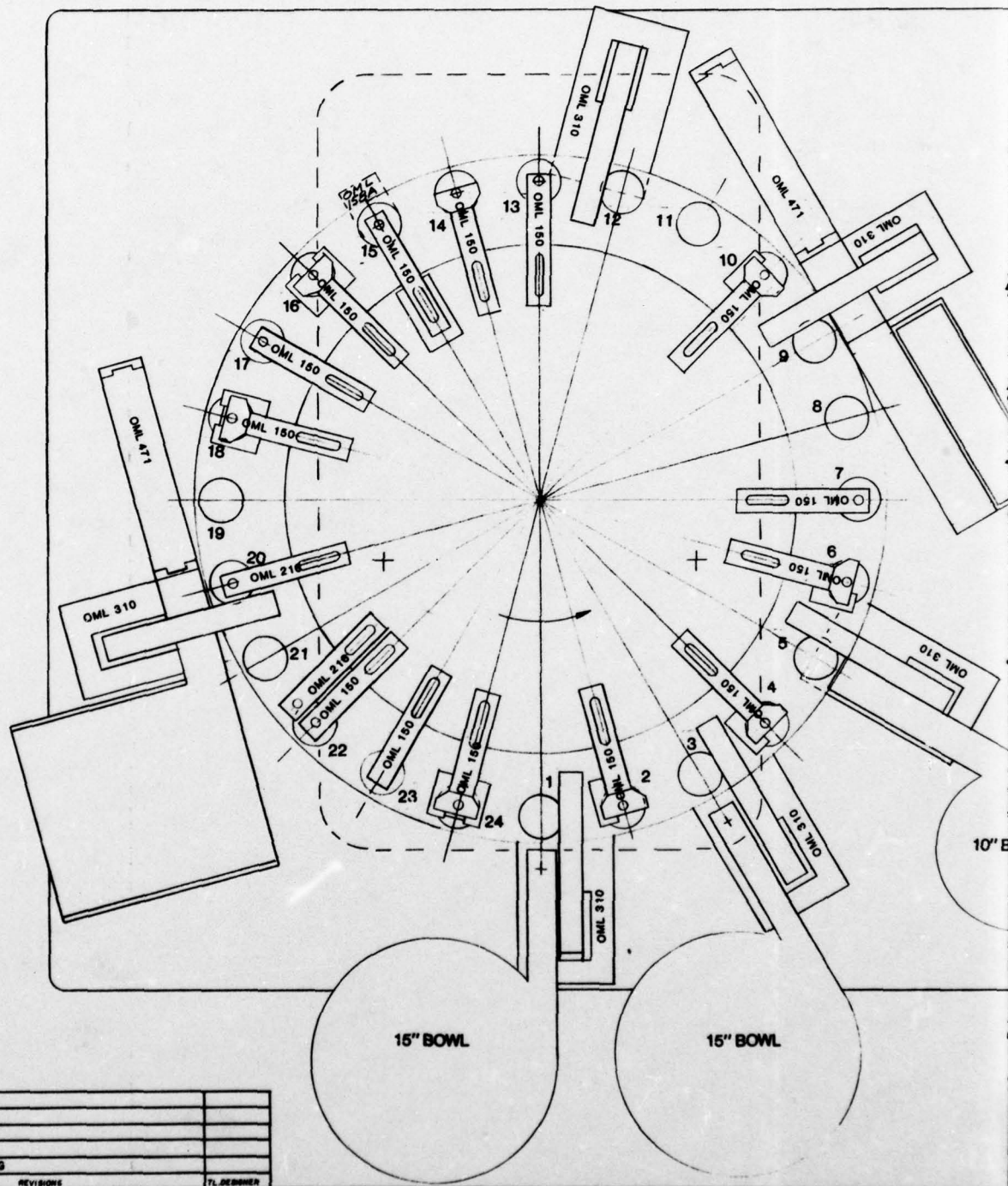
**APPLICABLE PIECE PARTS**

• Release Mechanism Assembly	Print #9292991
• Case Washer	Print #9292989
• Locking Ball	Print #9298581
• Sleeve	Print #9292988
• Sensor Case	Print #9292987

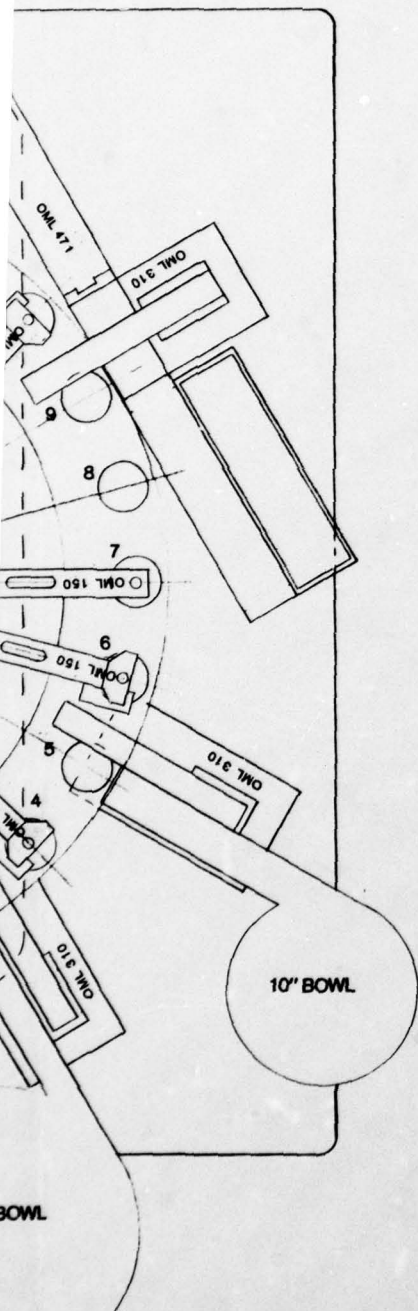
**MACHINE OPERATIONS**

This machine (Figure B-7) assembles the release mechanism assembly, case washer, locking balls, sleeve and sensor case. Crimping the sensor case onto the release mechanism along with the four locking balls maintains this assembly together. The small ball size adds to the complexity of this assembly. The sequence of operations is as follows:

1. The sleeve is fed from a vibratory bowl onto the nest.
2. The sensor case is fed from a vibratory bowl onto the nest.
3. The case washer is fed from a vibratory bowl and positioned in the nest adjacent to the sleeve and sensor case.
4. The nest is now compounded allowing the washer and sensor case to drop approximately 1/2 inch lower than the sleeve. This allows for positioning of the locking balls in the sleeve at a later station.

[illegible]

4 3 2 1  
DON'T FORGET SAFETY



OML 189 BASE

STA.	OPERATION	PART NO.
1	HOPPER ORIENT. METER & FEED SLEEVE	9292988
2	PROBE P & P	
3	HOPPER ORIENT. METER & FEED SENSOR CASE	9292987
4	PROBE P & P	
5	HOPPER FEED WASHER	9292989
6	PROBE P & P OF CASE & WASHER	
7	COMPOUND NEST	
8	IDLE	
9	MAG. FEED RELEASE MECH.	9292991
10	PROBE P & P	
11	IDLE	
12	VACUUM FEED (4) BALLS TO NEST RING	9298581
13	PROBE PRESENCE OF BALLS	
14	RADIALLY FEED BALLS	
15	RELEASE NEST COMPOUND. POSITION CASE WASHER	
16	PROBE PRECRIMP HEIGHT	
17	CRIMP SENSOR CASE (SUPPORT NEST)	
18	PROBE CRIMP	
19	IDLE	
20	EJECT TO MAGAZINE	9292986
21	IDLE	
22	REJECT TO CONTAINER	
23	EXERSIZE NEST ACTION, VACUUM NEST	
24	PROBE EMPTY NEST DEAD CAM NEST SUPPORTS	

REP. DESIGNATOR H28108567742	REV. REQ.	NAME	FINISHED SIZE	DESCRIPTION	MATERIAL	HEAT TRT.	FIN.
TOLERANCES UNLESS NOTED		SCALE 1/4	DATE 12-21-77	Honeywell			
MACH. 1:1 ±.0000		TL. DRGR. HOLM		TOP LAYOUT			
ONE P.L.C. DEC. ±.0000		DRAWN		SENSOR CASE & SLEEVE ASS'Y			
TWO P.L.C. DEC. ±.0000		CHECKED		XM74 GEMSS E.T.L. SAAFA 1			
THREE P.L.C. DEC. ±.0000		APPV. JHA	9-18-77	SHEET NO. 1			
FOUR P.L.C. DEC. ±.0000		ENG. QRS	9-19-77	77-9610-AA			
REF. 6000		D.C.					



5. The release mechanism is fed from a magazine onto the sleeve with the correct orientation.
6. Four 0.040 diameter locking balls are fed by a vacuum pickup to a nest ring. This nest ring is compounded allowing the balls to run down four ramps into their correct position in the assembly.
7. With the four balls in position the nest is compounded again and the washer and case are adjacent to the sleeve and release mechanism. The balls are now contained in the assembly.
8. The assembly is permanently fixed together by crimping the sensor case.
9. After the crimping operation, the sensor case and sleeve assembly is ejected into a magazine.

**MACHINE #12B**  
**SENSOR CASE AND SPRING ASSEMBLY**

**APPLICABLE PIECE PARTS**

- |                                     |                |
|-------------------------------------|----------------|
| • Release Mechanism & Case Assembly | Print #9292986 |
| • Eject Spring                      | Print #9298578 |
| • Booster Spring                    | Print #9292990 |
| • Eyelet, Interface                 | Print #9292981 |

**MACHINE OPERATIONS**

This machine (Figure B-8) is required to assemble the release mechanism and case assembly (obtained from Machine #11) with the ejection spring, booster spring and interface eyelet. The two springs are temporarily held in compression with a retaining pin. This retaining pin is removed on the following assembly machine. The sequence of machine operations is as follows:

1. The sensor case and sleeve assembly is fed from a magazine onto the nest. This assembly is inverted in the process so the release mechanism is located at the bottom of the nest.
2. Before the first spring is inserted, the nest pins are exercised to ensure correct position of the sensor case in the nest. These nest pins are later used to contain the booster spring in compression.

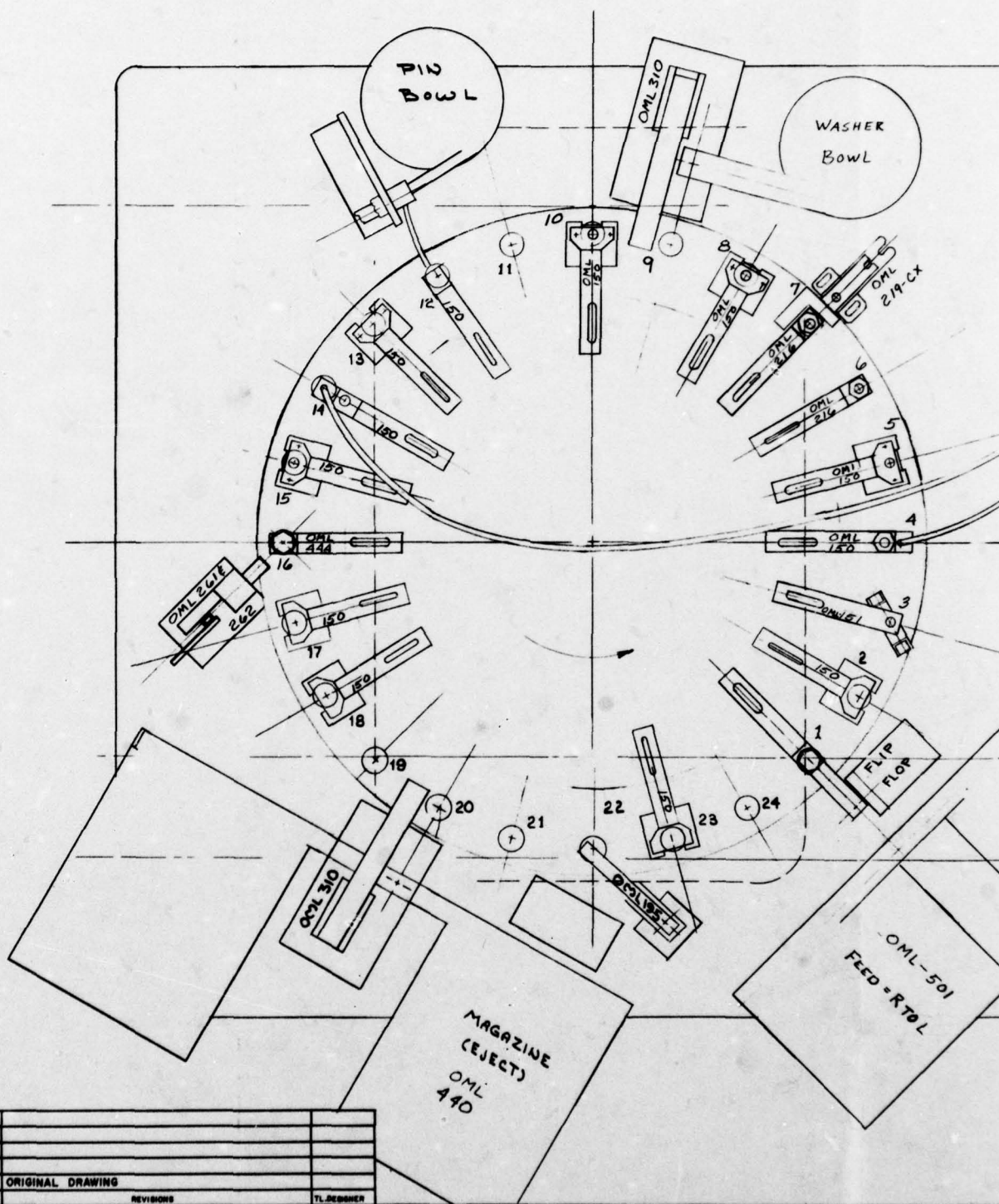


Figure B-8. Machine#12B Sensor Case and



3  
DON'T FORGET SAFETY

2

1

STA	OPERATION
1	MAG. FEED SENSOR CASE & SLEEVE ASSY.
2	PROBE PRESENCE , POSITION & FORM
3	EXERCISE NEST PINS
4	MAG. FEED BOOSTER SPRING
5	PROBE PRESENCE
6	PRE-POSITION SPRING
7	COMPRESS SPRING & CLOSE NEST
8	PROBE PRESENCE & POSITION
9	HOPPER FEED WASHER
10	PROBE PRESENCE
11	IDLE
12	HOPPER FEED RETAINING PIN
13	PROBE PRESENCE & POSITION
14	MAG. FEED EJECTION SPRING
15	PROBE PRESENCE
16	COMPRESS SPRING & INSERT RETAINING PIN
17	PROBE PRESENCE & POSITION OF PIN
18	PROBE PRESENCE OF SPRING
19	IDLE
20	EJECT TO MAGAZINE
21	IDLE
22	REJECT
23	PROBE EMPTY NEST
24	IDLE

WASHER  
Bowl

OML-440  
SPRING FEEDER  
(2) STACKED VERTICALLY  
FEED = L TO R

MAGAZINE  
FEEDER

OML-501  
FEED = RT0L

OML 189 BASE

REF. DESIGNATOR	DET.	REQ.	NAME	FINISHED SIZE	DESCRIPTION	MATERIAL	HEAT TRT.	FIL.
TOLERANCES UNLESS NOTED			SCALE	DATE	Honeywell			
MACH. 1:1 <sup>1</sup> FORMED 1:2 <sup>1</sup>			TL-DWR	W.G.C.	TOP VIEW LAYOUT			
ONE PLE. DEC. +2.000			DETAILS		SENSOR CASE & SPRING ASSY.			
TWO PLE. DEC. +2.000			CHECKED		XM74 GEMSS E.T.L. SAAF 128			
THREE PLE. DEC. +2.000			APV.	4-26-78	SHEET NO. 1			
FOUR PLE. DEC. +2.000			ENL.	4-21-78	NO. SHEETS 1			
REF. DWGS.			D.C.		77-0610-AF			

3. After the nest pins are exercised, the booster spring is inserted, prepositioned, compressed and retained in the case by the nest pins.
4. The interface eyelet is fed from a vibratory bowl and positioned into the sensor case.
5. Now, a temporary retaining pin is fed to a V slot located on the top of the nest. This pin is now adjacent to the release mechanism and case assembly.
6. Next, the ejection spring is fed to the nest and compressed. The nest slides are retracted and the retaining pin is pushed into place.
7. The sensor case and spring assembly is ejected to a magazine which will feed into Machine #12.

**MACHINE #12**  
**SENSOR CASE & BOBBIN ASSEMBLY**

**APPLICABLE PIECE PARTS**

- |                                 |                |
|---------------------------------|----------------|
| • Cap                           | Print #9298579 |
| • Ball Lock                     | Print #9298581 |
| • Anchor                        | Print #9292980 |
| • Bobbin Assembly               | Print #9292982 |
| • Sensor Case & Spring Assembly |                |

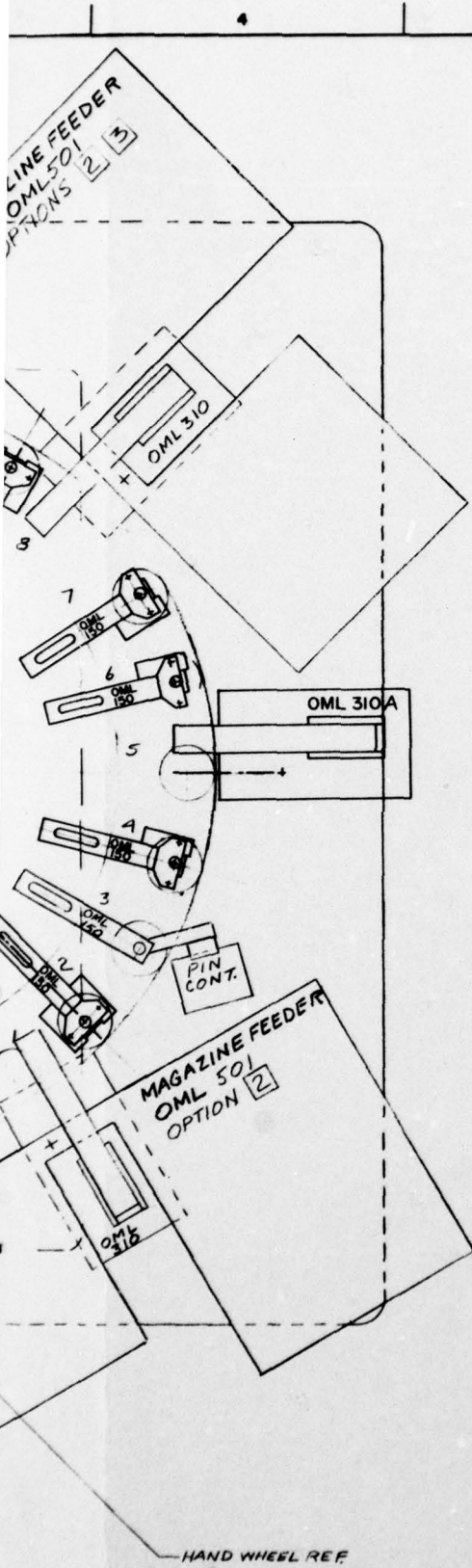
**MACHINE OPERATIONS**

This machine (Figure B-9) is required to assemble the cap, anchor, bobbin assembly and locking balls into the sensor case and spring assembly. The sensor case and spring assembly is acquired from machine #12B and already contains the eyelet and ejector and booster springs. Once these piece parts are assembled, they are contained by crimping the sleeve and cap tabs. The sequence of machine operations is as follows:

1. The sensor case and spring assembly is fed into the nest.
2. The pin containing the springs within the case is removed and nest slides are used to maintain the springs in compression.
3. Two locking balls are fed and positioned into the assembly.
4. The bobbin assembly is now fed and positioned into the case.







3  
DON'T FORGET SAFETY

STA NO.	
1	MAG FEED CASE
2	PROBE P & P
3	REM. PIN. CLOSE NEST SLIDES
4	PROBE PIN REMOVED, RAISE NEST POST
5	VAC. FEED (2) BALLS
6	PROBE PRES. LOWER & RAISE NEST POST
7	PROBE BALLS LOCATION
8	MAG FEED BOBBIN
9	PROBE P & P
10	IDLE
11	FEED ANCHOR
12	PROBE P & P
13	IDLE
14	BLANK & FEED CAP
15	PROBE P & P
16	CLINCH SLEEVE TABS
17	CLINCH CAP TABS
18	PROBE HEIGHT
19	DRILL BOBBIN
20	PROBE DRILLED BOBBIN
21	EJECT (INVERT) TO MAG
22	IDLE (VACUUM MANIFOLD)
23	REJECT TO CONTAINER
24	VAC. & PROBE EMPTY NEST

REV. DESIGNATOR	REV.	REV.	NAME	FINISHED SIZE	DESCRIPTION	MATERIAL	WEIGHT	TRY.	PH.
TOLERANCES UNLESS NOTED			SCALE 1/4"	DATE 10/12/78	Honeywell				
MACH 1.1" FRACTIONAL			DETAILER		TOP LAYOUT				
ONE PLE. DEC. +1.000			CHECKED		SENSOR CASE & BOBBIN ASS'Y				
TWO PLE. DEC. +1.000			APPV. JHA	4-28-78	XM74 GEMSS ETL				
THREE PLE. DEC. +1.000			ENG. JCS	4-21-78	SAAF# 12				
FOUR PLE. DEC. +1.000			REV.		77-9610-AB				
REF. DWG.			REV.						

5. The anchor is fed and positioned on top of the bobbin assembly.
6. Now the cap is blanked from strip stock and positioned on top of the anchor.
7. The sleeve tabs are crimped to hold the assembly together just after the nest slides are retracted.
8. The cap tabs are crimped to lock the assembly together.
9. Now the bobbin assembly within the case is drilled separating the weighted and sensor wire end of the bobbin assembly.
10. Finally this sensor case and bobbin is ejected from the machine into a magazine.



**MACHINE #13**  
**SENSOR AND BREAKWIRE ASSEMBLY**

**APPLICABLE PIECE PARTS**

- |                                   |                |
|-----------------------------------|----------------|
| • Sensor Case and Bobbin Assembly | Print #9292976 |
| • Wire, Mag., Elect               | Print #9298576 |
| • Ring, Retaining                 | Print #9298577 |
| • Cover Post                      | Print #9298575 |

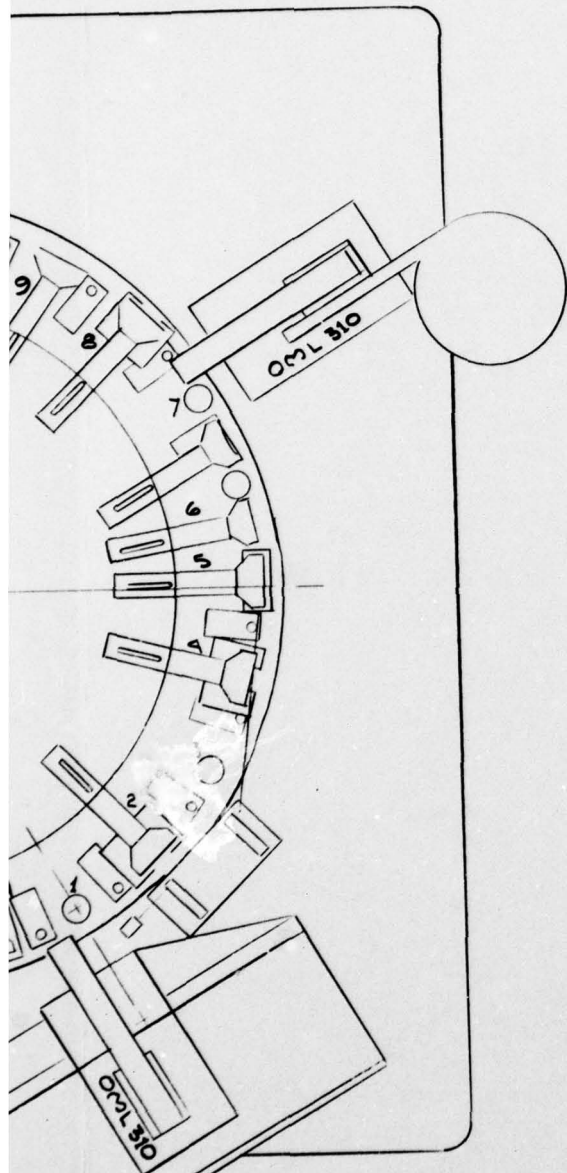
**MACHINE OPERATIONS**

This machine (Figure B-10) is designed to weld the breakwire to the two breakwire terminals. Several operations precede and follow the welding station. This is for preparatory purposes and continuity checks. The two other piece parts involved in this assembly machine are the retaining ring and plastic post cover. The sequence of machine operations is as follows:

1. First, the sensor case and bobbin assembly are fed from a magazine to the nest.
2. The breakwire is fed from a remote station and the insulation is burned off. Then this section of wire is positioned onto the breakwire terminals. To maintain control of the breakwire, jaws on the nest are activated.
3. A retaining ring and cover post are fed onto the release mechanism and positioned. This helps to further capture the breakwire.

Figure B-10. Machine #13 Sensor and Breakwire

4 3 2 1  
DON'T FORGET SAFETY



1	MAGAZINE FEED SENSOR CASE & BOBBIN ASSY	D
2	PROBE PRESENCE & POSITION	
3	POSITION FEED & BURNOFF INSULATION	
4	PROBE PRESENCE	
5	POSITION BREAKWIRE	
6	IDLE	
7	HOPPER ORIENT METER & FEED RETAINING RING	
8	PROBE PRESENCE & POSITION	
9	FINAL DRILL BOBBIN	
10	PROBE HOLE DEPTH	C
11	HOPPER ORIENT METER & FEED COVER POST	
12	PROBE PRESENCE & POSITION	
13	FLUX STATION	
14	WELD	
15	WELD	
16	CUT BREAKWIRE	
17	PROBE CONTINUITY	
18	TRIM WIRES	B
19	PROBE CONTINUITY	
20	EJECT TO STICK MAGAZINE	
21	IDLE	
22	AIR BLAST & VACUUM	
23	REJECT TO CONTAINER	
24	PROBE EMPTY NEST & AIR BLAST	A

REF. DESIGNATOR	DET.	REQ.	NAME	FINISHED SIZE	DESCRIPTION	MATERIAL	QTY	TOL.
TOLERANCES UNLESS NOTED				SCALE	DATE	Honeywell		
				1:1	4-28-78	NAME		
MACH. 21" FORGED STEEL				CHECKED	4-28-78	TOP VIEW LAYOUT		
ONE P.L.C. REQ. 1.000				DETAILER	4-28-78	SENSOR & BREAKWIRE ASSY		
TWO P.L.C. REQ. 2.000				INSP.	4-28-78	XM 74 QEM88 E.T.L. S.A.A.F. 13		
THREE P.L.C. REQ. 2.000				ENR.	4-28-78	DRAWING NO.		
FOUR P.L.C. REQ. 2.000				D.E.	4-28-78	77-9610-AC		
REF. 0000								

2



TS 578-4869

4. The bobbin assembly goes through a final drilling operation to clear any excess material in the assembly.
5. Now flux is applied and the breakwire is welded to each terminal.
6. The breakwire is cut and trimmed. A continuity probe is used to ensure a complete weld.
7. Finally the sensor and breakwire assembly is ejected to a magazine.

MACHINE #13A  
SENSOR EPOXY FIXTURE

APPLICABLE PIECE PARTS

- Tripline Sensor, Extend Range      Print #9292972

MACHINE OPERATIONS

This machine (Figure B-11) is required to apply epoxy to the tripline sensor following the application of two (2) No. 26 Gage Lead Wires (hand line operation). This epoxy serves to protect the electrical connections completed on Machine #13. The fixture utilizes the magazines containing the tripline sensors as its nest. The sequence of machine operations is as follows:

1. The magazines containing the tripline sensors are indexed through the fixture.
2. Five separate stations apply epoxy at selected locations onto the release mechanism which combine to form one continuous epoxy coating to end of sensor.
3. Following the application of epoxy the loaded magazines are cured in an oven for a period of time.

3.2.2 Physical Characteristics --

Production Area:

- a. The estimated weight of the Honeywell Standard Type 189 Base tooled is 9500 pounds. The space required for the typical

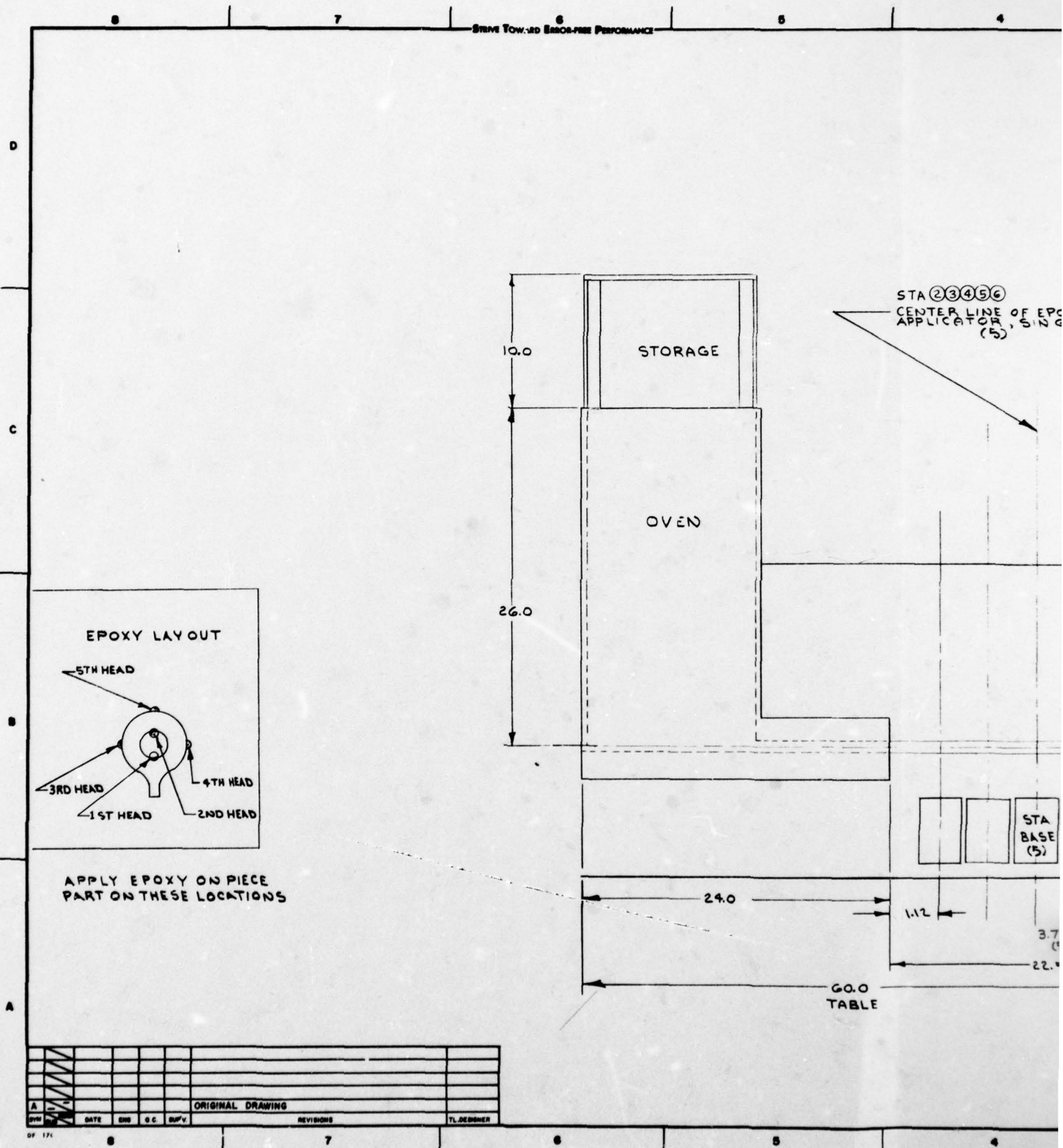


Figure B-11. Machine #13A Sensor Epoxy



DON'T FORGET SAFETY

STA 23456  
CENTER LINE OF EPOXY  
APPLICATOR, SINGLE HEAD  
(5)

STA ①  
CENTER LINE OF PROBE

HORIZONTAL  
MAGAZINE  
LOADER

#### FINISH

- ① CADMIUM PLATE
- ② BLACK OXIDE
- ③ CHROMIUM PLATE
- ④ ANODIZE (Specify Color) (GOLD)
- ⑤ ZINC PLATE

#### NOTE:

UNLESS OTHERWISE SPECIFIED:  
BREAK UNNECESSARY SHARP CORNERS  
APPROXIMATELY 1/32".  
SCREWS AND DOWELS TO SUIT.  
STAMP A.I.S.I. TYPE OR BRAND  
NAME ON ALL TOOL STEEL ITEMS.  
MAKE NO CHANGES IN DESIGN WITHOUT  
CONSULTING TOOL DESIGN DEPARTMENT.

MARK THE FOLLOWING INFORMATION  
ON NEW OR DUPLICATE ITEMS AS  
SHOWN ON THE TOOL ORDER:  
PREFIX, TOOL NUMBER, COPY NUMBER,  
OWNING AGENCY  
CUSTOMER TOOL NUMBER

101-52

G.O.O  
TABLE

REF. DESIGNATOR	REV. NO.	NAME	FINISHED SIZE	DESCRIPTION	MATERIAL	REV. TYP. PCL.
TOLERANCES UNLESS NOTED			SCALE	DATE	Honeywell	
MACH. 2:1 FORMED & C			TL. 0000	W.G.Q. 12-21-77	NAME	
ONE P.L. DEC. 1.040			DRAWN		TOP VIEW LAYOUT	
TWO P.L. DEC. 1.000			CHECKED		SENSOR ASSY. FIXTURE	
THREE P.L. DEC. 1.000			APPV.	JWA 5-11-78	XM 74 GEMSS E.T.L. SAAF 13A	
FOUR P.L. DEC. 1.0000			DRW.	GJB 4-21-78	SHEET NO. 17	
REF. DWG.			R.C.		77-9610-AD	

SAAF is 400 square feet of floor space (20 ft. x 20 ft. square). This area will support parts and assemblies. The need for aisles and cart storage space is not included in the 400 square feet for the SAAF.

- b. The space required for a typical spring winder (there are two on the E.T.L. Sensor), including stress relief ovens (two), would be 625 square feet of floor space. This area would support the carts used to load and store the completed springs, and to deliver the springs to SAAF 12B.
- c. The space required for an ultrasonic degreaser for cleaning of some parts and especially for cleaning SAAF #11 Nest would be 25 square feet (5 ft. x 5 ft. square).
- d. The space required for mixing epoxy for SAAF #13 would be 100 square feet (10 ft. x 10 ft. square). This area would require a special exhaust system to the outside of the building.
- e. The space required for sensor lot acceptance testing would be 600 square feet (60 ft. long x 10 ft. wide). This would have to be a clear area 20 ft. high at mid-point in the length of the test area.
- f. Allow 10% of total for aisles, break area and foreman's office.

**Production Area Required:**

a. 400 sq. ft. x 10 machines	4000 sq. ft.
b. Spring winding area	625 sq. ft.
c. Ultrasonic degreaser	25 sq. ft.

d. Epoxy mixing area	100 sq. ft.
e. Sensor LAT area	600 sq. ft.
Subtotal	5350 sq. ft.

f. 5350 x 10%

Grand total 5885 sq. ft.

Factory floor plan layouts for the area in Bldg. 103, Twin Cities Army Ammunition Plant, which has been assigned for assembly of the XM74 GEMSS Sensor are shown in Exhibit A. This requirement is shown in two phases: (1) The initial production facility incorporating only one each of the required machines and (2) the area required with an expansion to meet the PEP study requirement of 200,000 Sensors/month on a 1:8:5 shift basis.

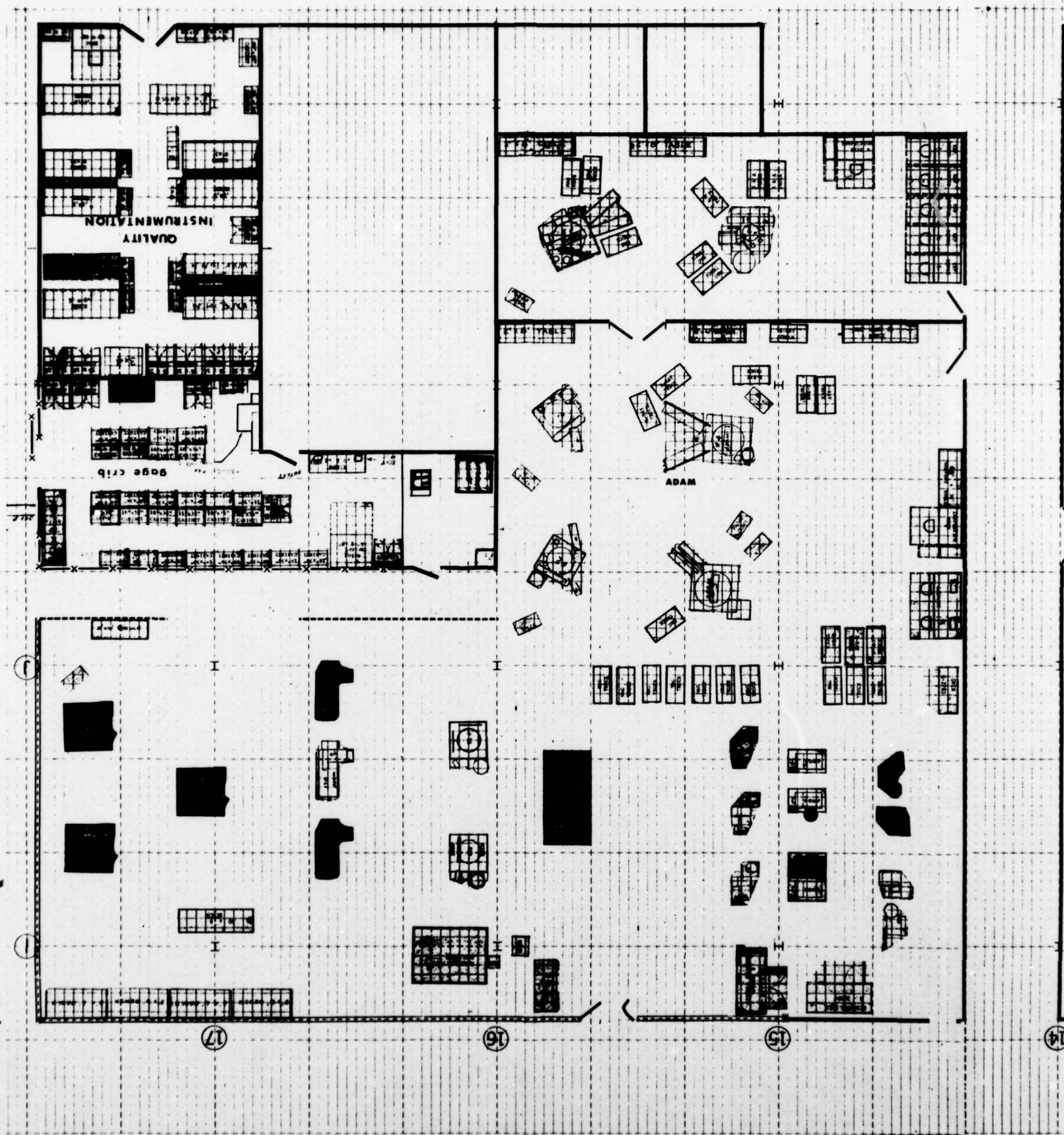
- g. There are no security requirements with the SAAF or the E.T.L. Sensor with regard to the manufacturing area.
- h. There are two areas where health considerations must be addressed, and are listed in Subparagraph "d" above. The out-gassing of the epoxy must be exhausted outside the work area (building) immediately. The epoxy is dispensed on Machine 13A so it must be equated with the exhaust to remove outgas from the epoxy.

The E.T.L. SAAF will meet all OSHA requirements prior to their installation on the E.T.L. production floor.

**3.2.3 Reliability** -- The reliability of each machine in the GEMSS XM74 Extended Sensor production line will have as a design goal a minimum mean-time-between failure (MTBF) of 8.5 hours. A failure is defined as any unscheduled machine stoppage during a scheduled production period which requires repair maintenance personnel action due to breakage or severe misalignment (i.e., malfunction) of any machine subassembly or component.













The time the machine runs is the production time associated with machine cycling, but not including any cycling during repairs or other cycling during which good product could not be produced. During the machine performance demonstrations and qualification testing, the number of failures and the production time will be collected as needed to calculate the machine mean-time-between-failure so that reliability growth can be analyzed.

3.2.4 Maintainability -- The maintainability of each machine used in the GEMSS Extended Tripline Sensor production line will have as a design goal a mean-time-to-repair (MTTR) of 1.75 hours. To achieve this goal, the machines will be designed to comply with the following:

a. Modularization -- The machines shall be designed to make maximum use of modularized subassemblies for ease of maintenance and to reduce the number and type of repair parts and assemblies required to support maintenance.

b. Accessibility -- Maximum use will be made of design techniques that will provide ready accessibility for replacement or servicing.

c. Interchangeability -- Components will be interchangeable without requirements to calibrate or adjust at time of replacement.

d. Maintenance Periods -- The system will be designed to that preventive maintenance is required only in off-production hours.

All machines are analyzed for MTTR using the definition of failure presented in Paragraph 3.2.3. The repair time will include preparation time, fault location time, fault correction time, adjustment/calibration time and check-out time. If two or more repair personnel are involved, only elapsed time

will be measured. However, the number, skills, and elapsed time for each repair person shall be recorded.

During the machine performance demonstrations and qualification testing, the number, type of failure, and the time to repair will be collected so that machine maintainability characteristics can be analyzed.

3.2.5 Availability -- The machines used in the GEMSS XM74 Extended Trip-line Sensor production line shall be designed to minimize the downtime (MTTR) and to maximize the uptime (MTBF) for each machine. The machines shall be designed to have a minimum inherent availability of 85 percent. The inherent availability is defined as:

$$\text{Availability} = \frac{\text{MTBF}}{\text{MTBF} + \text{MTTR}} \times 100\%$$

Where MTBF = Mean Time Between Failures

MTTR = Mean Time to Repair

### 3.2.6 Environmental Conditions

a. Storage -- The SAAF equipment metal surfaces shall be protected during storage with a high grade preservative oil/grease. It shall further be covered with a plastic sheet pulled tightly to reduce openings around the fixture. All moisture shall be removed from the compressed air system. The electrical power shall be disconnected by removal of the fuzes. The SAAF number shall be affixed to the outside surfaces of the plastic shield, together with date of storage and next inspection scheduled date. Storage temperature recommended 40°F to 90°F in low humidity area.

b. Shipment -- The SAAF shall be loaded on a special built shipping platform which will support the Base Machine and provide bracing for all

arms which may extend beyond the SAAF table. The Electrical Control Box shall be braced as required to prevent vibration in shipping. The Upper Tooling Plate shall be in the down position during all movement of the SAAF's. Long magazine feeding stations (feed in or feed out) may be secured to the shipping platform. The SAAF shall further be protected by a wood frame plastic covered "Green House" construction to protect from water and dirt during shipment. The SAAF shall be protected from temperatures below 40°F. The mode of transportation recommended is the "Air Ride" offered by National Moving Concerns. Excessive shock and vibrations may result in damaged electrical systems as well as out of line conditions on metal parts.

c. Operation -- The SAAF equipment must be kept clean and well lubricated during normal operations. The build up of moisture in the Air System must be periodically drained. All cam followers and oil reservoirs must be serviced on a regular monthly schedule.

d. Environment -- The E.T.L. SAAF's have no effect on the natural environment except for SAAF 13A which dispenses epoxy onto the sensor. The epoxy outgassing is removed from the SAAF area by a special exhaust system. The mixing of the outgassing with the outside air has no effect on the environment.

3.2.7 Transportability -- Transporting assembly machines must be conducted with the care normally exercised in moving machinery. Moving must be by experienced personnel using proper equipment and safety precautions. There are no abnormal requirements for transporting.

### 3.3 Design and Construction

Contractor's background in assembly machine production must show a consistently high accomplishment on machine efficiency and reliability. All acceptable



industrial standards followed to maintain the quality in the design and construction of each machine. Military standards are to be followed when applicable.

3.3.1 Material, Process and Parts -- The materials used in standardized components are materials which lend themselves to the particular application. All castings are stress relieved to obtain maximum life and stability. All critical areas on machine bases and tooling are either painted, anodized or plated to provide a protective finish and add aesthetic value to the machine. Standardized components are selected on the basis of their particular application. The components to be of a correct and proportioned size so as not to approach their practical fatigue life. All applicable industrial standards are to be followed.

3.3.2 Tooling -- All special tooling is designed to meet the specific requirements listed in paragraphs 3.2.3, 3.2.4 and 3.2.5 and acceptable industrial standards.

3.3.3 Name Plates and Marking -- There is no general criteria used for identification and markings on electrical and pneumatic lines and controls. Identification here is only used when necessary. Lines are identified whenever hydraulic or water lines are concerned to avoid confusion in the machine system.

3.3.4 Interchangeability -- Standardized tooling and castings should be utilized in the design and assembly of assembly machines. Some of the standardized features include: probes, parts feeding mechanisms, magazine feeders, work stations, etc. This major emphasis on standardization throughout reduces the number and type of repairs required to support maintenance and maintain interchangeability at a maximum.

3.3.5 Safety -- Through the use of good design and the proper selection of material, machine safety shall be in conformance with the intent, and general interpretation of the following safety publications and codes:

- a. AMCR385-100, AMC Safety Manual
- b. AR40-5, Health and Environment
- c. MIL-STD-882A, System Safety Program Requirements
- d. Occupational Safety and Health Act (OSHA)
- e. National Electric Code for Hazardous Operations.

The Safety Engineering effort will be coordinated with RAM and HFE efforts to establish an integrated program. A safety review of each machine shall be conducted to determine compliance with the above-listed publications.

3.3.6 Human Performance/Human Engineering -- Each machine in the GEMSS XM74 Extended Tripline Sensor production line shall be in conformance with the intent and general interpretation of the following documents:

- a. MIL-STD-1472B -- Human Engineering Design Criteria for Military Systems Equipment and Facilities.
- b. MIL-HDBK-759 -- Military Standardization Handbook, Human Factors Engineering Design for Army Material.

A human factors review of each machine shall be conducted prior to release of the machine for use in the normal production environment to determine compliance with the above-listed documents.

3.3.7 Standards of Manufacture -- The standards of manufacture for Special Automatic Assembly Machines used in the assembly of the XM74 GEMSS

Extended Tripline Sensor will be in accordance with the items described on the following specification sheets for automatic assembly machines (Ref: Exhibit B). These are general specifications. Detailed specifications will be generated prior to actual machine build and will be recorded on a form as shown in Exhibit C.

### 3.4 Major Component Characteristics

Each component listed in 3.1.1 is self explanatory for its performance and physical characteristics. Your attention is invited to the attached booklet entitled, Automation Standard Equipment, for additional information.

## 4.0 QUALITY ASSURANCE PROVISIONS

The performance and physical requirements of the GEMSS Extended Tripline Sensor Special Automatic Assembly Line shall be demonstrated on each Special Automatic Assembly Fixture (SAAF).

### 4.1 General

Development Contractor personnel shall witness the tests/verifications of each SAAF in the Automated Assembly Laboratory. SAAF completeness and physical requirements shall be verified prior to the demonstration test. SAAF performance shall be verified during the demonstration test. The conformance of the Extended Tripline Sensor assemblies to QAP-GEMSS-1, shall be verified during and following each demonstration test.



EXHIBIT B

HONEYWELL	SPECIFICATION SHEET AUTOMATIC ASSEMBLY MACHINE PRELIMINARY	REVISION	ORIGINAL
		DATE	31 AUGUST 1978
		NAME	R. FAIRCHILD
		CC: M.L.	J. BUDNICKI
MACHINE:	XM74 GEMSS ERT SENSOR	M.L.	P. VAN BROCKLIN
PART OP:	GENERAL SPECIFICATION ALL MACHINES	QUAL	C. FLAMMING/R. KEEFE
TOOL NO:		P.E.	G. ADAMS
ACCEPTANCE:	1 HR		
ITEM			

1. Machine Counters

- a. Cycle counter - 7 digit total cycle quantity, non-resettable.
- b. Good assemblies - 6 digit, resettable.
- c. 6 digit total cycle resettable.
- d. Mount counter as low as possible on cabinet.

2. Identification Areas

- a. All nest locations with stamped number.
- b. Air Gauges and valve marked with recommended settings.
- c. Limit switches noted Normally Open (NO) or Normally Closed (NC).
- d. Station numbers permanently affixed.

3. Machine Controls

- a. One remote jog card - female receptable.
- b. Index start/stop button on each end of each machine within easy reach of operator unless noted otherwise.
- c. All presence and position probes stop up unless noted.
- d. All orient motors stop with cycle stop.
- e. Probes with count feature will be adjustable - one, three, or stop up.
- f. All feeder bowls and vibratory tracks to stop with main motor start/stop switch.
- g. Individual air pressure regulators for each air operated work station when required for critical adjustments.
- h. Lighted numbers to be displayed for probe stations, jam switches, etc.
- i. Explanation of all lighted numbers to be displayed on two sides of electrical cabinet.
- j. All open air blow lines to stop with cycle start/stop switch.

4. Safety

- a. The machine must meet Honeywell requirements for safety.
- b. Must meet OSHA requirement for pinch points and notes.
- c. Guards around explosive handling stations shall fit contour of station and have door to allow easy accessibility.
- d. Noise-muffle all air valves and Venturi vacuums, avoid air assist on eject.

5. Tool Inspection - Machine Check List

- a. Proper assemble, alignment and security of machine components.
- b. Dimensional and positional variations of nests and other selected components.
- c. Proper function of machine components.
- d. All cams to be keyed.
- e. Critical nest dimensions to be identified by Production Engineer must be inspected and verified by Tool Inspection prior to assembly.

EXHIBIT B

HONEYWELL	SPECIFICATION SHEET AUTOMATIC ASSEMBLY MACHINE	REVISION	ORIGINAL
		DATE	31 AUGUST 1978
	PRELIMINARY	NAME	R. FAIRCHILD
MACHINE:	XM74 GEMSS ERT SENSOR	CC:	M.L. J. BUDNICKI
PART OP:	GENERAL SPECIFICATIONS ALL MACHINES		M.L. P. VAN BROCKLIN
TOOL NO:		QUAL	R. KEEFE
ACCEPTANCE:	1 HR	P.E.	G. ADAMS
ITEM:			

6. General Acceptance

- Complete design package of each machine to include electrical schematic, and program documentation. All cam layouts and charts.
- Machines will be provided with a tool cabinet with setup masters, cam templates and all special tools.
- A critical spare parts list will be included with each machine, listing parts that should be kept in stock to preclude production delays. Drawing(s) shall be included if needed to describe the part(s).
- An Operation and Maintenance Manual shall be supplied with each machine.

7. Acceptance Demonstrations

- Machine Base - Each machine base will be run-in for a period of 24 hours minimum at full cycle to demonstrate base reliability prior to machine assembly.
- Tooled Machine Without Parts - Each completely tooled machine will be run less parts at full cycle rate for a period of 24 hours minimum to demonstrate proper machine fabrication.
- Completed Machine With Parts - Each machine will be accepted for production of sensor qualification build after demonstrating a minimum of 30 percent of estimated production rate (i.e., a quantity of "acceptable units" equal to one hour estimated production rate must be fabricated in 3.33 hours (maximum).
- Machine caused defects shall be no greater than XX% of the acceptable units. Acceptable units include "non-machine caused" defective units. The machine base, feed bowls, and other components will run at the specified cycle rate during this acceptance. The operator loading during this run will be the same as the anticipated operator loading for the initial product build. An operator can be Machine Lab Technician or Tool Maker.

8. General Tool Station Features

- Eject stations to magazines shall have parts jam stop switches and have fail safe concepts.
- Magazine stackers shall have a positive method of magazine alignment.
- Inspection stations and probes will have setup lights.
- Presence and position probes will normally follow each work and assembly station according to standard practice, unless otherwise noted.
- Avoid the use of air for eject, reject, and feeding of parts whenever practical.
- Feeder bowls to have slots or holes for escape of dirt, foreign material, short parts, etc.

EXHIBIT A

HONEYWELL	SPECIFICATION SHEET AUTOMATIC ASSEMBLY MACHINE PRELIMINARY	REVISION	ORIGINAL
		DATE	31 AUGUST 1978
		NAME	R. FAIRCHILD
MACHINE:	XM74 GEMSS ERT SENSOR	CC: M.L.	J. BUDNICKI
PART OP:	GENERAL SPECIFICATION ALL MACHINES	M.L.	P. VAN BROCKLIN
TOOL NO:	1 HR	QUAL	R. KEEFE
ACCEPTANCE:		P.E.	G. ADAMS
ITEM			

9. Extra Machine Base Features

Each machine shall have a variable speed motor.

10. Approvals Required

- a. Production Engineering approval is required on machine concepts, dial layouts, nests designs, inspection station designs and magazine designs and acceptance demonstrations.
- b. Production Engineering approvals are required on preliminary inspection station concepts and final inspection station designs before starting fabrication.
- c. Quality Engineering approvals on qualified inspection stations.

11. Magazine Requirements

Each machine package shall include a sufficient quantity of magazines to hold 2,000 parts or assemblies.



## TS 578-4869

[illegible]

## Exhibit C

3.3.8 Facilities

3.3.8 Facilities

FACILITY		FLOOR LOAD	CLEAR CEILING HT.	POWER	AIR	DRAIN	EXHAUST SYSTEM	FLOOR SPACE
Ultra Sonic Degreaser	400#	10 FT	460V AC	NO	YES	YES		8' x 8' SQ
Stress Relieve Oven	2000#	10 FT	440V AC	NO	NO	YES		10' x 12' SQ
Air Conditioner Equipment (Ceiling Mount)	N/A	20 FT	440V AC	NO	YES	NO		N/A
Humidity Chamber	100#	8 FT	110V AC	NO	NO	NO		8' x 8' SQ
Demagnetizer	500#	8 FT	460V AC	NO	NO	NO		5' x 5' SQ
Epoxy out-gassing Exhaust Equipment	N/A	8 FT	110V AC	YES	NO	YES		N/A

## Exhibit C

TS 578-4869

3.3.9

Page 1 of 2

UTILITY CONSUMPTION

SAAF #	ELECT POWER	COMPRESSED AIR	FUEL OIL	GAS	WATER	STEAM
7	2.29	950	-	-	-	-
8	2.29	950	-	-	-	-
9	2.29	950	-	-	-	-
10	2.44	950	-	-	-	-
10A	2.44	950	-	-	-	-
11	2.51	950	-	-	-	-
12	2.67	950	-	-	-	-
12A	.42	950	-	-	-	-
12B	2.67	950	-	-	-	-
12C	.42	950	-	-	-	-
13	13.72	950	-	-	-	-
13A	2.44	950	-	-	-	-
TOTAL	36.60	11,400	-	-	-	-
UNIT	KWH/M	CFH/M				

The data listed above is from a 1977 Utility Useage Analysis, and comparable machines used on the ADAM Program at Honeywell, Twin City Arsenal, Building 103 New Brighton, Minnesota.



## Exhibit C

Page 2 of 2

3.3.9

## UTILITY CONSUMPTION/OPERATIONAL RATE

ELECTRICAL POWER KWH/HR					COMPRESSED AIR CF/HR				
SAAF #	KWH/HR	160 HR/MO	320 HR/MO	500 HR/MO	SAAF #	CF/HR	160 HR/MO	320 Hr/MO	500 HR/MO
7	2.29	366.4	732.8	1145.0	7	950	152,000	304,000	475,000
8	2.29	366.4	732.8	1145.0	8	950	152,000	304,000	475,000
9	2.29	366.4	732.8	1145.0	9	950	152,000	304,000	475,000
10	2.44	390.4	780.8	1220.0	10	950	152,000	304,000	475,000
10A	2.44	390.4	780.8	1220.0	10A	950	152,000	304,000	475,000
11	2.51	401.6	803.2	1255.0	11	950	152,000	304,000	475,000
12	2.67	427.2	854.4	1335.0	12	950	152,000	304,000	475,000
12A	.42	67.2	214.4	210.0	12A	950	152,000	304,000	475,000
12B	2.67	427.2	854.4	1335.0	12B	950	152,000	304,000	475,000
12C	.42	67.2	214.4	210.0	12C	950	152,000	304,000	475,000
13	13.72	2195.2	4390.4	6860.0	13	950	152,000	304,000	475,000
13A	2.44	390.4	780.8	1220.0	13A	950	152,000	304,000	475,000
TOTAL	36.60	5856.0	11,712.0	18,300.0	TOTAL	11,400	1,824,000	3,848,000	5,700,000
	-	KWH/MO	KWH/MO	KWH/MO		-	CF/MO	CF/MO	CF/MO

#### 4.2 Quality Conformance

a. Prior to the demonstration test, the following shall be verified:

- Identification area
- Proper function of feed, work and probe stations
- Control system must stop the SAAF and provide a numerical display to indicate the cause of the stoppage.
- Counters presence and function
- Safety
- Workmanship

This verification will be part of the physical and special criteria conformance.

b. During the conduct of the demonstration test, the following shall be recorded:

- Date of demonstration test
- Start time
- Stop time (Stops will be recorded by number displayed and/or reason for stop and duration of stop)
- Observed time
- Total run time
- Total down time
- Cycle rate (CPM)
- Total cycle count
- Total parts count
- Lot number of parts and material assembled

- Cause of failure
- Corrective action
- Time to determine corrective action
- Time to perform corrective action
- Total down time per failure

This data will be part of the performance and reliability analysis.

c. During and/or following the demonstration test, samples of SAAF-accepted assemblies shall be selected and inspected and/or tested for conformance to QAP-GEMSS-1. The inspection/test results will be part of the Quality evaluation.

## 5.0 PREPARATION FOR DELIVERY

Each assembly machine built is to be debugged before delivery and is to be considered a completed unit. Each machine is to be appropriately crated or packaged so that damage will not be incurred when moved from one facility to another.



APPENDIX C

INSPECTION PROCEDURES

FOR

THE EXTENDED TRIPLINE SENSOR ASSEMBLY

Specification: QAP-GEMSS-1 dtd 23 June 1977

APPENDIX C

INDEX - INSPECTION PROCEDURES  
EXTENDED TRIPLINE SENSOR ASSEMBLY

<u>DRAWING NO.</u>	<u>REVISION NO.</u>	<u>NOMENCLATURE</u>	<u>INSPECTION PROCEDURE ISSUE NO.</u>
9292972	B	Tripline Sensor, Extended Range (2 Parts: Pre-seal and Post-Seal)	2
9292976	B	Sensor Case & Bobbin Assembly	2
9292980	B	Anchor	2
9292981	B	Eyelet, Interface	2
9292982	B	Bobbin Assembly	2
9292983	B	Weight, Bobbin	2
9292985	B	Bobbin	2
9292986	B	Release Mechanism & Case Assembly	2
9292987	B	Case, Sensor	2
9292988	B	Sleeve	2
9292989	B	Washer, Case	2
9292990	B	Spring, Booster	2
9292991	B	Release Mechanism Assembly	2
9292998	B	Diaphragm Assembly	2
9298575	Orig.	Cover, Post	2
9298576	Orig.	Wire, Magnet, Electrical	2
9298577	Orig.	Ring, Retaining, Breakwire	2
9298578	Orig.	Spring, Ejection	2
9298579	Orig.	Cap	2
9298581	Orig.	Ball, Locking	2
9298582	Orig.	Spring, Washer	2
9298586	Orig.	Ring, Ball Lock	2
9298587	Orig.	Housing Assembly	2
9298588	Orig.	Cup, Housing	2
9298589	Orig.	Tube, Housing	2

APPENDIX C

- 2 -

<u>DRAWING NO.</u>	<u>REVISION NO.</u>	<u>NOMENCLATURE</u>	<u>INSPECTION PROCEDURE ISSUE NO.</u>
9298591	Orig.	Tape	2
9298592	Orig.	Thread, Polyester	2
9298597	Orig.	Plate, Diaphragm	2
9298598	Orig.	Diaphragm	2
9298599	Orig.	Eyelet, Diaphragm	2
9298601	Orig.	Terminal, Breakwire	2
9298617	Orig.	Urethane Compound	2
9298618	Orig.	Gasket, Fibrous	2
9298619	Orig.	Paper, Gasket	2



# INSPECTION PROCEDURE

PART NAME SENSOR ASSY. (PRE SEAL)	PROGRAM EXT. TRIP LINE	IP ISSUE 2	SAMPLING PLAN REVISION	DATE 7 AUG 78
PART NUMBER 9292972	PART FAMILY	OPERATION NUMBER 0135	REL/MOD B	TALLY NUMBER
Q&ID	INSPECTION DEPARTMENT	APPROVALS	SHEET 1	OF 1

ITEM	CHARACTERISTICS	RE	CD	S WK	S C	SP	L	AQL	N	C	EQUIPMENT
1	A VISUAL - PRE SEAL 1 PARTS MUST BE OF GOOD WORKMANSHIP AND MUST BE FREE OF DAMAGE	4.2.4 E-201 4.2.5 E-202						65			VISUAL
2	CHECK FOR PRESENCE OF STRAIGHT BREAKWIRE, STAKED AND SOLDERED IN TWO PLACES. SOLDER JOINT MUST BE SMOOTH AND SMOOTH. BREAKWIRE MUST NOT EXTEND OUTSIDE THE O.D. OF THE ASSEMBLY.	4.2.4 E101						40			3 POWER MAGNIFIER
3	CHECK THAT THE BREAKWIRE RETAINING RING IS PROPERLY ENGAGED OVER THE GOBBIN POST	4.2.4 E102						40			3 POWER MAGNIFIER

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# INSPECTION PROCEDURE

PART NAME SENSOR ASSY (POST SEAL)	PROGRAM EXT. TRIP UNIT	IP ISSUE 2	SAMPLING PLAN REVISION	DATE 17 AUG 78
PART NUMBER 9292972	PART FAMILY	OPERATION NUMBER 0295	REV/MOD B	TALLY NUMBER
QCID M33	INSPECTION DEPARTMENT	APPROVALS	SHEET 1	OF 3

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	AQL	N	C	EQUIPMENT
A. VISUAL (POST SEAL).											
1	CHECK THAT SEAL ON BACK IS PRESENT COMPLETE AND NOT DAMAGED	4.2.5	ER01			9		.65			3 POWER MAGNIFIER AND LIGHT
2	CHECK THAT SEAL MATERIAL DOES NOT EXTEND BEYOND LIP OF DIAPHRAGM.										
B. PHYSICAL AND DIMENSIONAL											
1. 536 MAX O.D.			A2			9		.40			GAGE
C. FUNCTIONAL											
1. CHECK THAT CONTINUITY FROM BREAKWIRE TERMINAL LEAD TO OTHER BREAKWIRE TERMINAL LEAD IS NOT GREATER THAN 10 OHMS			A1			SP			3150		DIGITAL MULTIMETER

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# INSPECTION PROCEDURE

PART NAME SLIPCO A-501 (POST SEAL)	PROGRAM EXP. TRIP LINE	ISSUE 2	SAMPLING PLAN REVISION	DATE 15 AUG 78
PART NUMBER 9208072	PART FAMILY	OPERATION NUMBER 0295	REV/MOD B	TALLY NUMBER
QID M 33	INSPECTION DEPARTMENT	APPROVALS	SHEET 2 OF 3	

ITEM	CHARACTERISTICS	RE	CD	SS WK	SC R	SP	L	LAQL	N	C	EQUIPMENT
2	50 UNITS SHOULD BE SELECTED FROM THE PRECEEDING INSPECTION (CI) , TESTED ACCORDING TO THE CONDITIONS LISTED IN NOTE 3, AND EVALUATED ACCORDING TO THE FOLLOWING:  .1 NON RELEASE ( SPECIFIC PROCEDURE FOR PERFORMING THIS TEST IS PENDING AND CONFORMANCE IS WAIVED UNTIL A RESOLUTION IS REACHED) .2 RELEASE .3 TRIPLINE DEPLOYMENT .4 BREAKWIRE FUNCTION IN .5 POST RETENTION AND THREAD BREAKING STRENGTH  A* FOR INFORMATION ONLY - NO REJECTION	4.2.6 " " "							50 1 50 3 50 3 A*		H28009650-E1 BARRIER UG-143L 1000 30' TAPE MEASURE 12811226654

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REV. 0  
PART NO. 9292972

HI-20 REV 9/71

# INSPECTION INSTRUCTION

SOURCE


DIST.

ITEM NO.	INSTRUCTIONS
C2	<p>1 MOUNT AN EXPENDED UNIT IN TEST FIXTURE (H28009656-E1)</p> <p>2 PLACE A BARRIER AT A DISTANCE OF 32 FEET FROM THE FRONT OF THE SENSOR AND AT A HEIGHT OF SEVEN INCHES ABOVE THE TOP OF THE SENSOR</p> <p>3 ADJUST THE AIRPRESSURE REGULATOR TO 300 +25 PSIG.</p> <p>4 SET TIMER FOR 200 MILLISECONDS MAXIMUM</p> <p>5 REMOVE EXPENDED UNIT FROM TEST FIXTURE</p>
C2.2	<p>1 PLACE TEST UNIT IN H28009656-E1 TESTER AND CONNECT H28112266E4 TESTER TO TEST UNIT LEADS (POLARITY IS INSIGNIFICANT)</p> <p>2 READ CONTINUITY ON METER</p> <p>3 PRESS BUTTON TO APPLY AIRPRESSURE TO RELEASE MECHANISM</p> <p>4 RELEASE: THE EJECTION SPRING, SLEEVE, CAP, AND BOBBIN MUST BE RELEASED FROM THE CASE. IF ONE UNIT FAILS, REPLACE UNIT SO THAT 50 (FIFTY) UNITS REMAIN FOR NEXT ITEM OF TEST. IF TWO UNITS FAIL "RELEASE TEST", LOT IS REJECTED</p>
3	<p>1 TRIPLINE DEPLOYMENT: THE TRIPLINE MUST CLEAR THE BARRIER. IF 3 OR LESS UNITS FAIL, REPLACE UNITS IN SAMPLE SO THAT 50 (FIFTY) UNITS REMAIN FOR NEXT ITEM OF TEST. IF FOUR UNITS FAIL, REJECT LOT.</p>
4	<p>1 BREAKWIRE FUNCTION: ATTACH FORCE GAGE TO THE THREAD AT A DISTANCE OF APPROXIMATELY 32 FEET FROM CASE AND APPLY FORCE OF .3 TO .9 POUNDS TO BREAK "BREAK WIRE". BREAKING OF WIRE WILL BE INDICATED BY ILLUMINATION OF GREEN LIGHT ON TEST FIXTURE. THE FORCE APPLICATION SHOULD BE STEADY AND ALONG THE AXIS OF THE 30SPIN POST. THE BOBBIN POST SHOULD REMAIN IN THE CASE. IF 3 OR LESS UNITS FAIL, REPLACE UNITS IN SAMPLE SO THAT 50 UNITS REMAIN FOR NEXT ITEM OF TEST. IF 4 UNITS FAIL, REJECT LOT.</p>
5	<p>1 POST RETENTION AND THREAD BREAKING STRENGTH: AFTER 32.4.1 IS ACCOMPLISHED, CONTINUE THE PULL FORCE UNTIL THE THREAD BREAKS OR UNTIL A PULL FORCE OF 1.2 POUNDS IS REACHED. THE POST SHOULD REMAIN IN PLACE IN THE CASE. (INFORMATIONAL)</p>
<p>NOTE: NO MORE THAN ONE OF THE 2 ALLOWABLE DEFECTIVES SHALL EXCEED 1.2 POUNDS.</p>	
<p>NAME: SENSOR ASSEMBLY (Post SEAL)</p>	
<p>INSP. OP. 0298 DEVICE NO. XM 74 INSPECTING DEPT. 6021</p>	
<p>ISSUE NO. 2 DATE 16 AUG 78 PAGE 3 OF 3 WRITER Tamar APP. 182</p>	

INSP. OP. REV. 0 PART NO. 9292972

# INSPECTION PROCEDURE

PROGRAM EXT TRIP LINE		IP ISSUE 2	SAMPLING PLAN REVISION		DATE 7 AUG 78
PART FAMILY		OPERATION NUMBER 0395	REV/MOD B	TALLY NUMBER	
INSPECTION DEPARTMENT		APPROVALS		SHEET 1 OF 2	

ITEM	CHARACTERISTICS	RE	CD	S W	S K	C R	SP	L	AQL	N	C	EQUIPMENT
A 1011	<p>XRAY TO BE TAKEN WITH ASSEMBLY POSITIONED AS SHOWN IN SKETCH "A". XRAY TECHNICIAN WILL INTERPRET PRINTS FOR IDENTIFICATION OF LOCKING CALLS AND NOTIFY INSPECTION FOR VERIFICATION</p>  <p>SKETCH "A" SENSOR CASE AND BOBBIN ASSEMBLY</p>									50	1	XRAY
B	VISUAL AND DIMENSIONAL											
1	CHECK THAT THE FOUR CAP TABS OVERLAY THE FOUR CRIMPED SLEEVE TABS (SMALL GAP UNDER CRIMPS IS PERMISSIBLE.)	4.2.5	101				9		40			VISUAL
2	CHECK FOR PRESENCE OF ANCHOR-VISIBLE THROUGH HOLE IN CAP END OF THE ASSEMBLY.											

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PART NAME CENT. R CASE AND BOBELL ASSY	PROGRAM EXT. TPIE 1000	ISAMPLING PLAN REVISION	DATE 15 AUG 78
PART NUMBER 9292976	PART FAMILY EXT. TPIE 1000	OPERATION NUMBER 2	TALLY NUMBER
INSPECTION DEPARTMENT M 93	APPROVALS	REV/MOD B	SHEET 2 OF 2

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# INSPECTION PROCEDURE

PART NAME ANCHOR		PROGRAM EXT TRIP LINE		IP ISSUE 2		SAMPLING PLAN REVISION		DATE 7 AUG 78	
PART NUMBER 9298980		PART FAMILY		OPERATION NUMBER		REV/MOD B		TALLY NUMBER	
QCID M33		INSPECTION DEPARTMENT		APPROVALS		SHEET		OF 1	

ITEM	CHARACTERISTICS	RE	CD	S WK	S C	R	SP	L	AQL	N	C	EQUIPMENT
A	DIE/MOLD CHECK AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL QE									2	0	MSI
B	VISUAL											ROCKWELL TESTER
1	CHECK HARDNESS AND TEMPER TO R15N 80 TO 83 (MIL-S-46049)		A2							5	0	
2	PARTS TO BE FREE OF DAMAGE AND MUST BE OF GOOD WORKMANSHIP		C				9		1.0			VISUAL
3	PROPER DIE RADIUS, BURRS PERMITTED PER NOTE 5		B2				9		1.0			VISUAL
4	PROTECTIVE FINISH, LUSTERLESS GREEN		C				9		1.0			VISUAL
C	DIMENSIONAL											UG-7C
1	1.0050 ± .0005		B2				9		1.0			H28112274-GC1
2	CHECK PART PROFILE (2 POSITIONS EA.)		B2				9		1.0			COMPARATOR

## INSPECTION PROCEDURE

PART NAME <b>EYELET, INTERFACE</b>	PROGRAM <b>EXT TRIP LINE</b>	IP ISSUE <b>2</b>	SAMPLING PLAN REVISION	DATE <b>7 AUG 78</b>
PART NUMBER <b>929 2981</b>	PART FAMILY	OPERATION INITIALS	REV/MOD <b>B</b>	TALLY NUMBER
QCID <b>M33</b>	INSPECTION DEPARTMENT	APPROVALS	SHEET <b>1</b>	OF <b>1</b>

ITEM	CHARACTERISTICS	RE	CD	SS WK	C SP	L	AQL	N	C	EQUIPMENT
A VISUAL	1 PARTS TO HAVE BLACK OXIDE FINISH PARTS TO BE FREE OF DAMAGE AND MUST BE OF GOOD WORKMANSHIP 3 NO BURRS PERMITTED AS NOTED		C		9		1.0			VISUAL
			D		9		4.0			VISUAL
			B2		9		1.0			VISUAL
D DIMENSIONAL			B2		9		1.0			STD. PIN GAGE
1 .390 $\pm$ .005 DIA			D		9		4.0			1" MICR.
2 .475 $\pm$ .005			B2		9		1.0			DEPTH MICR.
3 .065 $\pm$ .010										

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# INSPECTION PROCEDURE

PART NAME <b>BOBBIN ASSY</b>	PROGRAM <b>EXT. TRIP LINE</b>	IP ISSUE <b>2</b>	SAMPLING PLAN REVISION	DATE <b>7 AUG 78</b>
PART NUMBER <b>929 2982</b>	PART FAMILY	OPERATION NUMBER <b>0195</b>	REV/MOD <b>B</b>	TALLY NUMBER
QID <b>M-33</b>	INSPECTION DEPARTMENT	APPROVALS	SHEET <b>1</b>	OF <b>3</b>

ITEM	CHARACTERISTICS	RE	CD	S WK	S R	C R	SP	L	AQL	N	C	EQUIPMENT
A VISUAL												
1	CHECK FOR PROPER SECURING OF BOTH ENDS OF THE THREAD TO THE BOBBIN  1.) LARGE END BOBBIN: THREAD SHOULD BE IN SLOT AND NOT VISIBLE THROUGH HOLE IN BOBBIN WEIGHT  2.) SMALL FLANGE END BOBBIN: THREAD SHOULD BE AS SHOWN IN PRINT AND SECURED WITH TAPE.  (NOTES	7C	B1				9		1.0			VISUAL
187		3C	B1				9		1.0			VISUAL
2	THE THREAD SHOULD NOT TOUCH THE SMALL SHAFT SECURING THE TWO FLANGES	A1					9		1.0			VISUAL
3	DIRECTION OF WIND SHALL BE CLOCKWISE WHEN VIEWED FROM FLANGE END.	A1					9		1.0			VISUAL
4	WORKMANSHIP: NO LOOSE THREADS SHALL EXTEND OVER BODY OF BOBBIN OTHER THAN AS SHOWN ON PRINT AND NO CRACKS SHALL BE VISIBLE AT MOLD PARTING LINE	A1					9		1.0			VISUAL

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# INSPECTION PROCEDURE

PART NAME BOBBIN ASSEMBLY	PROGRAM EXT. TRIC LINE	IP ISSUE 2	SAMPLING PLAN REVISION	DATE 14 AUG 78
PART NUMBER 9832002	PART FAMILY	OPERATION NUMBER	REV/MOD B	TALLY NUMBER
QTY M-33	INSPECTION DEPARTMENT	APPROVALS	SHEET 2 OF 3	

ITEM	CHARACTERISTICS	RE	CD	S W	S K	C R	SP	L	AQL	N	C	EQUIPMENT
B DIMENSIONAL AND THREAD STRENGTH												
1	CUT OFF POST WITH AN "EXACTO KNIFE" BEING CAREFUL NOT TO CUT THREAD OR UNWIND FROM THE POST. (INSERT A #2-56 X 3/8" S.T. SCREW TO SECURE WEIGHT WHILE MEASURING THE THREAD LENGTH) PLACE POST IN VICE. UNWIND THREAD HOLDING WEIGHT END OF BOBBIN. MEASURE THREAD LENGTH PER NOTE 2 (46 ± 3 FEET)									38	1	BENCH VICE, C CLAMP, TAPE MEASURE.
2	REPEAT UNWINDING PROCEDURE DESCRIBED IN 1 ABOVE FOR REMAINDER OF SAMPLES TO BE TESTED. WITH POST SECURED IN VICE, ATTACH A FORCE GAGE TO THE THREAD AT A DISTANCE OF 2.0 FT FROM VICE. ASSURE THREAD IS STILL WOUND ON POST AND OVER FLANGE. PULL THREAD EVENLY IN A DIRECTION OPPOSITE TO THE POST. THE THREAD AND TAPED END MUST WITHSTAND AN AXIAL FORCE PER NOTE 3 (1.2 LBS)									38	1	BENCH VICE C CLAMP UG-143 L1000

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#A IF THREAD TANGLES, REPLACE UNIT.

# INSPECTION PROCEDURE

PART NAME BOBBIN ASSEMBLY	PROGRAM EXT. TRIP LINE	ISSUE 2	SAMPLING PLAN REVISION	DATE 14 AUG 78
PART NUMBER 9898982	INSPECTION DEPARTMENT	OPERATION NUMBER	REV/MOD B	TALLY NUMBER
QCID M-33	APPROVALS	SHEET 3 OF 3		

ITEM	CHARACTERISTICS	RE	CD	S W	S K	C R	SP	L	AQL	N	C	EQUIPMENT
3	CONTINUE WITH SAMPLE FROM 2. LEAVE SEVERAL WRAPS OF THREAD ON BOBBIN AND SECURE WEIGHT END OF BOBBIN IN VICE SUCH THAT AN AXIAL LOAD MAY BE APPLIED. APPLY AXIAL LOAD PER NOTE 3 (1.2 POUNDS)									38	1	BENCH VICE C LAMP UG-143 L1000

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# INSPECTION PROCEDURE

PART NAME <b>WEIGHT, BOBBIN</b>		PROGRAM <b>EXT. TRIPLINE</b>		SAMPLING PLAN REVISION		DATE <b>7 AUG 78</b>	
PART NUMBER <b>9292983</b>		PART FAMILY		REV/MOD <b>B</b>		TALLY NUMBER	
QID <b>M33</b>		INSPECTION DEPARTMENT		APPROVALS		SHEET <b>1 OF 1</b>	

ITEM	CHARACTERISTICS	RE	CD	S W	S K	C R	SP	L	AQL	N	C	EQUIPMENT
A	VISUAL PARTS TO BE FREE OF DAMAGE AND MUST BE OF 1 GOOD WORKMANSHIP		C				9		1.0			VISUAL
B	DIMENSIONAL											
1	.280 ± .005 DIA.	B2					9		1.0			1" MICR.
2	.180 ± .010	B2					9		1.0			1" MICR.
3	.100 ± .005 DIA.	B2					9		1.0			GH438

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## INSPECTION PROCEDURE

PART NAME	PROGRAM	ISSUE	SAMPLING PLAN REVISION	DATE
BOBBIN	EXT. TRIP LINE	2		7 AUG 78
PART NUMBER	PART FAMILY	OPERATION NUMBER	REV/MOD	TALLY NUMBER
9292985			B	
3EID	INSPECTION DEPARTMENT	APPROVALS		
M33				
				SHEET 1 OF 2

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	LAQL	N	C	EQUIPMENT
A	<b>DIE/MOLD CHECK</b> AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL QE		A2						1	0	MSI
B	BC CERTIFICATION PER NOTE 2 ON PILOT LOT		A2								VISUAL
C	<b>VISUAL</b> PARTS MUST BE FREE OF DAMAGE, EXCESSIVE FLASH (NOTE 5) 1. AND MUST BE OF GOOD WORKMANSHIP. 2. .34 DIA X $\approx$ 135° FLANGE PRESENT 3. PRESENCE OF .070 $\pm$ .002 "D" (NOTE 4)	D7 H7	A2 B2 B2			9 9		1.0 1.0			VISUAL VISUAL VISUAL VISUAL .069 DIA. PIN MUST ENTER FOR INFO. ONLY
D	<b>DIMENSIONAL</b>		B2			9		1.0			1" MICR.
1	.032 $\pm$ .005		B2					1.0			DEPTH MICR.
2	.175 $\pm$ .005		B2					1.0			1" MICR. (6X8142-C2)
3	.685 $\pm$ .010 (2 PLACES)		B2					1.5			KODAK CHART
4	.218 $\pm$ .005		B2					1.0			"
5	.255 $\pm$ .005		B2					1.0			"
6	.015 $\pm$ .005 (2 PLACES)		B2					1.0			"
7	.050 $\pm$ .008		B2					1.0			"

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# INSPECTION PROCEDURE

PART NAME <b>BOBBIN</b>		PROGRAM <b>EXT. TRIP LINE</b>		ISSUE <b>2</b>		SAMPLING PLAN REVISION		DATE <b>8 AUG 78</b>	
PART NUMBER <b>9292985</b>		PART FAMILY		OPERATION NUMBER		REV/MOD <b>B</b>		TALLY NUMBER	
FIELD		INSPECTION DEPARTMENT		APPROVALS		SHEET <b>2 OF 2</b>			

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	AQL	N	C	EQUIPMENT
8	.020 +.005		B2			9		1.0			KODAK CHART
9	.005 MAXR		B2					1.5			"
10	.056 -.005 DIA		B2					1.0			"
11	.058 -.003 DIA		B2					1.0			"
12	.077 -.004 DIA		B2					1.0			"
13	.098 -.004 DIA		B2					1.5			"
14	.005 +.005		B2					1.5			"

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# INSPECTION PROCEDURE

PART NAME		RELE. JE MECHANISM AND CASE ASSEMBLY		PROGRAM		EXT. TRIP LINE		ISSUE		SAMPLING PLAN REVISION		DATE	
PART NUMBER		9292986		INSPECTION DEPARTMENT		APPROVALS		OPERATION NUMBER		REV/MOD		TALLY NUMBER	
QCID		9292986						0295		B		7 AUG 78	
												SHEET 1 OF 2	

ITEM	CHARACTERISTICS	RE	CD	S	S	C	SP	L	AQL	N	C	EQUIPMENT
A	VISUAL											
1	CHECK FOR PRESENCE OF PARTS: .1 "SLEEVE" (9292988) .2 "RELEASE MECHANISM ASSEMBLY" (9292991) .3 "CASE, SENSOR" (9292987)		A2 A2 A2				9 9 9		.65 .65 .65			VISUAL " "
2	PARTS MUST BE FREE OF DAMAGE AND MUST BE OF GOOD WORKMANSHIP						9		1.0			VISUAL
3	1 CHECK FOR DAMAGED "DIAPHRAGM ASSEMBLY" 2 CHECK "SLEEVE" FOR DENTS AND STRAIGHT TABS 3 CHECK "CASE, SENSOR" FOR DENTS. CHECK "RELEASE MECHANISM ASSEMBLY" FOR PROPER SEATING IN "CASE, SENSOR". THE METAL CASE OF THE "HOUSING ASSEMBLY" MUST NOT BE VISIBLE IN OPEN PORTS BETWEEN THE CRIMPED AREAS, AND THE NOZZLE OF THE RELEASE MECHANISM MUST BE LOCATED WITHIN AN OPEN PORT						9 9 9		1.0 1.0 1.0			VISUAL " " VISUAL
4	CHECK FOR CRIMP ALL AROUND											VISUAL



# INSPECTION PROCEDURE

PART NAME PELI OF MECHANISM AND		PROGRAM		DATE	
CASE ASSEMBLY		EXT. TRIP LINE		15 AUG 1978	
PART NUMBER		PART FAMILY		TALLY NUMBER	
9292286		M-33		B	
INSPECTION DEPARTMENT		APPROVALS		SHEET	
M-33				2 OF 2	

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	AQL	N	C	EQUIPMENT
5	CHECK THAT SLEEVE IS LOOSE AND IS NOT BINDING WITHIN THE "CASE, SENSOR".					9		1.0			VISUAL
B	XRAY										
1	XRAY SAMPLE FOR FOUR LOCKING BALLS								50	1	XRAY

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# INSPECTION PROCEDURE

PART NAME <b>CASE SENSOR</b>	PROGRAM <b>EXT. TRIP LINE</b>	IP ISSUE <b>2</b>	SAMPLING PLAN REVISION	DATE <b>7 AUG 78</b>
PART NUMBER <b>9292987</b>	PART FAMILY	OPERATION NUMBER	REV/MOD <b>B</b>	TALLY NUMBER
QCID <b>M33</b>	INSPECTION DEPARTMENT	APPROVALS	SHEET <b>1</b>	OF <b>1</b>

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	AQL	N	C	EQUIPMENT
A	<b>DIE/MOLD CHECK</b> AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL QE		A2						2	0	MSI
B	BQ CERTIFICATION REQUIRED PER NOTE 3		A2						PILOT LOT		VISUAL
C	VISUAL PARTS TO BE FREE OF DAMAGE AND MUST BE 1 OF GOOD WORKMANSHIP		C			9		1.0			VISUAL
D	DIMENSIONAL					9		1.0			MICR.
1	1.930 ± .010		B2								UG-312
2	2.865 ± .010 4 PLACES		B2								STD. PIN GAGE
3	3.495 ± .005 DIA.		B2								MSI
4	4.785 ± .008 PER NOTE 5		B2								MSI
5	5.025 MAX PER NOTE 6		B2								H28109188-G1
6	6.475 ± .005 DIA, PER NOTE 8		B2								STD. PIN GAGE
7	7.446 ± .010 DIA.		B2								

# INSPECTION PROCEDURE

PART NAME <b>SLEEVE</b>		PROGRAM <b>EXT TRIP LINE</b>		DATE <b>7 AUG 79</b>	
PART NUMBER <b>9292988</b>		PART FAMILY <b>2</b>		TALLY NUMBER	
QETO <b>M33</b>		INSPECTION DEPARTMENT <b>APPROVALS</b>		REV. 400 <b>B</b>	
SHEET <b>1 OF 2</b>					

ITEM	CHARACTERISTICS	RE	CD	S S W K	C R	L	SP	LAQL	N	C	EQUIPMENT
A	DIE/MOLD CHECK AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL		A2						2	O	MSI
B	CERTIFICATION										
1	BQ CERTIFICATION PER NOTE 3		A2								VISUAL
C	VISUAL PARTS MUST BE FREE OF DAMAGE AND MUST BE OF 1 GOOD WORKMANSHIP		C				9	1.0			VISUAL
2	DIE RADIUS SIDE AS SHOWN		A2				9	1.0			VISUAL
198											
D	DIMENSIONAL										
1	.0125 ± .0015						9	1.0			MICR. UG-156
2	.061 ± .004 DIA (4 PLACES)		B2				9	1.0			GH 254
3	AL.010 DIA ON .061 DIA										H2811228161
4	.11 - .01 (4 PLACES)										VERNIER CALIPER
5	.75 - .01										"
6	.18 - .01										DEPTH MICR.
7	.355 ± .005										STD. PIN GAGE

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HONEYWELL INC HOPKINS MN DEFENSE SYSTEMS DIV  
PRODUCIBILITY ENGINEERING AND PLANNING (PEP) OF THE XM74 GEMSS --ETC(U)  
AUG 78 R FAIRCHILD  
47419

DAAK10-77-C-0047

NL

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3 OF 3

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A0-3-2

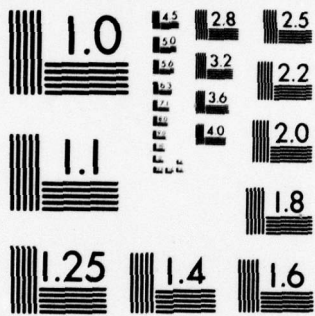


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3-79

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MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS-1963-A

## INSPECTION PROCEDURE

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# INSPECTION PROCEDURE

PART NAME <b>WASHER, CASE</b>		PROGRAM <b>EXT. TRIP LINE</b>		SAMPLING PLAN REVISION		DATE <b>7 AUG 78</b>	
PART NUMBER <b>929 850C</b>		PART FAMILY		REV/MOD <b>B</b>		TALLY NUMBER	
QCID <b>M33</b>		INSPECTION DEPARTMENT		APPROVALS		SHEET <b>1</b> OF <b>1</b>	

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	AQL	N	C	EQUIPMENT
A	CERTIFICATION										
1	BQ CERTIFICATION PER NOTE 3		A2								VISUAL
B	VISUAL AND DIMENSIONAL PARTS TO BE FREE OF DAMAGE AND MUST BE OF 1 GOOD WORKMANSHIP		C								VISUAL
2	.392 ± .002 DIA.		B2			9		1.0			GH3301
3	.474 ± .003 DIA.		B2								1" MICR, UG-111
4	.415 ± .005 DIA.		B2								" " "
5	.150 ± .005		B2								" " "
6	6 A .002 OF .392		B2								OPEN SET-UP

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# INSPECTION PROCEDURE

PART NAME <b>SPRING, BOOSTER</b>	PROGRAM <b>EXT. TRIP LINE</b>	IP ISSUE <b>2</b>	SAMPLING PLAN REVISION	DATE <b>7 AUG 79</b>
PART NUMBER <b>9292990</b>	PART FAMILY	OPERATION NUMBER	REV/MOD <b>B</b>	TALLY NUMBER
QCID <b>M33</b>	INSPECTION DEPARTMENT	APPROVALS	SHEET <b>1 OF 1</b>	

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	AQL	N	C	EQUIPMENT
<b>A CERTIFICATION</b>											
1	BQ CERTIFICATION PER SPECIAL DATA ITEM A		A2								VISUAL
2	BQ CERTIFICATION PER SPECIAL DATA ITEM C		A2								VISUAL
<b>B VISUAL</b>											
1	ENDS CLOSED BUT NOT GROUND SPECIAL DATA ITEM B		C								VISUAL
2	PARTS TO BE FREE OF DAMAGE AND MUST BE OF		B2								VISUAL
	GOOD WORKMANSHIP										
<b>C DIMENSIONAL</b>											
1	.385 $\pm$ .010		B2			9		1.0			STD. PIN GAGE
2	.395 $\pm$ .010 PER NOTE 8		B2			9		1.0			" "
3	3.00 $\pm$ .20		B2								SCALE
4	.455 MAX DIA. PER NOTE 9		B2								M28112284-G1
5	LOAD AT COMPRESSED LENGTH OF .40 IN = 5 $\pm$ 0.5 LB. PER NOTE 7		B2								SPRING TESTER + M28112271-G2

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## INSPECTION PROCEDURE

PART NAME		PROGRAM		IP ISSUE		SAMPLING PLAN REVISION		DATE	
RELEASE MECHANISM ASS'Y		EXT TRIP LINE		2				7 AUG 78	
PART NUMBER		PART FAMILY		OPERATION NUMBER		REV/MOD		TALLY NUMBER	
9292991				0295		B			
QCTO		INSPECTION DEPARTMENT		APPROVALS				SHEET	
M-33								1 OF 1	

ITEM	CHARACTERISTICS	RE	CD	S W	S K	C R	SP	L	AQL	N	C	EQUIPMENT
A VISUAL												
1 360° CRIMP			B2				9		1.0			VISUAL
2 2 SPRING WASHERS PRESENT AND POSITIONED PROPERLY			A2							55	1	XRAY
3 ASSEMBLY TO BE OF GOOD WORKMANSHIP AND FREE OF DEFECTS			A2				9		1.0			VISUAL
4 CHECK THAT TWO "TERMINALS, BREAKWIRE" ARE PRESENT AND PROPERLY STACKED			A2				5		1.0			VISUAL
200												
B DIMENSIONAL												
1 .157 ± .002			A2				9		1.0			H28009650-G-1

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# INSPECTION PROCEDURE

PART NAME DIAPHRAGM ASS'Y		PROGRAM EXT. TRIP LINE		ISSUE 2		SAMPLING PLAN REVISION		DATE 7 AUG 78	
PART NUMBER 9252998		PART FAMILY		OPERATION NUMBER 0395		REV/MOD D		TALLY NUMBER	
INSPECTION DEPARTMENT		APPROVALS						SHEET 1 OF 1	

ITEM	CHARACTERISTICS	RE	CD	S WK	C R	SP	L	AQL	N	C	EQUIPMENT
A VISUAL											
1 PRESENCE OF PARTS: DIAPHRAGM, EYELET, PLATE, AND GASKET			A2			9		1.0			VISUAL
2 CHECK "FLARE ALL AROUND" PER NOTES 2 & 3			B2			9		1.5			VISUAL
3 CHECK PARTS FOR FOLLOWING:											
1.) BROKEN OR SPLIT DIAPHRAGM PLATE NOZZLE			B2			9		1.0			VISUAL
2.) DIAPHRAGM DAMAGED IN FUNCTIONAL AREA (FUNCTIONAL AREA: THAT PORTION HALF WAY UP THE O D FLANGE TO THE EYELET)			B2			9		1.0			VISUAL
3.) EYELET FLANGE TO BE FLAT AND UNDEFORMED			B2			9		1.0			VISUAL
4 CHECK GASKET FOR PROPER ORIENTATION OF TONGUE PER NOTE 4			B2			9		1.0			VISUAL
5 CHECK DIAPHRAGM FOR NOTCH AND ORIENTATION PER NOTE 5			B2			9		1.0			VISUAL
6 DIMENSIONAL											
1.016 ± .003 (METAL TO METAL)			A1			9		1.0			VISUAL

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# INSPECTION PROCEDURE

PART NAME COVER, POST		PROGRAM EXT. TRIP LINE		ISSUE 2		SAMPLING PLAN REVISION		DATE 7 AUG 78	
PART NUMBER 9298575		PART FAMILY		OPERATION NUMBER		REV/MOD		TALLY NUMBER	
QCTD		INSPECTION DEPARTMENT		APPROVALS		ORIG		SHEET 1 OF 1	

ITEM	CHARACTERISTICS	RE	CD	S W	S K	C R	SP	L	AQL	N	C	EQUIPMENT
A	<p><b>DIE/MOLD CHECK</b></p> <p>AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL Q'E</p>	A2								2	0	MSI
B	<p><b>VISUAL AND DIMENSIONAL CHECK THE PART PROFILE USING CHART (ATTEMPT TO SELECT SAMPLE SUCH THAT EACH CAVITY OF MOLD IS EQUALLY REPRESENTED)</b></p> <p>2 CAVITY IDENT PER NOTE 3</p> <p>3 PARTS TO BE FREE OF DAMAGE AND OF GOOD WORKMANSHIP</p>	B2					9	1.0				H28112269 6-1 + COMPARATOR
		D				9	1.0					VISUAL
		C				9	1.0					VISUAL



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# INSPECTION PROCEDURE

PART NAME		PROGRAM		ISSUE		SAMPLING PLAN REVISION		DATE	
WIRE MAGNETIC ELECTRICAL		EXT. TRIP LINE		2				7 AUG 78	
PART NUMBER		PART FAMILY		OPERATION NUMBER		REV/MOD		TALLY NUMBER	
9298576						ORIG			
QID		INSPECTION DEPARTMENT		APPROVALS				SHEET	
M33								1 OF 1	

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	AQL	N	C	EQUIPMENT
A	VISUAL AND DIMENSIONAL										
1	IDENTIFY PART AND DASH NUMBER- PART IDENTIFICATION AND CONFIGURATION		B1						BA		VISUAL
2	PARTS TO BE OF GOOD WORKMANSHIP AND FREE OF DAMAGE.		E 202			9		.65			VISUAL
3	CHECK WIRE - APPROXIMATELY ONE FOOT FROM ONE END, THE OUTSIDE DIAMETER MUST FALL WITHIN PRINT SPECIFICATION		E 201					*A			MICROMETER
4	CHECK FOR POLYURETHANE COATING		C					*A			VISUAL

203

\*A

SELECT ONE SAMPLE FROM THE BEGINNING OF EACH SPOOL OR COIL. FAILURE OF THE SAMPLE SHALL REJECT THE SPOOL/COIL FOR USE IN PRODUCTION



# INSPECTION PROCEDURE

PART NAME		PROGRAM		ISSUE		SAMPLING PLAN REVISION		DATE	
RIVE RETAINING, BREAKWIRE		EXT. TRIP LINE		2				7 AUG 78	
PART NUMBER		PART FAMILY		OPERATION: CUMUL		REV/MOD		TALLY NUMBER	
9298577						ORIG			
QCTD		INSPECTION DEPARTMENT		APPROVALS				SHEET	
M33								1 OF 1	

ITEM	CHARACTERISTICS	RE	CD	SS WK	C R	SP	L	AQL	N	C	EQUIPMENT
A	DIE/MOLD CHECK AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL		A2						3	0	MSI
B	VISUAL										
1	45° - 5° PRESENCE OF		B2	9	"			1.0			VISUAL
2	.005 ±.003 PRESENCE OF		B2	"	"			1.0			"
3	DIE RADIUS SIDE (AS NOTED)		B2	9	"			1.0			"
4	PARTS TO BE OF GOOD WORKMANSHIP AND FREE OF DAMAGE		C	"	"			1.0			"
C	DIMENSIONAL										
1	.095 ±.005		B2	9	"			1.0			STD MICR.
2	.058 ±.002		B2								GH 2000
3	.010 ±.001		B2								MICR
4	.013 ±.003		B2								MICR
5	Ø .002 DIA		B2								OG 8070

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# INSPECTION PROCEDURE

PART NAME <b>SPRING-EJECTION</b>	PROGRAM <b>EXT. TRIP LINE</b>	ISSUE <b>2</b>	SAMPLING PLAN REVISION	DATE <b>7 AUG 78</b>
PART NUMBER <b>9298578</b>	PART FAMILY	OPERATION NUMBER	REV/MOD <b>ORIG</b>	TALLY NUMBER
QCID <b>M33</b>	INSPECTION DEPARTMENT	APPROVALS	SHEET OF	

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	AQL	N	C	EQUIPMENT
A	CERTIFICATION										
1	BQ CERT PER NOTE 2		A2								VISUAL
2	BQ CERT PER NOTE 3		A2								VISUAL
B	DIMENSIONAL										
1	.455 MAX DIA. PER NOTE 11		B2								RING GAGE AND SPRING TESTER AND WEIGHT-LE SPRING TEST NUT
2	LOAD AT COMPRESSED LENGTH PER NOTE 10		A2			9		1.0			VERNIER CALIPER
3	2.00 ± .20 FREE LENGTH		B2								STD. PIN GAGE
4	.365 ± .010 PER NOTE 6		B2								

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# INSPECTION PROCEDURE

PART NAME <b>CAP</b>	PROGRAM <b>EXT. TRIP LINE</b>	ISSUE <b>2</b>	SAMPLING PLAN REVISION	DATE <b>7 AUG 78</b>
PART NUMBER <b>9298579</b>	PART FAMILY	OPERATION NUMBER	REV/MOD <b>ORIG</b>	TALLY NUMBER
QTY <b>M33</b>	INSPECTION DEPARTMENT	APPROVALS	SHEET <b>1</b> OF <b>1</b>	

ITEM	CHARACTERISTICS	RE	CD	S W	S R	C SP	L	AQL	N	C	EQUIPMENT
A	<b>DIE/MOLD CHECK</b> AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL		A2						3	0	MSI
B	<b>CERTIFICATION</b>										
1	BC CERTIFICATION PER NOTE 2		A2						PLAT LOT		VISUAL
C	<b>VISUAL</b>										
1	208 PRESENCE OF ZINC COATING (NOTE 3)		C			9		1.0			VISUAL
D	<b>DIMENSIONAL</b>										
1	.474 - .010 DIA		B2			9		1.0			1" MIC
2	.12 ±.01 (4PLACES)		B2			9		1.0			1" MIC

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PART NAME	PROGRAM	ISSUE	SAMPLING PLAN REVISION	DATE
BALL LOCKING	EXT. TRIP LINE	2		7 AUG 78
PART NUMBER	PART FAMILY	OPERATION (1) IMPL R	REV/MOD	TALLY NUMBER
9228581			ORIG	
AGEID	INSPECTION DEPARTMENT	APPROVALS	SHEET	
M33			OF	

ITEM	CHARACTERISTICS	RE	CD	S WK	S CR	C SP	L AQL	N	C	EQUIPMENT
A	VISUAL									
1	PARTS MUST BE OF GOOD WORKMANSHIP AND FREE OF NICKS, DENTS, SCRATCHES OR OTHER DAMAGE		A2			9	1.0			VISUAL
B	DIMENSIONAL									
1	.0469 ± .0002 SPHER. DIA.		B2			9	1.0			UG 309 A
C	SPECIAL									
1	HARDNESS PER NOTE 2		A2			9	1.0			ROCKWELL HARDNESS TESTER MICROSCOPE AND UG-158
2	FINISH PER NOTE 1		A2			9	1.0			

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# INSPECTION PROCEDURE

PART NAME <b>SPRING, WASHER</b>		PROGRAM <b>EXT. TRIP LINE</b>		DATE <b>7 AUG 78</b>	
PART NUMBER <b>9298582</b>		PART FAMILY <b>ORIG</b>		TALLY NUMBER	
QCID <b>M33</b>		INSPECTION DEPARTMENT		REV/MOD	
APPROVALS		OPERATION NUMBER <b>2</b>		SHEET <b>1</b> OF <b>1</b>	

ITEM	CHARACTERISTICS	RE	CD	SS WK	SC R	SP	L	AQL	N	C	EQUIPMENT
A	CERTIFICATION										
1	BQ CERTIFICATION PER NOTE 2.		A2								VISUAL
D	VISUAL PARTS MUST BE OF GOOD WORKMANSHIP AND 1 TO BE FREE OF DAMAGE AND BURRS.		C			9		1.0			VISUAL
C	SPECIAL SPRING LOAD REQUIREMENTS PER NOTE 3.		A2			9		1.0			MSI

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# INSPECTION PROCEDURE

PART NAME <b>RING BALL LOCK</b>	PROGRAM <b>EXT. TRIP LINE</b>	DATE <b>7 AUG 78</b>
PART NUMBER <b>929B586</b>	OPERATION NUMBER <b>2</b>	TALLY NUMBER
QCTD <b>M33</b>	INSPECTION DEPARTMENT <b>APPROVALS</b>	REVISION <b>ORIG</b>
SHEET <b>1 of 2</b>		

ITEM	CHARACTERISTICS	RE	CD	SS WK	CS R	SP	L	AQL	N	C	EQUIPMENT
A	DIE/MOLD CHECK AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL								2	0	M 51
B	CERTIFICATION										
1	BC CERTIFICATION FOR NOTE 2		A2								VISUAL
C	VISUAL										
1	CAVITY IDENTIFICATION PER NOTE 3		D			9		1:0			VISUAL
D	DIMENSIONAL										
1	.261 ± .005 (2 PLACES)		B2		9		1:0				GAGE (GX8017)
2	.092 ± .004		B2								MICR.
3	.307 ± .004 DIA.		B2								MICR.
4	.171 ± .004 DIA.		B2								GH 3369
5	.294 ± .003 (2 PLACES)		B2								UG-111 R
6	.075 ± .004		B2								UG 18+1
7	.146 ± .002 DIA		B2								GH 921
8	.004 DIA OF .171		B2								H28009940-61462

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# INSPECTION PROCEDURE

REF NAME	PROGRAM	ISSUE	SAMPLING PLAN REVISION	DATE
RING BALL LOCK	EXT. TRIP LINE	2		9 AUG 78
PART NUMBER	PART FAMILY	OPERATION NUMBER	REV/MOD	TALLY NUMBER
9228586			ORIG	
ZEID	INSPECTION DEPARTMENT	APPROVALS		SHEET
				2 OF 2

ITEM	CHARACTERISTICS	RE	CD	S WK	C R	S P	L	AQL	N	C	EQUIPMENT
9	A .002 TOTAL OF .261		B2			9		1.0			TOOL MIC.
10	B .003 TOTAL OF .005		B2			9		1.0			TOOL MAKERS MICROSCOPE

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# INSPECTION PROCEDURE

PART NAME <b>HOUSING ASSY</b>		PROGRAM <b>EXT TRIP LINE</b>		ISSUE <b>2</b>		SAMPLING PLAN REVISION		DATE <b>7 AUG 78</b>	
PART NUMBER <b>9298587</b>		PART FAMILY		OPERATION NUMBER <b>0125</b>		REV/MOD <b>ORIG</b>		TALLY NUMBER	
INSPECTION DEPARTMENT <b>M-32</b>		APPROVALS		REVISIONS		REVISIONS		SHEET <b>1 OF 1</b>	

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	AQL	N	C	EQUIPMENT
A	VISUAL										
1	PRESNCE AND POSITIONING OF CUP AND TUBE		B2			9		1.0			VISUAL
2	PRESNCE OF "CRIMP ALL AROUND"		B2			9		1.0			VISUAL
3	ASSEMBLY SHALL BE OF GOOD WORKMANSHIP AND FREE OF DEFECTS		B2			9		1.0			VISUAL
B	DIMENSIONAL										
1	1.005 MAX		B2			9		1.0			VISUAL OR DIAL INDICATOR
2	Φ A B .004 DIA		B1			9		1.0			OG-8026
211											
C	FUNCTIONAL										
1	CRIMP SHALL WITHSTAND A SEPARATION FORCE OF 10.0 LBS WITHOUT LODSENING. (NOTE 2)		A1						35	0	OG-8036

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# INSPECTION PROCEDURE

PART NAME <b>CUP HOUSING</b>		PROGRAM <b>EXT TRIP LINE</b>		IP ISSUE <b>2</b>	SAMPLING PLAN REVISION	DATE <b>7 AUG 78</b>
PART NUMBER <b>9298588</b>		PART FAMILY		OPERATION NUMBER	REV/MOD <b>ORIG</b>	TALLY NUMBER
INSPECTION DEPARTMENT <b>M33</b>		APPROVALS		SHEET <b>1 OF 2</b>		

ITEM	CHARACTERISTICS	RE	CD	S	S	C	SP	L	AQL	N	C	EQUIPMENT
				W	K	R						
A	<b>DIE/MOLD CHECK</b> AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL Q.E.									2	0	
B	<b>CERTIFICATION</b> 1 BQ CERT PER NOTE 2		A2						PILOT LOT			VISUAL
C	<b>VISUAL</b> PARTS TO BE FREE OF DAMAGE AND MUST BE OF GOOD WORKMANSHIP		C				9		1.0			VISUAL
D	<b>DIMENSIONAL</b>		B2				9		1.0			GAGE (EX8134)
1	.121 +.002 DIA.											GH510
2	.045 +.005 DIA											GH016
3	.050 +.003 DIA											1" MICR
4	.352 -.004 DIA											1" MICR
5	.132 +.003											1" MICR
6	.474 -.003 DIA.											STD. PIN GAGE
7	.311 +.004 DIA.											H2811103-61
8	.036 -.004 (2 PLACES)											

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# INSPECTION PROCEDURE

PART NAME		PROGRAM		ISSUE		SAMPLING PLAN REVISION		DATE	
CUP HOUSING		EXT. TRIP LINE		2				9 AUG 78	
PART NUMBER		PART FAMILY		OPERATION NUMBER		REV/MOD		TALLY NUMBER	
9298588						ORIG			
QCTD		INSPECTION DEPARTMENT		APPROVALS				SHEET	
								2 OF 2	

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	C SP	L	AQL	N	C	EQUIPMENT
9	.010-.003 (2PLACES)		B2			9		1.5			H2811103-61
10	4 A.004 OF .474 DIA.		B2					1.0			OG-8067
11	4 A.004 OF .311 DIA.		B2								"
12	4 A.B.C.004 OF .050 DIA. & .045 DIA.		B2								OG-8069
13	4 D.A.004 OF .352 DIA.		B2								OG-8068
14	4 A.B.004 TOTAL OF .036		B2								H2811103-61

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# INSPECTION PROCEDURE

PART NAME <b>TUBE HOUSING</b>		PROGRAM <b>EXT. TRIP LINE</b>		ISSUE <b>2</b>		SAMPLING PLAN REVISION		DATE <b>7 AUG 78</b>	
PART NUMBER <b>9298589</b>		PART FAMILY		OPERATION NUMBER		REV/MOD <b>ORIG</b>		TALLY NUMBER	
QID <b>M33</b>		INSPECTION DEPARTMENT		APPROVALS		SHEET <b>1 OF 2</b>			

ITEM	CHARACTERISTICS	RE	CD	S WK	C R	SP	LAQL	N	C	EQUIPMENT
A	CERTIFICATION									
1	BC CERTIFICATION PER NOTE 2		A2							VISUAL
B	VISUAL PARTS MUST BE OF GOOD WORKMANSHIP AND MUST BE FREE OF DAMAGE AND BURRS		C			9	1.0			VISUAL
214										
C	DIMENSIONAL		BR			9	1.0			1" MICR
1	.144 ± .003 DIA.					9	1.0			1" MICR
2	.120 ± .002 DIA.					9	1.0			GH 2087 GAGE
3	.085 ± .004									
4	Ø 1.002 DIA ON .085									
5	Ø 1.002 DIA ON .120									
6	Ø 1.004 TOTAL ON .040									
7	Ø 1.004 TOTAL									
8	.045 ± .005									
9	.032 ± .010									
10	.049 ± .002									
11	.040 ± .003 (2 PLACES)									
12	.050 ± .003 DIA (2 PLACES)									

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# INSPECTION PROCEDURE

PART NAME TUBE HOUSING	PROGRAM EXT. TRIP LINE	ISSUE 2	SAMPLING PLAN REVISION 1	DATE 9 AUG 78
PART NUMBER 9298589	PART FAMILY	OPERATION NUMBER	REV/MOD ORIG	TALLY NUMBER
QCID	INSPECTION DEPARTMENT	APPROVALS	SHEET 2 OF 2	

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	AQL	N	C	EQUIPMENT
13	.017 MIN		82			9		1.0			GAGE
14	.010-.002										
15	.030-.005										
16	.060-.004										
17	.126-.003										



## INSPECTION PROCEDURE

PART NAME <b>TAPE</b>	PROGRAM <b>EXT TRIP LINE</b>	ISSUE <b>2</b>	SAMPLING PLAN REVISION	DATE <b>7 AUG 78</b>
PART NUMBER <b>9208591</b>	INSPECTION DEPARTMENT	OPERATION NUMBER	REV/MOD <b>ORIG</b>	TALLY NUMBER
QCID <b>M33</b>	APPROVALS	SHEET <b>1 OF 1</b>		

ITEM	CHARACTERISTICS	RE	CD	SIS	WIK	C	SP	L	AQL	N	C	EQUIPMENT
A	VISUAL VERIFY THAT TAPE IS PROPER PART NUMBER (853) AND 1 FROM AN APPROVED SOURCE (3M)									EA LOT		VISUAL
2	SHELF LIFE - 9 MONTHS, CODE D									EA LOT		MANUAL

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# INSPECTION PROCEDURE


PART NAME <b>THREAD, POLYESTER</b>		PROGRAM <b>EXT. TRIP LINE</b>		DATE <b>7 AUG 78</b>	
PART NUMBER <b>SP08592</b>		ISSUE <b>2</b>		TALLY NUMBER	
QCTD <b>M33</b>		OPERATION NUMBER		REV/MOD <b>ORIG</b>	
INSPECTION DEPARTMENT		APPROVALS		SHEET <b>1 OF 1</b>	

ITEM	CHARACTERISTICS	RE	CD	SS WK	SC R	CS R	LAQL	N	C	EQUIPMENT
A SPECIAL										
1	INSPECT THREAD FOR BREAKING STRENGTH AS SPECIFIED IN NOTE J-B.		E 101				HA	EA SPOOL	O	PULL FORCE GAGE
B VISUAL										
1	THREAD COLOR PER NOTE J-C. NO VISIBLE DAMAGE, CUTSTRANDS, OR SHARP KINKS -		C					EA SPOOL	O	VISUAL
2	CHECK 1 PIECE 5' LONG FROM EACH SPOOL		C					EA SPOOL	O	VISUAL

HA  
SELECT ONE SAMPLE FROM THE BEGINING OF EACH SPOOL.  
FAILURE OF THE SAMPLE SHALL REJECT THE SPOOL FOR  
USE IN PRODUCTION,

# INSPECTION PROCEDURE

PART NAME <b>PLATE DIAPHRAGM</b>		PROGRAM <b>EXT. TRIP LINE</b>		SAMPLING PLAN REVISION		DATE <b>7 AUG 78</b>	
PART NUMBER <b>9298597</b>		PART FAMILY		REV/MOD <b>ORIG</b>		TALLY NUMBER	
QCD <b>M33</b>		INSPECTION DEPARTMENT		APPROVALS		SHEET <b>1 OF 2</b>	

ITEM	CHARACTERISTICS	RE	CD	SS WK	SS CR	SP	L	AQL	N	C	EQUIPMENT
A	<b>DIE/MOLD CHECK</b> AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL Q E		A2						2	0	MSI
B	<b>CERTIFICATION</b> BC CERTIFICATION PER NOTE 2		A2						ADT LOT		VISUAL
C	<b>VISUAL</b> PARTS TO BE OF GOOD WORKMANSHIP AND MUST BE FREE OF DAMAGE AND EXCESSIVE FLASH.		C			9		1.0			VISUAL
1	1 OF DAMAGE AND EXCESSIVE FLASH.		D			9		1.0			VISUAL
2	2 CAVITY IDENTIFICATION PER NOTE 4		M			9		.40			VISUAL
3	3 INLET PORT MUST NOT BE BLOCKED		NO			9		1.0			VISUAL
4	4 NO CRACKS PRESENT WHEN SLIGHT FINGER PRESSURE APPLIED AS INDICATED 		A2								
	① APPLY FINGER PRESSURE (SLIGHT) ② INSPECT FOR CRACKS										
C	<b>DIMENSIONAL</b>		B2								MICR.
1	1 .102 - .005 DIA.		B2			18		1.0			MICR.
2	2 .450 - .005 DIA		B2			18		1.0			MICR.

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# INSPECTION PROCEDURE

PART NAME <b>PLATE DIAPHRAGM</b>	PROGRAM <b>EXT. TRIP LINE</b>	ISSUE <b>2</b>	SAMPLING PLAN REVISION	DATE <b>9 AUG 78</b>
PART NUMBER <b>922B597</b>	PART FAMILY	OPERATION NUMBER	REV/MOD <b>ORIG</b>	TALLY NUMBER
INSPECTION DEPARTMENT	APPROVALS	SHEET <b>2 OF 2</b>		

ITEM	CHARACTERISTICS	RE	CD	S WK	S C R	SP	L	AQL	N	C	EQUIPMENT
3	.162 +.004 DIA.		B2			9		10			GH1485
4	.052 -.004										GH158 +.001 INDICATOR
5	PRESENCE OF .006 +.004 (4 PLACES)										MICROSCOPE
6	.028 +.004 DIA										GH2046
7	.070 -.003 DIA										UG-111
8	.070 -.010 (2 PLACES)										M2B1586-G1, 6-158,
9	.026 +.004										16158A OR EQUIVALENT
10	.230 +.004 DIA.										" " " "
											GH2606

# INSPECTION PROCEDURE

PART NAME <b>DIAPHRAGM</b>	PROGRAM <b>EXT TRIP LINE</b>	ISSUE <b>2</b>	SAMPLING PLAN REVISION	DATE <b>7 AUG 78</b>
PART NUMBER <b>9298528</b>	PART FAMILY	OPERATION NUMBER	REV/MOD <b>ORIG</b>	TALLY NUMBER
QCID <b>M33</b>	INSPECTION DEPARTMENT	APPROVALS	SHEET <b>1 OF 1</b>	

ITEM	CHARACTERISTICS	RE	CD	S WK	S C	R	SP	L	AQL	N	C	EQUIPMENT
A	<b>DIE/MOLD CHECK</b> AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL QE		A2							2	0	MSI
B	<b>VISUAL</b> 1 PARTS MUST BE OF GOOD WORKMANSHIP 2 PARTS MUST BE CAPABLE OF PASSING OVER .250 DIA MANDREL WITHOUT SPLITTING PER NOTE 3 3 PARTS MUST BE FREE OF DAMAGE		E 201 E 101 C				9		.65 .40 1.0			VISUAL .250 PIN MICROSCOPE VISUAL
C	<b>CERTIFICATION</b> 1 BC CERTIFICATION PER NOTE 2 (MATERIAL) 2 BC CERTIFICATION PER NOTE 4 (VIRGIN MATERIAL)		A2 A2									VISUAL VISUAL

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PROJECT NAME EYELET, DIAPHRAGM	PROGRAM EXT. TRIP LINE	ISSUE 2	SAMPLING PLAN REVISION	DATE 7 AUG 78
PART NUMBER 9290599	PART FAMILY	OPERATION NUMBER	REV/MOD ORIG	TALLY NUMBER
INSPECTION DEPARTMENT	APPROVALS	SHEET 1 OF 1		

**A** **DIE / MOLD CHECK**

**AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL Q E**

**B | CERTIFICATION**

1	BC CERT FOR NOTE 2 (MATERIAL)
---	-------------------------------

C | VISUAL

VISUAL  
 PARTS MUST BE OF GOOD WORKMANSHIP AND TO BE FREE  
 OF DAMAGE

# DIMENSIONAL

1.142 - .004 DIA.

2/.120 MIN DIA

$$3.080 - .005$$

4.002 R MAX

S  $\varnothing$  A .005 OF .210 DIA

## EQUIPMENT

MS1

# VISUAL

## VISUAL

GH 3360

STD PIN GAGE

1" MICR.

## VISUAL

H 28010136-61



# INSPECTION PROCEDURE

PART NAME <b>TERMINAL, BREAKWIRE</b>		PROGRAM <b>EXT. TRIP LINE</b>		DATE <b>7 AUG 78</b>	
PART NUMBER <b>929B601</b>		OPERATION NUMBER <b>2</b>		TALLY NUMBER	
INSPECTION DEPARTMENT <b>1A33</b>		APPROVALS		REV/MOD <b>ORIG</b>	
QCID		SHEET		1 OF 1	

ITEM	CHARACTERISTICS	RE	CD	SIS WK	CSP R	L AQL	N C	EQUIPMENT
A	<b>DIE/MOLD CHECK</b> AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL QE		A2				2 0	M 91
B	<b>CERTIFICATION</b> 1 BC CERT. PER NOTE 3.		B2					DESTRUCTIVE TEST AND MEASURE THICKNESS - GAGE
C	<b>VISUAL</b> 1 PROTECTIVE FINISH PER NOTE 3 2 PARTS TO BE PROPERLY IDENTIFIED AS -1 OR -2		B2		9	1.0		VISUAL
			B2		9	1.0		VISUAL

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PART NAME	PROGRAM	ISSUE	SAMPLING PLAN REVISION	DATE
URETHANE COMPOUND	EXT. TRIP LITE	2		17 AUG 78
PART NUMBER	PART FAMILY	OPERATION NUMBER	REV/MOD	TALLY NUMBER
9298617				
QCID	INSPECTION DEPARTMENT	APPROVALS		SHEET
M33				OF 1

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# INSPECTION PROCEDURE

PART NAME <b>GASKET, FIBROUS</b>		PROGRAM <b>EXT. TRIP LINE</b>	IP ISSUE <b>2</b>	SAMPLING PLAN REVISION	DATE <b>7 AUG 78</b>
PART NUMBER <b>929 B618</b>		PART FAMILY	OPERATION NUMBER	REV/MOD <b>ORIG</b>	TALLY NUMBER
INSPECTION DEPARTMENT <b>M 33</b>		APPROVALS	SHEET <b>1 OF 1</b>		

ITEM	CHARACTERISTICS	RE	CD	S S C S P L A Q L N C	W K R	10	10	EQUIPMENT
A	<b>DIE/MOLD CHECK</b> AN ACCEPTABLE DIE/MOLD CHECK FOR EACH DIE OR MOLD CAVITY TO THE LATEST PRINT REVISION FOR EACH VENDOR MUST BE IN THE INSPECTION FOLDER. IF NOT, CALL		A2					MSI
B	<b>DIMENSIONAL</b>							
1	.365 - .010		B2		9	10		COMPARATOR + CHART
2	.158 +.004 DIA.		B2		9	10		"

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INSPEC JME

PART NAME PAPER SKETCH		PROGRAM EXT TRIP LINE		ISSUE 2		SAMPLING PLAN REVISION		DATE 9 AUG 78	
PART NUMBER 9298619		PART FAMILY		OPERATION NUMBER		REV/MOD ORIG		TALLY NUMBER	
INSPECTION DEPARTMENT M33		APPROVALS						SHEET 1 OF 1	

ITEM	CHARACTERISTICS	RE	CD	S WK	S C	SP	L	AQL	N	C	EQUIPMENT
A	CERTIFICATION										
1	BC CERT REQUIRED PER NOTE 4 (REQUIREMENTS)		A2								VISUAL
B	VISUAL										
1	VERIFY THAT PART NUMBER IS CORRECT PARTS MUST BE OF GOOD WORKMANSHIP AND MUST BE FREE OF DAMAGE.		A2								VISUAL
2			C			9		1.0			VISUAL

APPENDIX D  
COST DATA

## APPENDIX D COST DATA

Cost data has been developed for the following contract requirements:

- Unit Product Cost (UPC) for the baseline design
- UPC for production engineered design
- Production facility costs for a 200,000 units per month production line rate.

Cost summaries on the following pages provide detail on costs as follows:

UPC Baseline	\$4.514 each
UPC PEP Version	\$3.033 each

The UPC costs include the cost of the Release Mechanism as provided by the ADAM line. The \$1.568 each price for the Release Mechanism is a figure from 1977 production. The current 1978 price shows a reduction in this figure which would further reduce the \$3.033 price. Additional savings will also be realized when the leadwire soldering is automated and when anticipated Release Mechanism reductions are accomplished, the price per sensor should be approximately \$2.25.

The UPC costs have not been developed to show a cost vs. quantity relationship. Information if presented however, which shows the line rates for various combinations of assembly machines. Cost information for line rates other than 200,000 units per month can be furnished upon request.

The production facility costs are also presented in detail in this appendix which includes separate cost information for debug and machine demonstration hardware.



## **XM74 GEMSS EXTENDED TRIPLINE SENSOR FINAL UPC**

**Type of Contract:** PEP Contract DAAK10-77-C-0047

**Customer** ARRADCOM

**Type of Proposal:** C

**Procurement Manager:** All material estimates are same as preliminary UPC of July, 1977. H. Spillers reviewed MPR's in July, 1977 for parts unique to XM74. ADAM Sensor parts taken from ADAM Contract 0048.

**Production Manager:** Ken Jenson

**Quality/Delivery Period:** Quantity of 200,000/Month was stated in the contract. It is assumed that this would be for one year. Since no delivery time was specified, the costs are given in 1977 dollars.

### **Groundrules/Assumptions:**

1. This is a budgetary estimate.
2. Estimate is in 1977 dollars and is assumed to be the second year of production (i.e., 2,400,000 sensors) after an initial quantity of 165,000. Maximum rate is to be 200,000 per month.
3. The cost of the release mechanism assembly is taken from the current ADAM contract with adjustments for total quantity and assembly losses for assembly into sensors.
4. Cost of "D Test" units have been estimated on the basis of 4 lots of 50 units each per month. The cost for the "D Test" units are covered in the assembly scrap rate.

5. The UPC is based on 1:8:5 production.
6. It is assumed that the use of Government Furnished Equipment will be rent free.
7. All tooling and equipment necessary for quantity estimated has been designed, de-bugged and will be available. The non-recurring costs for the machines, tooling and facilities are not included in this estimate.
8. The item design is as described by the drawings and specifications transmitted to Honeywell from ARRADCOM and received 7 June 1977 and revised in accordance with producibility recommendations submitted by Honeywell to ARRADCOM 15 August 1978.
9. Machine usage (i.e., the production line) for this production is based on the following assumptions:
  - a. Honeywell OML machines have been assumed for assembly operations. The specific machine requirements are listed on the attached table.
  - b. One operation remains as a manual, semi-automatic operation, namely, the soldering of (2) leadwires to the sensor. More development work is required for full automation of leadwire soldering.

Unit Cost/Price:

	Basic Cost July, 1977	Basic Cost 30 August 1978
Quantity	2,400,000	
Material	0.644	0.643
Factory Labor & Burden	2.346	1.129

Eng. Labor & Burden	0.211	0.201
Tooling plus shrinkage/surplus	<u>0.088</u>	<u>0.052</u>
Factory Cost	3.279	2.025
G&A at 19.7%	<u>0.646</u>	<u>0.399</u>
Total Cost	3.925	2.424
Profit at 15.0%	<u>0.589</u>	<u>0.364</u>
Price	4.514	2.788



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HONEYWELL COST SUMMARY									
DESCRIPTION: XM74 EXTENDED RANGE TRIPLINE JENIOR ASSEMBLY - CONTRACT C-0047									
SUBJECT: EXHAUSTORY UPG IN 1977B FINAL UPG FOR PEP									
RATE: 20,000,000 PER ONE YEAR AFTER 1ST YEAR OF 15,000,000									
Est. of 1977 (INITIAL)									
CONTRACT	FNR		REVENUE		EXPENSE		NET		
DATE: 8/10/78	8/10/78		8/10/78		8/10/78		8/10/78		
COST PER	EA	EA	EA	EA	EA	EA	EA	EA	EA
REVENUE SHEET	2,34	34	2,34	34	2,34	34	2,34	34	
SC MATERIAL									
OTHER MATERIAL									
SUBTOTAL MATERIAL	0.425	0.189	0.425	0.189	0.425	0.189	0.425	0.189	
MATERIAL	0.021	0.009	0.021	0.009	0.021	0.009	0.021	0.009	
TOTAL MATERIAL	0.446	0.198	0.446	0.198	0.446	0.198	0.446	0.198	
PRODUCTION LABOR	0.161	0.756	0.161	0.756	0.161	0.756	0.161	0.756	
PRODUCTION ENGINEER	0.339	1.021	0.339	1.021	0.339	1.021	0.339	1.021	
INSPECTION LABOR	0.018	0.021	0.018	0.021	0.018	0.021	0.018	0.021	
INSPECTION ENGINEER	0.014	0.006	0.014	0.006	0.014	0.006	0.014	0.006	
SUBTOTAL	0.978	2.012	0.978	2.012	0.978	2.012	0.978	2.012	
SUPPLIES	0.008	0.016	0.008	0.016	0.008	0.016	0.008	0.016	
QUALITY ENGR. LABOR	0.014	0.009	0.014	0.009	0.014	0.009	0.014	0.009	
QUALITY ENGR. BURDEN	0.011	0.007	0.011	0.007	0.011	0.007	0.011	0.007	
PROD'N ENGR. LABOR	0.035	0.027	0.035	0.027	0.035	0.027	0.035	0.027	
PROD'N ENGR. BURDEN	0.026	0.020	0.026	0.020	0.026	0.020	0.026	0.020	
PROD'N ENGR. LABOR	0.024	0.004	0.024	0.004	0.024	0.004	0.024	0.004	
PROD'N ENGR. BURDEN	0.021	0.003	0.021	0.003	0.021	0.003	0.021	0.003	
TOOLING/GAGES	0.022	0.042	0.022	0.042	0.022	0.042	0.022	0.042	
FACTORY COST	1.139	2.140	1.139	2.140	1.139	2.140	1.139	2.140	
SGI/Tech. 0	0.224	0.422	0.224	0.422	0.224	0.422	0.224	0.422	
SUBTOTAL	1.363	2.562	1.363	2.562	1.363	2.562	1.363	2.562	
GEN. & ADM. 0	0.205	0.384	0.205	0.384	0.205	0.384	0.205	0.384	
TOTAL COST	1.568	2.946	1.568	2.946	1.568	2.946	1.568	2.946	
PROFIT	0.205	0.384	0.205	0.384	0.205	0.384	0.205	0.384	
PRICE	1.773	3.330	1.773	3.330	1.773	3.330	1.773	3.330	
CAS416	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	
TOTAL PRICE	1.773	3.330	1.773	3.330	1.773	3.330	1.773	3.330	

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[illegible]



# Honeywell

DESCRIPTION: ST. PETERS TOWER SERVICE LESS APPROX. 2000 SHEET OF 3 OF 7  
REQ NO. OR CONTRACT NO. 710717 DATE 7/27/77  
QUANTITY 2,500,000 @ 200.00 /MO. BID RATES 1 / 1 PY ENR

## COST ESTIMATE DETAIL

PRODUCTION LABOR AND BURDEN 1977										MATERIAL /EA	
DEPT.	STANDARD HOURS	EFF. %	SALV. %	SCRAP %	ADJUSTED HOURS	LABOR RATE	LABOR COST	BURDEN RATE	BURDEN COST	TYPE	STANDARD COST W/ADJ % SCRAP
MB02	28.10	90	4	4	26.60	9.19	243.86	183%	288.77	PURCHASE	.10899
MB02	0.51	80	1	1.2	0.78	9.59	7.43	343%	25.60		.10899
MB03	50.00	85	1	2	64.83	9.19	596.96	133%	706.18		
							756.24		1,021.42	SUBTOTAL PURCH.	
										SUBCONTRACT	
										SUBTOTAL S.C.	
										OTHER	
										SUBTOTAL OTHER	
TOTAL										TOTAL	

## INSPECTION AND Q.C. ENG LABOR AND BURDEN

TYPE	ESTIMATED 19 77										TOTAL	
	LABOR RATE	LABOR	BURDEN RATE	BURDEN	MM/HR	LABOR RATE	LABOR	BURDEN RATE	BURDEN	LABOR	BURDEN	
REC. INSPECTION	35	9.92	77%									
MFG. INSPECTION	410.5											
ASS'Y. INSPECTION	211.3											
TOTAL INSPECTION COST/EA		7844.78		0.039								
Q.C. ENGR. PROC.												
Q.C. ENGR. MFG.												
Q.C. ENGR. ASS'Y.	360	12.64	41.54%	77%								
Cal Tech.	16	10.50	168-									
TOTAL Q.C. ENG COST/EA		4592		0.023								



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**Honeywell**

DESCRIPTION ESTIMATED TRAINING SERVICE APT - Broomfield CO SHEET 4 OF 7

RFQ NO. OR CONTRACT NO. 1977 # DATE 7/20/77

QUANTITY 200 H/HR / MO. BID RATES / BY

**COST DETAIL**

ENGINEERING LABOR AND BURDEN										
TYPE	I.T.O.	MM OR \$	ESTIMATED 1977			ESTIMATED 19			TOTAL	
			MM/HR	LABOR RATE	BURDEN RATE	MM/HR	LABOR RATE	BURDEN RATE	LABOR	BURDEN
PRODUCTION ENGR.			5.0	246.8	12,340	74.8				
TOTAL PRODUCTION ENG.					12,340					
COST FA					0.043	0.046				
PRODUCT ENG.										
DEVELOPMENT ENG.			15.0	246.8	4440	84.2				
QA/TESTING			20.0	170.8	512	82.3				
EVAL ENGR										
MISC. TECH. & DRAFT										
E.V. TECH. ST. ANAL.										
CM Tech (S-CENT)			39.0	19.08	993					
MATERIALS			20.0	181.2	769					
TOTAL PROD ENG					5814					
COST / EA					0.029	0.024				
TOOLING ESTIMATE										
TYPE	LABOR RATE	BURDEN RATE	HRS	LABOR	BURDEN	HRS	LABOR	BURDEN	AUTO-TOOLING	
TOOL DESIGN										
TOOL MAKE										
MACHINE LAB										
MAINTENANCE										
SUBTOTAL LABOR & BURDEN										
PURCHASE TOOLING										
VENDOR TOOLING										
MATL. ACQ.										
TOTAL										
COST /										





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Honeywell														
COST ESTIMATE DETAIL														
DESCRIPTION: <u>EXTENDED RANGE TRIPLINE SENSOR 100 RB. MACHSHEET</u> OF <u>7</u>														
RFO NO. OR CONTRACT NO. <u>244,000</u> DATE <u>9/10/78</u>														
QUANTITY <u>244,000</u> @ <u>200,000</u> /MO. BID RATES <u>1</u> BY <u>FHE</u>														
PRODUCTION LABOR AND BURDEN 1977														
DEPT.	STANDARD HOURS	EFF. %	SALV. %	ADJUSTED HOURS	LABOR RATE	LABOR COST	BURDEN RATE	BURDEN COST	TYPE	STANDARD COST	MATERIAL /EA	ADJUSTED COST W/15% SCRAP		
Mach NB 105	15.83	90	4	19.06	8.19	156.10	133%	207.61	PURCHASE					
Mach NB 105	13.61	85	8	17.65	9.19	144.55	133%	192.35			159.74	187.93		
Mach NB 502	0.51	80	8	0.78	9.53	7.43	343%	2.66						
						308.08		425.36	SUBTOTAL PURCH.					
									SUBCONTRACT					
ALTERNATE IF LEADWIRE SENSORS ARE INCLUDED IN MACHINE PURCHASE														
Mach NB 105	15.83	90	4	19.06	8.19	156.10	133%	207.61						
Mach NB 105	6.61	85	8	8.57	8.19	70.19	133%	93.35	SUBTOTAL S.C.					
Mach NB 502	0.51	80	8	0.78	9.53	7.43	343%	2.66	OTHER					
						233.72		326.76	SUBTOTAL OTHER					
TOTAL									TOTAL					
INSPECTION AND Q.C. ENG LABOR AND BURDEN														
TYPE	ESTIMATED 19					ESTIMATED 19					TOTAL			
	MM/HR	LABOR RATE	LABOR	BURDEN RATE	BURDEN	MM/HR	LABOR RATE	LABOR	BURDEN RATE	BURDEN	LABOR	BURDEN	LABOR	BURDEN
REC. INSPECTION														
MFG. INSPECTION														
ASSY. INSPECTION														
TOTAL INSPECTION COST/														
Q.C. ENGR. PROC.														
Q.C. ENGR. MFG.														
Q.C. ENGR. ASSY.														
TOTAL Q.C. ENG COST/														

HC-301 REV 11/75



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# Honeywell

## COST DATA BY PART/MAKE OR BUY SUMMARY

SOURCE CODE:  
A. COST  
B. QUALITY  
C. DELIVERY  
D. TOOLING  
E. FACILITIES  
F. ENGINEERING CONTROL  
G. PROCESSING  
H. OVERLOAD  
I. UNDERLOAD  
J. INTER DIVISION  
K. OTHER

SYSTEM \_\_\_\_\_  
DEVICE NUMBER \_\_\_\_\_  
DEVICE NAME TRIPLINE SENSOR EXTENDER  
BID QUANTITY \_\_\_\_\_

SHEET 7 OF 7  
INITIAL HR DATE 8/25/74  
TOTAL STD COST LISTING DTD \_\_\_\_\_

PART NUMBER	QASH NO.	R E V	PART DESCRIPTION	USAGE PER DEVICE	UNIT COST	PURCH. QTY	MATERIAL QTY	MAKE QTY	SOURCING
9292972			Sensor Assy						
9292976			Sensor Case / Base Assy						
9292976A			Case-Sensor-In-Asst						
9292976B			Key-Mech-Case Assy						
9292977			Key-Mech Assy						
9292978			Spring Washer						
9292979			Housing Assy						
9292980			Top Housing						
9292981			Cap Housing						
9292982			Diaphragm Assy						
9292983			Gasket						
9292984			Diaphragm						
9292985			Diaphragm						
9292986			Eyelet						
9292987			Plate Diaphragm						
9292988			Ring Lock						
9292989			Term. R.H.						
9292990			" L.H.						
9292991			Bay Locking						
9292992			Case-Sensor						
9292993			Spring						
9292994			Washer, Case						
9292995			Spring, Booster						
9292996			Washer						
9292997			Spring, Elect.						
9292998			Spring Assy						
9292999			Spring, Diaphragm						
9293000			Term						
9293001			Assembly						
9293002			Bay Locking						
9293003			Cap						
9293004			Pin-Case						
9293005			Ring, Spring						
9293006			Washer, Diaphragm						
9293007			Diaphragm Wat.						
TOTAL THIS SHEET						199.74	199.74	199.74	199.74
ACCUMULATED TOTALS						199.74	199.74	199.74	199.74

TRACEABILITY INDEX:  
BILL OF MATERIALS  
MATERIALS QUOTE  
LABOR ESTIMATES

PGS \_\_\_\_\_ THRU \_\_\_\_\_  
HC-41 REV 6/75

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9/1/78  
FNR  
SHEET 1 OF 11

XM 74 EXTENDED TRIPLINE SENSOR PEP PROGRAM

BUDGETARY IPF IN 1979+1980 DOLLARS (25% B in 1980)

TOTAL MACHINES FOR 20000/MO.		FACTORY	G+A@	TOTAL	FOR B	PRICE
		COST	15.1%	COST	10%	
<u>MACHINES: BUILD OF QTY SHOWN</u>						
#7	BOBBIN + WEIGHT ASSY.	4	839,689			
8	PISTON/CRANK ASSEMBLY	1	140,103			
9	HOUSING ASSEMBLY	1	123,489			
10	SENSOR RELEASE MECH.	2	338,445			
10A	TERMINAL ASSEMBLY	2	88,760			
11	SENSOR CASE + SLEEVE ASSY.	2	364,189			
12	SENSOR CASE + BOBBIN ASSY.	2	397,992			
12A	SPRING WIN + SRS	4	137,910			
12B	SENSOR CASE + SPRING ASSY.	2	330,638			
13	SENSOR + BREAKWIRE ASSY.	2	309,811			
13A	EPHY DISPENS. FITURE	2	80,870			
	MAGAZINES		74,412			
GRAND TOTAL MACHINES + MAGAZINES		3,250,257	490,789	3,741,046	374,105	4,115,151

BUILD SUPPORT

QUANTITY ENGR 520 HRS @ 26.48 = 13,770  
RELIABL. ENGR 399 HRS @ 40.51 = 15,920  
DESIGN ENGR 347 HRS @ 40.51 = 14,057  
PROD'N ENGR 6412 HRS @ 30.32 = 194,412  
INSPECTION 375 HRS @ 19.66 = 7,373

TOTAL BUILD SUPPORT

245,832 37,075 282,607 28,261 310,868

DEMONSTRATION TESTS

QUANTITY ENGR 4028 HRS @ 26.48 = 106,661  
RELIABL. ENGR 2126 HRS @ 40.51 = 86,124  
DESIGN ENGR 692 HRS @ 40.51 = 28,033  
PROD'N ENGR 1000 HRS @ 30.32 = 30,320  
INSPECTION 750 HRS @ 19.66 = 14,745

TOTAL DEMONSTRATION TESTS

265,883 40,140 306,021 30,603 336,624

OTHER COSTS

AREA PREP: BATT ENGR 500 HRS @ 30.32 = 15,160  
MAINT. 1940 HRS @ 21.18 = 41,085  
MATERIAL 5400 @ 4.44 = 23,976  
LOT ACCEPT. TEST ENGR. FOR IMP. TEST = 62,640  
LOT ACCEPT. TEST ENGR. FOR FINE TEST = 16,204

TOTAL OTHER COSTS

162,995 24,522 187,517 18,692 206,209

DE-BUG. + ACCEPTANCE HARDWARE

347,249 52,435 399,684 39,968 439,652

GRAND TOTALS

4,274,316 644,969 4,919,285 491,629 5,410,914



SHEET 2 OF 11

CUSTOMER ARRADCOM

HM-190 REV 1/76



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DATE 8/30/78 BY FNR  
BID RATES 8/7/78

## MACHINE COST ESTIMATES

CUSTOMER ARRADCOM

DEVICE/PROJECT XM74 EXTENDED TELEPHONE SERVICE P.E.P.

## BUDGETARY

☐ FIRM☐ **GOVERNMENT**

COMMERCIAL

DEPARTMENT	LABOR RATE	TERMINAL ASSEMBLY		SEWING CASES + SEWING APP		SEWING CASES + BAGGING APP		SPONGE WADERS	
		#10 A HRS	DOLLARS	#11 HRS	DOLLARS	#12 HRS	DOLLARS	#12 A HRS	DOLLARS
MATERIAL									
ACQ 4.4 %			8,413		38,630		36,403		34,761
			392		1,700		1,602		1,529
TOOL ROOM L	14.77	558	8,242	4,002	58,110	4,332	63,494	839	13,392
B 65 %			5,357		38,422		41,590		8,055
TOOL DESIGN L	8.63	50	782	105	1,641	164	2,563	121	1,891
B 80 %			235		492		769		567
MACH LAB L	15.87	398	6,316	983	15,600	1,236	19,615	189	2,999
B 138 %			8,716		21,528		27,069		4,139
MAINT L	18.89	343	4,764	811	14,265	785	14,904	143	1,986
B 32 %			1,524		3,605		3,489		636
L									
B %									
FACTORY COST			45,841		191,993		207,988		68,955
TOTAL COST W/ % G&A									
PRICE W/ % PROFIT									
NO. ADD'L MACH		1 @	43,519	1 @	172,195	1 @	192,804	1 @	68,955
SHIP COST									
TOTAL			86,760		364,188		397,892		137,910

SWEET 3 OF 11

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SHEET 4 OF 11

CUSTOMER ARRADCOM MACHINE COST ESTIMATES DATE 8/30/78 BY ENR ISSUE 1  
BID RATES 8/7/78

DEVICE/PROJECT XM 74 EXTENDED TROUBLE SENSORS REP.

☐ BUDGETARY

☐ FIRM

☐ GOVERNMENT

☐ COMMERCIAL

DEPARTMENT	LABOR HRS DOLLARS	Sensor Case + Spares Assy		Sensor + Beacons Assy		Empty Dispensing Fixture		MAGAZINES
		HRS	DOLLARS	HRS	DOLLARS	HRS	DOLLARS	
	15.47 448,719							
MATERIAL								
ACQ 4.4 %			34,431		34,988		7,652	43,527
			6,515		6,539		337	6,915
TOOL ROOM L	14.77	3,236	47,796	2,778	46,031	548	8,094	-
B 6.5 %			3,467		26,670		5,361	
TOOL DESIGN L	15.63	158	2,470	71	6,110	48	750	96
B 30 %			741		333		225	450
MACH LAB L	15.87	6,155	18,330	1,155	18,330	322	5,110	48
B 1.38 %			25,245		25,245		7,052	1,052
MAINT L	18.89	725	10,070	753	10,459	360	5,000	-
B 32 %			8,222		3,347		6,600	-
L								
B								
FACTORY COST			174,937		163,102		41,081	49,206
TOTAL COST W/ % G&A								
PRICE W/ % PROFIT								
NO. ADD'L MACH		10	155,701	10	144,309	10	395,389	10
SHIP COST								
TOTAL			330,638		309,911		80,870	98,412

100-100 REV 1/78



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SHEET 8 OF 11

FILE NO.		HONEYWELL COST SUMMARY	
DESCRIPTION		XM74 EXTENDER TOWING SEAME DE-ICE HARDWARE	
		FOR IPE MACHINES	
CONTRACT		SHEET _____ OF _____	
BY:	FNR		
DATE:	9/30/78		
COST PER			
PURCH. MATERIAL			
S.C. MATERIAL			
OTHER MATERIAL			
SUBTOTAL MATERIAL	246,794		
MAT'L ACQ. 0.44	10,809		
TOTAL MATERIAL	257,603		
PRODUCTION LABOR	33,232		
PRODUCTION BURDEN	6,964		
INSPECTION LABOR			
INSPECTION BURDEN			
SUBTOTAL			
SHRINK/SURPLUS 0			
QUALITY ENGR. LABOR			
QUALITY ENGR. BURDEN			
PROD'TN ENGR. LABOR			
PROD'TN ENGR. BURDEN			
PROD'T ENGR. LABOR			
PROD'T ENGR. BURDEN			
TOOLING/GAGES			
FACTORY COST	347,249		
GEN. & ADM. 0			
TOTAL COST			
PROFIT 0			
TOTAL PRICE			

HC-400 REV 7/74



SHEET 9 OF 11

MC-301 REV 11/75

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SHEET 10 OF 11

SHEET 1 OF 2

COST ESTIMATE RECAP  
DEVICE NO. EXTENDED RANGE TANKING SMOKE DESCRIPTION DE-BUY HARDWARE  
DATE 3/13/73 PROD. ENG. E. P. BROWN ON 3-13-73 BY 15,600  
CPE NO.            MATERIAL SOURCE  
1. Purchased  
2. Subcontract  
3. Inventory

INFORMATION FROM PRODUCTION ENGINEERING										INFORMATION FOR COST ESTIMATING & PRICING									
Part No.	Part Name	Q	U	REL. PER M	MEG	ASSY	MAT'L	MTG. LABOR	EXTD.	QTY	UNIT PRICE	TOTAL PRICE	QTY	UNIT PRICE	TOTAL PRICE	QTY	UNIT PRICE	TOTAL PRICE	QTY
27	270001 WIPER	1																	
28	280001 WIPER	1																	
29	290001 WIPER	1																	
30	300001 WIPER	1																	
31	310001 WIPER	1																	
32	320001 WIPER	1																	
33	330001 WIPER	1																	
34	340001 WIPER	1																	
35	350001 WIPER	1																	
36	360001 WIPER	1																	
37	370001 WIPER	1																	
38	380001 WIPER	1																	
39	390001 WIPER	1																	
40	400001 WIPER	1																	
41	410001 WIPER	1																	
42	420001 WIPER	1																	
43	430001 WIPER	1																	
44	440001 WIPER	1																	
45	450001 WIPER	1																	
46	460001 WIPER	1																	
47	470001 WIPER	1																	
48	480001 WIPER	1																	
49	490001 WIPER	1																	
50	500001 WIPER	1																	
51	510001 WIPER	1																	
52	520001 WIPER	1																	
53	530001 WIPER	1																	
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59	590001 WIPER	1																	
60	600001 WIPER	1																	
61	610001 WIPER	1																	
62	620001 WIPER	1																	
63	630001 WIPER	1																	
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67	670001 WIPER	1																	
68	680001 WIPER	1																	
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70	700001 WIPER	1																	
71	710001 WIPER	1																	
72	720001 WIPER	1																	
73	730001 WIPER	1																	
74	740001 WIPER	1																	
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80	800001 WIPER	1																	
81	810001 WIPER	1																	
82	820001 WIPER	1																	
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92	920001 WIPER	1																	
93	930001 WIPER	1																	
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96	960001 WIPER	1																	
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98	980001 WIPER	1																	
99	990001 WIPER	1																	
100	1000001 WIPER	1																	



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SHEET 2 of 2

SHEET 11 of 11

COST ESTIMATE RECORD  
DESCRIPTION: DE-GRASS HARDWARE

DEVICE NO. 51178  
DATE 5/1/78  
PROG. ENG. 51178

A. Estimate  
B. Quote  
C. History

MATERIAL SOURCE  
1. Purchased  
2. Subcontract

A. ADAM - 5/1/78  
B. Estimate of 5/1/78 per 15.00

INFORMATION FROM PRODUCTION ENGINEERING										INFORMATION FOR COST ESTIMATING & PRICING									
Part No.	Part Name	Q	V	UFG	ASSY	MAT'L	Mfg. Labor Extended	Atty Labor	Material	Act	Est	Est	Est	Est	Est	Est	Est	Est	Est
21000001	21000001																		
21000002	21000002																		
21000003	21000003																		
21000004	21000004																		
21000005	21000005																		
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